

# The Open Source CFD Toolbox

User Guide

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# Chapter 1 Introduction

This guide accompanies the release of version 2.1.0 of the Open Source Field Operation and Manipulation (OpenFOAM) C++ libraries. It provides a description of the basic operation of OpenFOAM, first through a set of tutorial exercises in chapter 2 and later by a more detailed description of the individual components that make up OpenFOAM.

OpenFOAM is first and foremost a C++ library, used primarily to create executables, known as *applications*. The applications fall into two categories: *solvers*, that are each designed to solve a specific problem in continuum mechanics; and *utilities*, that are designed to perform tasks that involve data manipulation. The OpenFOAM distribution contains numerous solvers and utilities covering a wide range of problems, as described in chapter 3.

One of the strengths of OpenFOAM is that new solvers and utilities can be created by its users with some pre-requisite knowledge of the underlying method, physics and programming techniques involved.

OpenFOAM is supplied with pre- and post-processing environments. The interface to the pre- and post-processing are themselves OpenFOAM utilities, thereby ensuring consistent data handling across all environments. The overall structure of OpenFOAM is shown in Figure 1.1. The pre-processing and running of OpenFOAM cases is described



Figure 1.1: Overview of OpenFOAM structure.

in chapter 4. In chapter 5, we cover both the generation of meshes using the mesh generator supplied with OpenFOAM and conversion of mesh data generated by third-party products. Post-processing is described in chapter 6.

# Chapter 2

# Tutorials

In this chapter we shall describe in detail the process of setup, simulation and postprocessing for some OpenFOAM test cases, with the principal aim of introducing a user to the basic procedures of running OpenFOAM. The *\$FOAM\_TUTORIALS* directory contains many more cases that demonstrate the use of all the solvers and many utilities supplied with OpenFOAM. Before attempting to run the tutorials, the user must first make sure that they have installed OpenFOAM correctly.

The tutorial cases describe the use of the blockMesh pre-processing tool, case setup and running OpenFOAM solvers and post-processing using paraFoam. Those users with access to third-party post-processing tools supported in OpenFOAM have an option: either they can follow the tutorials using paraFoam; or refer to the description of the use of the third-party product in chapter 6 when post-processing is required.

Copies of all tutorials are available from the *tutorials* directory of the OpenFOAM installation. The tutorials are organised into a set of directories according to the type of flow and then subdirectories according to solver. For example, all the icoFoam cases are stored within a subdirectory *incompressible/icoFoam*, where *incompressible* indicates the type of flow. If the user wishes to run a range of example cases, it is recommended that the user copy the *tutorials* directory into their local *run* directory. They can be easily copied by typing:

mkdir -p \$FOAM\_RUN cp -r \$FOAM\_TUTORIALS \$FOAM\_RUN

# 2.1 Lid-driven cavity flow

This tutorial will describe how to pre-process, run and post-process a case involving isothermal, incompressible flow in a two-dimensional square domain. The geometry is shown in Figure 2.1 in which all the boundaries of the square are walls. The top wall moves in the x-direction at a speed of 1 m/s while the other 3 are stationary. Initially, the flow will be assumed laminar and will be solved on a uniform mesh using the icoFoam solver for laminar, isothermal, incompressible flow. During the course of the tutorial, the effect of increased mesh resolution and mesh grading towards the walls will be investigated. Finally, the flow Reynolds number will be increased and the pisoFoam solver will be used for turbulent, isothermal, incompressible flow.



Figure 2.1: Geometry of the lid driven cavity.

# 2.1.1 Pre-processing

Cases are setup in OpenFOAM by editing case files. Users should select an xeditor of choice with which to do this, such as emacs, vi, gedit, kate, nedit, *etc.* Editing files is possible in OpenFOAM because the I/O uses a dictionary format with keywords that convey sufficient meaning to be understood by even the least experienced users.

A case being simulated involves data for mesh, fields, properties, control parameters, etc. As described in section 4.1, in OpenFOAM this data is stored in a set of files within a case directory rather than in a single case file, as in many other CFD packages. The case directory is given a suitably descriptive name, *e.g.* the first example case for this tutorial is simply named **cavity**. In preparation of editing case files and running the first **cavity** case, the user should change to the case directory

cd \$FOAM\_RUN/tutorials/incompressible/icoFoam/cavity

#### 2.1.1.1 Mesh generation

OpenFOAM always operates in a 3 dimensional Cartesian coordinate system and all geometries are generated in 3 dimensions. OpenFOAM solves the case in 3 dimensions by default but can be instructed to solve in 2 dimensions by specifying a 'special' empty boundary condition on boundaries normal to the (3rd) dimension for which no solution is required.

The cavity domain consists of a square of side length d = 0.1 m in the x-y plane. A uniform mesh of 20 by 20 cells will be used initially. The block structure is shown in Figure 2.2. The mesh generator supplied with OpenFOAM, blockMesh, generates meshes from a description specified in an input dictionary, blockMeshDict located in the constant/polyMesh directory for a given case. The blockMeshDict entries for this case are as follows:

```
- C++
2
                                     OpenFOAM: The Open Source CFD Toolbox
                  F ield
3
                  O peration
                                     Version: 2.1.0
4
                                   Т
                                                www.OpenFOAM.org
5
                  A nd
                                     Web:
                  M anipulation
6
7
    FoamFile
8
9
        version
                     2.0;
10
```



Figure 2.2: Block structure of the mesh for the cavity.

```
format
                              ascii;
11
                              dictionary;
blockMeshDict;
            class
12
            object
13
14
      }
//
                                                                                                           * * * * * //
                   *
                       *
                                      *
15
          *
             *
                *
                         *
                             *
                                *
                                   *
                                         *
                                                        *
                                                           *
                                                                                                      *
                                                                                                         *
16
      convertToMeters 0.1;
17
18
      vertices
19
20
      (
            ^{21}
22
^{23}
24
25
26
27
28
      );
29
30
      blocks
31
32
            hex (0 1 2 3 4 5 6 7) (20 20 1) simpleGrading (1 1 1)
33
      );
34
35
      edges
36
      (
);
37
38
39
      boundary
40
41
      (
            movingWall
42
            {
43
                  type wall;
44
45
                  faces
46
                  (
                        (3 7 6 2)
47
                  );
48
            }
fixedWalls
{
49
50
51
                  type wall;
52
                  faces
53
54
                   (
                        \begin{array}{cccc} (0 & 4 & 7 & 3) \\ (2 & 6 & 5 & 1) \\ (1 & 5 & 4 & 0) \end{array}
55
56
57
                  );
58
59
            frontAndBack
60
61
                  type empty;
faces
(
62
63
64
                        (0 \ 3 \ 2 \ 1) \\ (4 \ 5 \ 6 \ 7)
65
66
                  );
67
            }
68
```

```
69 );

70 

71 mergePatchPairs

72 (

73 );

74 

75 // ***********
```

The file first contains header information in the form of a banner (lines 1-7), then file information contained in a *FoamFile* sub-dictionary, delimited by curly braces  $(\{\ldots\})$ .

#### For the remainder of the manual:

For the sake of clarity and to save space, file headers, including the banner and *FoamFile* sub-dictionary, will be removed from verbatim quoting of case files

The file first specifies coordinates of the block vertices; it then defines the blocks (here, only 1) from the vertex labels and the number of cells within it; and finally, it defines the boundary patches. The user is encouraged to consult section 5.3 to understand the meaning of the entries in the *blockMeshDict* file.

The mesh is generated by running **blockMesh** on this **blockMeshDict** file. From within the case directory, this is done, simply by typing in the terminal:

#### blockMesh

The running status of **blockMesh** is reported in the terminal window. Any mistakes in the *blockMeshDict* file are picked up by **blockMesh** and the resulting error message directs the user to the line in the file where the problem occurred. There should be no error messages at this stage.

#### 2.1.1.2 Boundary and initial conditions

Once the mesh generation is complete, the user can look at this initial fields set up for this case. The case is set up to start at time t = 0 s, so the initial field data is stored in a 0 sub-directory of the *cavity* directory. The 0 sub-directory contains 2 files, p and U, one for each of the pressure (p) and velocity (**U**) fields whose initial values and boundary conditions must be set. Let us examine file p:

```
[0 2 - 2 0 0 0 0];
   dimensions
17
18
   internalField
                   uniform 0;
19
20
   boundaryField
21
22
   {
       movingWall
23
24
       {
                           zeroGradient;
25
           type
       }
26
27
       fixedWalls
28
29
                           zeroGradient;
30
           type
       }
31
32
       frontAndBack
33
34
35
           type
                           empty;
       }
36
   }
37
38
                            39
```

There are 3 principal entries in field data files:

dimensions specifies the dimensions of the field, here *kinematic* pressure, *i.e.*  $m^2 s^{-2}$  (see section 4.2.6 for more information);

internalField the internal field data which can be uniform, described by a single value; or nonuniform, where all the values of the field must be specified (see section 4.2.8 for more information);

boundaryField the boundary field data that includes boundary conditions and data for all the boundary patches (see section 4.2.8 for more information).

For this case cavity, the boundary consists of walls only, split into 2 patches named: (1) fixedWalls for the fixed sides and base of the cavity; (2) movingWall for the moving top of the cavity. As walls, both are given a zeroGradient boundary condition for p, meaning "the normal gradient of pressure is zero". The frontAndBack patch represents the front and back planes of the 2D case and therefore must be set as empty.

In this case, as in most we encounter, the initial fields are set to be uniform. Here the pressure is kinematic, and as an incompressible case, its absolute value is not relevant, so is set to uniform 0 for convenience.

The user can similarly examine the velocity field in the 0/U file. The dimensions are those expected for velocity, the internal field is initialised as uniform zero, which in the case of velocity must be expressed by 3 vector components, *i.e.*uniform (0 0 0) (see section 4.2.5 for more information).

The boundary field for velocity requires the same boundary condition for the front-AndBack patch. The other patches are walls: a no-slip condition is assumed on the fixedWalls, hence a fixedValue condition with a value of uniform (0 0 0). The top surface moves at a speed of 1 m/s in the x-direction so requires a fixedValue condition also but with uniform (1 0 0).

#### 2.1.1.3 Physical properties

The physical properties for the case are stored in dictionaries whose names are given the suffix ... Properties, located in the Dictionaries directory tree. For an icoFoam case, the only property that must be specified is the kinematic viscosity which is stored from the transportProperties dictionary. The user can check that the kinematic viscosity is set correctly by opening the transportProperties dictionary to view/edit its entries. The keyword for kinematic viscosity is nu, the phonetic label for the Greek symbol  $\nu$  by which it is represented in equations. Initially this case will be run with a Reynolds number of 10, where the Reynolds number is defined as:

$$Re = \frac{d|\mathbf{U}|}{\nu} \tag{2.1}$$

where d and  $|\mathbf{U}|$  are the characteristic length and velocity respectively and  $\nu$  is the kinematic viscosity. Here d = 0.1 m,  $|\mathbf{U}| = 1 \text{ m s}^{-1}$ , so that for Re = 10,  $\nu = 0.01 \text{ m}^2 \text{ s}^{-1}$ . The correct file entry for kinematic viscosity is thus specified below:

#### 2.1.1.4 Control

Input data relating to the control of time and reading and writing of the solution data are read in from the *controlDict* dictionary. The user should view this file; as a case control file, it is located in the *system* directory.

The start/stop times and the time step for the run must be set. OpenFOAM offers great flexibility with time control which is described in full in section 4.3. In this tutorial

we wish to start the run at time t = 0 which means that OpenFOAM needs to read field data from a directory named 0 — see section 4.1 for more information of the case file structure. Therefore we set the startFrom keyword to startTime and then specify the startTime keyword to be 0.

For the end time, we wish to reach the steady state solution where the flow is circulating around the cavity. As a general rule, the fluid should pass through the domain 10 times to reach steady state in laminar flow. In this case the flow does not pass through this domain as there is no inlet or outlet, so instead the end time can be set to the time taken for the lid to travel ten times across the cavity, *i.e.* 1 s; in fact, with hindsight, we discover that 0.5 s is sufficient so we shall adopt this value. To specify this end time, we must specify the stopAt keyword as endTime and then set the endTime keyword to 0.5.

Now we need to set the time step, represented by the keyword deltaT. To achieve temporal accuracy and numerical stability when running icoFoam, a Courant number of less than 1 is required. The Courant number is defined for one cell as:

$$Co = \frac{\delta t |\mathbf{U}|}{\delta x} \tag{2.2}$$

where  $\delta t$  is the time step,  $|\mathbf{U}|$  is the magnitude of the velocity through that cell and  $\delta x$  is the cell size in the direction of the velocity. The flow velocity varies across the domain and we must ensure Co < 1 everywhere. We therefore choose  $\delta t$  based on the worst case: the maximum Co corresponding to the combined effect of a large flow velocity and small cell size. Here, the cell size is fixed across the domain so the maximum Co will occur next to the lid where the velocity approaches  $1 \text{ m s}^{-1}$ . The cell size is:

$$\delta x = \frac{d}{n} = \frac{0.1}{20} = 0.005 \text{ m}$$
(2.3)

Therefore to achieve a Courant number less than or equal to 1 throughout the domain the time step deltaT must be set to less than or equal to:

$$\delta t = \frac{Co \,\delta x}{|\mathbf{U}|} = \frac{1 \times 0.005}{1} = 0.005 \,\,\mathrm{s} \tag{2.4}$$

As the simulation progresses we wish to write results at certain intervals of time that we can later view with a post-processing package. The writeControl keyword presents several options for setting the time at which the results are written; here we select the timeStep option which specifies that results are written every nth time step where the value n is specified under the writeInterval keyword. Let us decide that we wish to write our results at times 0.1, 0.2,..., 0.5 s. With a time step of 0.005 s, we therefore need to output results at every 20th time time step and so we set writeInterval to 20.

OpenFOAM creates a new directory *named after the current time*, *e.g.* 0.1 s, on each occasion that it writes a set of data, as discussed in full in section 4.1. In the icoFoam solver, it writes out the results for each field, U and p, into the time directories. For this case, the entries in the *controlDict* are shown below:

17		
18	application	icoFoam;
19 20	startFrom	<pre>startTime;</pre>
21 22	startTime	0;
$\frac{23}{24}$	stopAt	endTime;
25 26	endTime	0.5;
27 28	deltaT	0.005;
29		

30	writeControl	<pre>timeStep;</pre>
31 32	writeInterval	20;
33 34 35	purgeWrite	0;
36	writeFormat	ascii;
37 38 39	writePrecision	6;
40	writeCompressio	n off;
$\frac{41}{42}$	timeFormat	general;
43 44	timePrecision	6;
$45 \\ 46 \\ 47$	runTimeModifiab	le true;
48 49	// **********	*********

#### 2.1.1.5 Discretisation and linear-solver settings

The user specifies the choice of finite volume discretisation schemes in the fvSchemes dictionary in the *system* directory. The specification of the linear equation solvers and tolerances and other algorithm controls is made in the fvSolution dictionary, similarly in the *system* directory. The user is free to view these dictionaries but we do not need to discuss all their entries at this stage except for pRefCell and pRefValue in the *PISO* sub-dictionary of the fvSolution dictionary. In a closed incompressible system such as the cavity, pressure is relative: it is the pressure range that matters not the absolute values. In cases such as this, the solver sets a reference level by pRefValue in cell pRefCell. In this example both are set to 0. Changing either of these values will change the absolute pressure field, but not, of course, the relative pressures or velocity field.

# 2.1.2 Viewing the mesh

Before the case is run it is a good idea to view the mesh to check for any errors. The mesh is viewed in paraFoam, the post-processing tool supplied with OpenFOAM. The paraFoam post-processing is started by typing in the terminal from within the case directory

#### paraFoam

Alternatively, it can be launched from another directory location with an optional **-case** argument giving the case directory, *e.g.* 

```
paraFoam -case $FOAM_RUN/tutorials/incompressible/icoFoam/cavity
```

This launches the ParaView window as shown in Figure 6.1. In the Pipeline Browser, the user can see that ParaView has opened cavity.OpenFOAM, the module for the cavity case. Before clicking the Apply button, the user needs to select some geometry from the Mesh Parts panel. Because the case is small, it is easiest to select all the data by checking the box adjacent to the Mesh Parts panel title, which automatically checks all individual components within the respective panel. The user should then click the Apply button to load the geometry into ParaView.

The user should then open the Display panel that controls the visual representation of the selected module. Within the Display panel the user should do the following as shown in Figure 2.3: (1) set Color By Solid Color; (2) click Set Ambient Color and select an appropriate colour e.g. black (for a white background); (3) in the Style panel,



Figure 2.3: Viewing the mesh in paraFoam.

select Wireframe from the Representation menu. The background colour can be set by selecting View Settings... from Edit in the top menu panel.

Especially the first time the user starts ParaView, it is recommended that they manipulate the view as described in section 6.1.5. In particular, since this is a 2D case, it is recommended that Use Parallel Projection is selected in the General panel of View Settings window selected from the Edit menu. The Orientation Axes can be toggled on and off in the Annotation window or moved by drag and drop with the mouse.

# 2.1.3 Running an application

Like any UNIX/Linux executable, OpenFOAM applications can be run in two ways: as a foreground process, *i.e.* one in which the shell waits until the command has finished before giving a command prompt; as a background process, one which does not have to be completed before the shell accepts additional commands.

On this occasion, we will run icoFoam in the foreground. The icoFoam solver is executed either by entering the case directory and typing

icoFoam

at the command prompt, or with the optional -case argument giving the case directory, *e.g.* 

#### icoFoam -case \$FOAM\_RUN/tutorials/incompressible/icoFoam/cavity

The progress of the job is written to the terminal window. It tells the user the current time, maximum Courant number, initial and final residuals for all fields.

# 2.1.4 Post-processing

As soon as results are written to time directories, they can be viewed using paraFoam. Return to the paraFoam window and select the Properties panel for the cavity.OpenFOAM case module. If the correct window panels for the case module do not seem to be present at any time, please ensure that: cavity.OpenFOAM is highlighted in blue; eye button alongside it is switched on to show the graphics are enabled;

To prepare paraFoam to display the data of interest, we must first load the data at the required run time of 0.5 s. If the case was run while ParaView was open, the output data in time directories will not be automatically loaded within ParaView. To load the data the user should click Refresh Times in the Properties window. The time data will be loaded into ParaView.

#### 2.1.4.1 Isosurface and contour plots

To view pressure, the user should open the Display panel since it controls the visual representation of the selected module. To make a simple plot of pressure, the user should select the following, as described in detail in Figure 2.4: in the Style panel, select Surface from the Representation menu; in the Color panel, select Color by  $\bigcirc$  and Rescale to Data Range. Now in order to view the solution at t = 0.5 s, the user can use the VCR Controls or Current Time Controls to change the current time to 0.5. These are located in the toolbars below the menus at the top of the ParaView window, as shown in Figure 6.4. The pressure field solution has, as expected, a region of low pressure at the top left of the cavity and one of high pressure at the top right of the cavity as shown in Figure 2.5.

With the point icon  $(\circ P)$  the pressure field is interpolated across each cell to give a continuous appearance. Instead if the user selects the cell icon, @P, from the Color by menu, a single value for pressure will be attributed to each cell so that each cell will be denoted by a single colour with no grading.

A colour bar can be included by either by clicking the Toggle Color Legend Visibility button in the Active Variable Controls toolbar, or by selecting Show Color Legend from the View menu. Clicking the Edit Color Map button, either in the Active Variable Controls toolbar or in the Color panel of the Display window, the user can set a range of attributes of the colour bar, such as text size, font selection and numbering format for the scale. The colour bar can be located in the image window by drag and drop with the mouse.

New versions of ParaView default to using a colour scale of blue to white to red rather than the more common blue to green to red (rainbow). Therefore *the first time* that the user executes ParaView, they may wish to change the colour scale. This can be done by selecting Choose Preset in the Color Scale Editor and selecting Blue to Red Rainbow. After clicking the OK confirmation button, the user can click the Make Default button so that ParaView will always adopt this type of colour bar.

If the user rotates the image, they can see that they have now coloured the complete geometry surface by the pressure. In order to produce a genuine contour plot the user should first create a cutting plane, or 'slice', through the geometry using the Slice filter

Select Color by interpolat Rescale to Data Range Select Surface	1
Properties Display Information	
Image: State	
Color	
Interpolate Colors	
🗷 Map Scalars	
Apply Texture None -	
Color by	
Edit Color Map Rescale to Data Range	
_ Slice	
Slice Direction 🔹	
Slice ()	
Style	
Representation Surface 🔹	
Interpolation Gouraud 🗸	
Material None 👻	
Point size 5.00	
Line width	
Opacity 1.00	
Volume mapper	

Figure 2.4: Displaying pressure contours for the cavity case.



Figure 2.5: Pressures in the cavity case.

as described in section 6.1.6.1. The cutting plane should be centred at (0.05, 0.05, 0.005) and its normal should be set to (0, 0, 1) (click the Z Normal button). Having generated the cutting plane, the contours can be created using by the Contour filter described in section 6.1.6.

#### 2.1.4.2 Vector plots

Before we start to plot the vectors of the flow velocity, it may be useful to remove other modules that have been created, *e.g.* using the Slice and Contour filters described above. These can: either be deleted entirely, by highlighting the relevant module in the Pipeline Browser and clicking Delete in their respective Properties panel; or, be disabled by toggling the eye button for the relevant module in the Pipeline Browser.

We now wish to generate a vector glyph for velocity at the centre of each cell. We first need to filter the data to cell centres as described in section 6.1.7.1. With the cavity.OpenFOAM module highlighted in the Pipeline Browser, the user should select Cell Centers from the Filter->Alphabetical menu and then click Apply.

With these Centers highlighted in the Pipeline Browser, the user should then select Glyph from the Filter->Alphabetical menu. The Properties window panel should appear as shown in Figure 2.6. In the resulting Properties panel, the velocity field, U, is automatically selected in the vectors menu, since it is the only vector field present. By default the Scale Mode for the glyphs will be Vector Magnitude of velocity but, since the we may wish to view the velocities throughout the domain, the user should instead select off and Set Scale Factor to 0.005. On clicking Apply, the glyphs appear but, probably as a single colour, *e.g.* white. The user should colour the glyphs by velocity magnitude which, as usual, is controlled by setting Color by U in the Display panel. The user should also select Show Color Legend in Edit Color Map. The output is shown in Figure 2.7, in which uppercase Times Roman fonts are selected for the Color Legend headings and the labels are specified to 2 fixed significant figures by deselecting Automatic Label Format and entering %-#6.2f in the Label Format text box. The background colour is set to white in the General panel of View Settings as described in section 6.1.5.1.

Note that at the left and right walls, glyphs appear to indicate flow through the walls. On closer examination, however, the user can see that while the flow direction is normal to the wall, its magnitude is 0. This slightly confusing situation is caused by ParaView choosing to orientate the glyphs in the x-direction when the glyph scaling off and the velocity magnitude is 0.

#### 2.1.4.3 Streamline plots

Again, before the user continues to post-process in  $\mathsf{ParaView}$ , they should disable modules such as those for the vector plot described above. We now wish to plot streamlines of velocity as described in section 6.1.8.

With the cavity.OpenFOAM module highlighted in the Pipeline Browser, the user should then select Stream Tracer from the Filter menu and then click Apply. The Properties window panel should appear as shown in Figure 2.8. The Seed points should be specified along a Line Source running vertically through the centre of the geometry, *i.e.* from (0.05, 0, 0.005) to (0.05, 0.1, 0.005). For the image in this guide we used: a point Resolution of 21; Max Propagation by Length 0.5; Initial Step Length by Cell Length 0.01; and, Integration Direction BOTH. The Runge-Kutta 2 IntegratorType was used with default parameters.

On clicking Apply the tracer is generated. The user should then select Tube from the Filter menu to produce high quality streamline images. For the image in this report, we

Specify Set S Select	Parameters panel Scale Factor 0.005 Scale Mode off yph Type Arrow
Properties Display Informati	
Apply @ Reset	X Delete
Scalars p	-
Vectors	-
Glyph Type Arro	
_⊡-Arrow	
Tip Resolution	6
Tip Radius 🔿	- 0.1
Tip Length ————————————————————————————————————	- 0.35
Shaft Resolution -	6
Shaft Radius 🛛	0.03
🕱 Orient	
Scale Mode off	·
Set Scale Factor 0.005	Edit
Maximum Number of Points 5000	
🕱 Mask Points	
🗷 Random Mode	

Figure 2.6: Properties panel for the Glyph filter.



Figure 2.7: Velocities in the cavity case.

Properties D	splay Inform		Set Max Set Init Set Inte	rameters panel Propagation to Length 0.5 ial Step Length to Cell Length 0.01 gration Direction to BOTH .ine Source and set points and resolution
			elete 🔋	
Vectors	U	-	]	
Max. Propagation	Length	-	0.5	
Initial Step Length	Cell Length	-	0.01	
Integration Direction	BOTH	-	]	
Max. Steps	2000		]	
Term. Speed	1e-12		]	
Integrator Type	Runge-Kutt	a 2 🔽	]	
Minimum Step Length	Cell Length	-	0.01	
Maximum Step Length	Cell Length	-	0.01	
Maximum Erro	r 1e-06		]	
_⊡-Seeds				
Seed Type	L	ine Sour	ce 💌	
🕱 Show Line				
Point1 0.05	0		0.005	
Point2 0.05	0.1		0.005	
	X Axi	S		
	Y Axis	5		
	Z Axis	5		
Resolution	2	1		

Figure 2.8: Properties panel for the **Stream Tracer** filter.



Figure 2.9: Streamlines in the cavity case.

used: Num. sides 6; Radius 0.0003; and, Radius factor 10. The streamtubes are coloured by velocity magnitude. On clicking Apply the image in Figure 2.9 should be produced.

# 2.1.5 Increasing the mesh resolution

The mesh resolution will now be increased by a factor of two in each direction. The results from the coarser mesh will be mapped onto the finer mesh to use as initial conditions for the problem. The solution from the finer mesh will then be compared with those from the coarser mesh.

# 2.1.5.1 Creating a new case using an existing case

We now wish to create a new case named cavityFine that is created from cavity. The user should therefore clone the cavity case and edit the necessary files. First the user should create a new case directory at the same directory level as the cavity case, *e.g.* 

cd \$FOAM\_RUN/tutorials/incompressible/icoFoam mkdir cavityFine

The user should then copy the base directories from the cavity case into cavityFine, and then enter the cavityFine case.

cp -r cavity/constant cavityFine
cp -r cavity/system cavityFine
cd cavityFine

### 2.1.5.2 Creating the finer mesh

We now wish to increase the number of cells in the mesh by using blockMesh. The user should open the *blockMeshDict* file in an editor and edit the block specification. The blocks are specified in a list under the blocks keyword. The syntax of the block definitions is described fully in section 5.3.1.3; at this stage it is sufficient to know that following hex is first the list of vertices in the block, then a list (or vector) of numbers of cells in each direction. This was originally set to (20 20 1) for the cavity case. The user should now change this to (40 40 1) and save the file. The new refined mesh should then be created by running blockMesh as before.

### 2.1.5.3 Mapping the coarse mesh results onto the fine mesh

The mapFields utility maps one or more fields relating to a given geometry onto the corresponding fields for another geometry. In our example, the fields are deemed 'consistent' because the geometry and the boundary types, or conditions, of both source and target fields are identical. We use the -consistent command line option when executing mapFields in this example.

The field data that mapFields maps is read from the time directory specified by startFrom/startTime in the *controlDict* of the target case, *i.e.* those **into which** the results are being mapped. In this example, we wish to map the final results of the coarser mesh from case cavity onto the finer mesh of case cavityFine. Therefore, since these results are stored in the 0.5 directory of cavity, the startTime should be set to 0.5 s in the *controlDict* dictionary and startFrom should be set to startTime.

The case is ready to run mapFields. Typing mapFields -help quickly shows that map-Fields requires the source case directory as an argument. We are using the -consistent option, so the utility is executed from withing the *cavityFine* directory by

```
mapFields ../cavity -consistent
```

The utility should run with output to the terminal including:

```
Source: ".." "cavity"
Target: "." "cavityFine"
Create databases as time
Source time: 0.5
Target time: 0.5
Create meshes
Source mesh size: 400 Target mesh size: 1600
Consistently creating and mapping fields for time 0.5
    interpolating p
    interpolating U
```

End

#### 2.1.5.4 Control adjustments

To maintain a Courant number of less that 1, as discussed in section 2.1.1.4, the time step must now be halved since the size of all cells has halved. Therefore deltaT should be set to to 0.0025 s in the *controlDict* dictionary. Field data is currently written out at an interval of a fixed number of time steps. Here we demonstrate how to specify data output at fixed intervals of time. Under the writeControl keyword in *controlDict*, instead of requesting output by a fixed number of time steps with the timeStep entry, a fixed amount of run time can be specified between the writing of results using the runTime entry. In this case the user should specify output every 0.1 and therefore should set writeInterval to 0.1 and writeControl to runTime. Finally, since the case is starting with a the solution obtained on the coarse mesh we only need to run it for a short period to achieve reasonable convergence to steady-state. Therefore the endTime should be set to 0.7 s. Make sure these settings are correct and then save the file.

### 2.1.5.5 Running the code as a background process

The user should experience running icoFoam as a background process, redirecting the terminal output to a *log* file that can be viewed later. From the *cavityFine* directory, the user should execute:

icoFoam > log & cat log

### 2.1.5.6 Vector plot with the refined mesh

The user can open multiple cases simultaneously in ParaView; essentially because each new case is simply another module that appears in the Pipeline Browser. There is one minor inconvenience when opening a new case in ParaView because there is a prerequisite that the selected data is a file with a name that has an extension. However, in OpenFOAM, each case is stored in a multitude of files with no extensions within a specific directory

Select U S Select	pen Display pa x from Line Se Select arc_leng ect Scatter PI	ries —— gth ——
	rmation	
Show Line Series in Chart		
Plot Type	Scatter Plot	
Attribute Mode	◇ Point Data	-
X Axis Data	de Dete	
O Use Array Index From Y A		
Use Data Array	arc_length	
Component	Distance	<b>*</b>
Line Series		
I Variable I Ux	Legend Name	
<ul> <li>Ox</li> <li>Uy</li> <li>Uz</li> <li>P</li> <li>U: Magnitude</li> <li>U: X</li> <li>U: Y</li> <li>U: Z</li> <li>original_coordinates: X</li> <li>original_coordinates: Z</li> <li>arc_length</li> </ul>	UY UZ P U: Magnitude U: X U: Y U: Z original_coordinates: Mag original_coordinates: Y original_coordinates: Y original_coordinates: Z arc_length	
Enable Line Series		
Line Color	O Choose Color	
Line Thickness	1	<b></b>
Line Style	Solid	-
Chart Axes	Bottom-Left	•

Figure 2.10: Selecting fields for graph plotting.

structure. The solution, that the paraFoam script performs automatically, is to create a dummy file with the extension *.OpenFOAM* — hence, the cavity case module is called cavity.OpenFOAM.

However, if the user wishes to open another case directly from within ParaView, they need to create such a dummy file. For example, to load the cavityFine case the file would be created by typing at the command prompt:

cd \$FOAM\_RUN/tutorials/incompressible/icoFoam touch cavityFine/cavityFine.OpenFOAM

Now the cavityFine case can be loaded into ParaView by selecting Open from the File menu, and having navigated the directory tree, selecting cavityFine.OpenFOAM. The user can now make a vector plot of the results from the refined mesh in ParaView. The plot can be compared with the cavity case by enabling glyph images for both case simultaneously.

### 2.1.5.7 Plotting graphs

The user may wish to visualise the results by extracting some scalar measure of velocity and plotting 2-dimensional graphs along lines through the domain. OpenFOAM is well equipped for this kind of data manipulation. There are numerous utilities that do specialised data manipulations, and some, simpler calculations are incorporated into a single utility foamCalc. As a utility, it is unique in that it is executed by

```
foamCalc <calcType> <fieldName1 ... fieldNameN>
```

The calculator operation is specified in <calcType>; at the time of writing, the following operations are implemented: addSubtract; randomise; div; components; mag; magGrad; magSqr; interpolate. The user can obtain the list of <calcType> by deliberately calling one that does not exist, so that foamCalc throws up an error message and lists the types available, *e.g.* 

```
>> foamCalc xxxx
Selecting calcType xxxx
unknown calcType type xxxx, constructor not in hash table
Valid calcType selections are:
8
(
randomise
magSqr
magGrad
addSubtract
div
mag
interpolate
components
)
```

The components and mag calcTypes provide useful scalar measures of velocity. When "foamCalc components U" is run on a case, say *cavity*, it reads in the velocity vector field from each time directory and, in the corresponding time directories, writes scalar fields Ux, Uy and Uz representing the x, y and z components of velocity. Similarly "foamCalc mag U" writes a scalar field magU to each time directory representing the magnitude of velocity.

The user can run foamCalc with the components calcType on both cavity and cavityFine cases. For example, for the cavity case the user should do into the *cavity* directory and execute foamCalc as follows:

# cd \$FOAM\_RUN/tutorials/incompressible/icoFoam/cavity foamCalc components U

The individual components can be plotted as a graph in ParaView. It is quick, convenient and has reasonably good control over labelling and formatting, so the printed output is a fairly good standard. However, to produce graphs for publication, users may prefer to write raw data and plot it with a dedicated graphing tool, such as gnuplot or Grace/xmgr. To do this, we recommend using the sample utility, described in section 6.5 and section 2.2.3.

Before commencing plotting, the user needs to load the newly generated Ux, Uy and Uz fields into ParaView. To do this, the user should click the Refresh Times at the top of the Properties panel for the cavity.OpenFOAM module which will cause the new fields to be loaded into ParaView and appear in the Volume Fields window. Ensure the new fields are selected and the changes are applied, *i.e.* click Apply again if necessary. Also, data is interpolated incorrectly at boundaries if the boundary regions are selected in the Mesh Parts panel. Therefore the user should deselect the patches in the Mesh Parts panel, *i.e.*movingWall, fixedWall and frontAndBack, and apply the changes.

Now, in order to display a graph in ParaView the user should select the module of interest, *e.g.*cavity.OpenFOAM and apply the Plot Over Line filter from the Filter->Data Analysis menu. This opens up a new XY Plot window below or beside the existing 3D View window. A PlotOverLine module is created in which the user can specify the end points of the line in the Properties panel. In this example, the user should position the line vertically up the centre of the domain, *i.e.* from (0.05, 0, 0.005) to (0.05, 0.1, 0.005), in the Point1 and Point2 text boxes. The Resolution can be set to 100.



Figure 2.11: Plotting graphs in paraFoam.

On clicking Apply, a graph is generated in the XY Plot window. In the Display panel, the user should set Attribute Mode to Point Data. The Use Data Array option can be selected for the X Axis Data, taking the arc\_length option so that the x-axis of the graph represents distance from the base of the cavity.

The user can choose the fields to be displayed in the Line Series panel of the Display window. From the list of scalar fields to be displayed, it can be seen that the magnitude and components of vector fields are available by default, *e.g.* displayed as U:X, so that it was not necessary to create Ux using foamCalc. Nevertheless, the user should deselect all series except Ux (or U:x). A square colour box in the adjacent column to the selected series indicates the line colour. The user can edit this most easily by a double click of the mouse over that selection.

In order to format the graph, the user should modify the settings below the Line Series panel, namely Line Color, Line Thickness, Line Style, Marker Style and Chart Axes.

Also the user can click one of the buttons above the top left corner of the XY Plot. The third button, for example, allows the user to control View Settings in which the user can set title and legend for each axis, for example. Also, the user can set font, colour and alignment of the axes titles, and has several options for axis range and labels in linear or logarithmic scales.

Figure 2.11 is a graph produced using ParaView. The user can produce a graph however he/she wishes. For information, the graph in Figure 2.11 was produced with the options for axes of: Standard type of Notation; Specify Axis Range selected; titles in Sans Serif 12 font. The graph is displayed as a set of points rather than a line by activating the Enable Line Series button in the Display window. Note: if this button appears to be inactive by being "greyed out", it can be made active by selecting and deselecting the sets of variables in the Line Series panel. Once the Enable Line Series button is selected, the Line Style and Marker Style can be adjusted to the user's preference.

# 2.1.6 Introducing mesh grading

The error in any solution will be more pronounced in regions where the form of the true solution differ widely from the form assumed in the chosen numerical schemes. For example a numerical scheme based on linear variations of variables over cells can only generate an exact solution if the true solution is itself linear in form. The error is largest in regions where the true solution deviates greatest from linear form, *i.e.* where the change
It is useful to have an intuitive appreciation of the form of the solution before setting up any problem. It is then possible to anticipate where the errors will be largest and to grade the mesh so that the smallest cells are in these regions. In the **cavity** case the large variations in velocity can be expected near a wall and so in this part of the tutorial the mesh will be graded to be smaller in this region. By using the same number of cells, greater accuracy can be achieved without a significant increase in computational cost.

A mesh of  $20 \times 20$  cells with grading towards the walls will be created for the liddriven cavity problem and the results from the finer mesh of section 2.1.5.2 will then be mapped onto the graded mesh to use as an initial condition. The results from the graded mesh will be compared with those from the previous meshes. Since the changes to the *blockMeshDict* dictionary are fairly substantial, the case used for this part of the tutorial, cavityGrade, is supplied in the *\$FOAM\_RUN/tutorials/incompressible/icoFoam* directory.

#### 2.1.6.1 Creating the graded mesh

The mesh now needs 4 blocks as different mesh grading is needed on the left and right and top and bottom of the domain. The block structure for this mesh is shown in Figure 2.12. The user can view the *blockMeshDict* file in the *constant/polyMesh* subdirectory of cavi-



Figure 2.12: Block structure of the graded mesh for the cavity (block numbers encircled).

tyGrade; for completeness the key elements of the *blockMeshDict* file are also reproduced below. Each block now has 10 cells in the x and y directions and the ratio between largest and smallest cells is 2.

```
convertToMeters 0.1;
17
18
           vertices
19
20
                       (0 0)
                                   0)
21
                       \begin{array}{c} (0 & 0 & 0) \\ (0 & .5 & 0 & 0) \\ (1 & 0 & 0) \\ (0 & 0 & .5 & 0) \\ (0 & .5 & 0 & .5 \\ (1 & 0 & .5 & 0) \\ (0 & 1 & 0) \\ (0 & 1 & 0) \end{array}
22
23
24
                                               0)
25
26
27
                              . 5
                                         0
                         ĺΟ
                                    1
0)
28
                               1
29
                              0
.5
                                   0
                         0
                                         1
30
                                         0.1)
                        (0)
31
                              õ
                                   0.1)
32
                        Ì0
                              0.5\ 0.1)
33
                                   0.5
.5 0
                                              0.1)
.1)
                              5
0
1
                        (0
34
                                   .5 0.
0.1)
35
                       (0
36
```

```
(0.5 1 0.1)
(1 1 0.1)
37
38
     );
39
40
     blocks
41
42
          hex (0 1 4 3 9 10 13 12) (10 10 1) simpleGrading (2 2 1)
43
          hex (1 2 5 4 10 11 14 13) (10 10 1) simpleGrading (0.5 2 1)
44
          hex (3 4 7 6 12 13 16 15) (10 10 1) simpleGrading (2 0.5 1)
45
          hex (4 5 8 7 13 14 17 16) (10 10 1) simpleGrading (0.5 0.5 1)
46
47
     );
48
     edges
49
     (
);
50
51
52
     boundary
53
54
     (
          movingWall
55
56
          ſ
57
               type wall;
58
               faces
59
                    (6 15 16 7)
(7 16 17 8)
60
61
               );
62
63
          fixedWalls
64
65
               type wall;
66
               faces
67
68
                       12 15 6)
9 12 3)
1 10 9)
                    (3
69
                    (Ō
70
                    (Õ
71
                       2 11 10)
5 14 11)
                    (1
(2
72
73
                    (5
                       8
                          17
                             14)
74
               );
75
76
          frontAndBack
77
78
79
               type empty;
               faces
80
81
                       3
                            1)
2)
4)
                    (0)
                          4
82
                       4 5 2)
6 7 4)
7 8 5)
                    (1
(3
83
84
                    (4
(9
85
                    (9 10 13 12)
(10 11 14 13)
(12 13 16 15)
(13 14 17 16)
86
87
88
89
               );
90
          }
91
92
     );
93
     mergePatchPairs
94
     ();
95
96
97
     11
                    98
```

Once familiar with the *blockMeshDict* file for this case, the user can execute blockMesh from the command line. The graded mesh can be viewed as before using paraFoam as described in section 2.1.2.

#### 2.1.6.2 Changing time and time step

The highest velocities and smallest cells are next to the lid, therefore the highest Courant number will be generated next to the lid, for reasons given in section 2.1.1.4. It is therefore useful to estimate the size of the cells next to the lid to calculate an appropriate time step for this case.

When a nonuniform mesh grading is used, blockMesh calculates the cell sizes using a geometric progression. Along a length l, if n cells are requested with a ratio of R between

$$\delta x_s = l \frac{r-1}{\alpha r - 1} \tag{2.5}$$

where r is the ratio between one cell size and the next which is given by:

$$r = R^{\frac{1}{n-1}}$$
(2.6)

and

$$\alpha = \begin{cases} R & \text{for } R > 1, \\ 1 - r^{-n} + r^{-1} & \text{for } R < 1. \end{cases}$$
(2.7)

For the cavityGrade case the number of cells in each direction in a block is 10, the ratio between largest and smallest cells is 2 and the block height and width is 0.05 m. Therefore the smallest cell length is 3.45 mm. From Equation 2.2, the time step should be less than 3.45 ms to maintain a Courant of less than 1. To ensure that results are written out at convenient time intervals, the time step deltaT should be reduced to 2.5 ms and the writeInterval set to 40 so that results are written out every 0.1 s. These settings can be viewed in the cavityGrade/system/controlDict file.

The startTime needs to be set to that of the final conditions of the case cavityFine, *i.e.*0.7. Since cavity and cavityFine converged well within the prescribed run time, we can set the run time for case cavityGrade to 0.1 s, *i.e.* the endTime should be 0.8.

#### 2.1.6.3 Mapping fields

As in section 2.1.5.3, use mapFields to map the final results from case cavityFine onto the mesh for case cavityGrade. Enter the *cavityGrade* directory and execute mapFields by:

cd \$FOAM\_RUN/tutorials/incompressible/icoFoam/cavityGrade mapFields ../cavityFine -consistent

Now run icoFoam from the case directory and monitor the run time information. View the converged results for this case and compare with other results using post-processing tools described previously in section 2.1.5.6 and section 2.1.5.7.

### 2.1.7 Increasing the Reynolds number

The cases solved so far have had a Reynolds number of 10. This is very low and leads to a stable solution quickly with only small secondary vortices at the bottom corners of the cavity. We will now increase the Reynolds number to 100, at which point the solution takes a noticeably longer time to converge. The coarsest mesh in case cavity will be used initially. The user should make a copy of the cavity case and name it cavityHighRe by typing:

```
cd $FOAM_RUN/tutorials/incompressible/icoFoam
cp -r cavity cavityHighRe
```

#### 2.1.7.1 Pre-processing

Enter the cavityHighRe case and edit the *transportProperties* dictionary. Since the Reynolds number is required to be increased by a factor of 10, decrease the kinematic viscosity by a factor of 10, *i.e.* to  $1 \times 10^{-3}$  m<sup>2</sup> s<sup>-1</sup>. We can now run this case by restarting from the solution at the end of the cavity case run. To do this we can use the option of setting the startFrom keyword to latestTime so that icoFoam takes as its initial data the values stored in the directory corresponding to the most recent time, *i.e.* 0.5. The endTime should be set to 2 s.

#### 2.1.7.2 Running the code

Run icoFoam for this case from the case directory and view the run time information. When running a job in the background, the following UNIX commands can be useful:

- **nohup** enables a command to keep running after the user who issues the command has logged out;
- nice changes the priority of the job in the kernel's scheduler; a niceness of -20 is the highest priority and 19 is the lowest priority.

This is useful, for example, if a user wishes to set a case running on a remote machine and does not wish to monitor it heavily, in which case they may wish to give it low priority on the machine. In that case the nohup command allows the user to log out of a remote machine he/she is running on and the job continues running, while nice can set the priority to 19. For our case of interest, we can execute the command in this manner as follows:

```
cd $FOAM_RUN/tutorials/incompressible/icoFoam/cavityHighRe
nohup nice -n 19 icoFoam > log &
cat log
```

In previous runs you may have noticed that icoFoam stops solving for velocity U quite quickly but continues solving for pressure p for a lot longer or until the end of the run. In practice, once icoFoam stops solving for U and the initial residual of p is less than the tolerance set in the *fvSolution* dictionary (typically  $10^{-6}$ ), the run has effectively converged and can be stopped once the field data has been written out to a time directory. For example, at convergence a sample of the *log* file from the run on the cavityHighRe case appears as follows in which the velocity has already converged after 1.62 s and initial pressure residuals are small; No Iterations O indicates that the solution of U has stopped:

```
\frac{1}{2}
      Time = 1.63
3
      Courant Number mean: 0.108642 max: 0.818175
DILUPBiCG: Solving for Ux, Initial residual = 7.86044e-06, Final residual = 7.86044e-06,
 5
      No Iterations 0
DILUPBICG: Solving for Uy, Initial residual = 9.4171e-06, Final residual = 9.4171e-06,
 \frac{6}{7}
      No Iterations 0
DICPCG: Solving for p, Initial residual = 3.54721e-06, Final residual = 7.13506e-07,
8
9
      No Iterations 4
time step continuity errors : sum local = 6.46788e-09, global = -9.44516e-19,
10
11
      cumulative = 1.04595e-17
DICPCG: Solving for p, Initial residual = 2.15824e-06, Final residual = 9.95068e-07,
12
13
      No Iterations 3
time step continuity errors : sum local = 8.67501e-09, global = 7.54182e-19,
^{14}_{15}
      cumulative = 1.12136e-17
ExecutionTime = 1.02 s ClockTime = 1 s
16
18
      Time = 1.635
19
20
21
      Courant Number mean: 0.108643 max: 0.818176
      DILUPBiCG: Solving for Ux, Initial residual = 7.6728e-06, Final residual = 7.6728e-06, No Iterations 0
22
23
```

```
DILUPBiCG: Solving for Uy, Initial residual = 9.19442e-06, Final residual = 9.19442e-06,
No Iterations 0
DICPCG: Solving for p, Initial residual = 3.13107e-06, Final residual = 8.60504e-07,
No Iterations 4
time step continuity errors : sum local = 8.15435e-09, global = -5.84817e-20,
cumulative = 1.11552e-17
DICPCG: Solving for p, Initial residual = 2.16689e-06, Final residual = 5.27197e-07,
No Iterations 14
time step continuity errors : sum local = 3.45666e-09, global = -5.62297e-19,
cumulative = 1.05292e-17
ExecutionTime = 1.02 s ClockTime = 1 s
```

### 2.1.8 High Reynolds number flow

View the results in paraFoam and display the velocity vectors. The secondary vortices in the corners have increased in size somewhat. The user can then increase the Reynolds number further by decreasing the viscosity and then rerun the case. The number of vortices increases so the mesh resolution around them will need to increase in order to resolve the more complicated flow patterns. In addition, as the Reynolds number increases the time to convergence increases. The user should monitor residuals and extend the endTime accordingly to ensure convergence.

The need to increase spatial and temporal resolution then becomes impractical as the flow moves into the turbulent regime, where problems of solution stability may also occur. Of course, many engineering problems have very high Reynolds numbers and it is infeasible to bear the huge cost of solving the turbulent behaviour directly. Instead Reynolds-averaged simulation (RAS) turbulence models are used to solve for the mean flow behaviour and calculate the statistics of the fluctuations. The standard  $k - \varepsilon$  model with wall functions will be used in this tutorial to solve the lid-driven cavity case with a Reynolds number of  $10^4$ . Two extra variables are solved for: k, the turbulent kinetic energy; and,  $\varepsilon$ , the turbulent dissipation rate. The additional equations and models for turbulent flow are implemented into a OpenFOAM solver called **pisoFoam**.

#### 2.1.8.1 Pre-processing

Change directory to the cavity case in the  $FOAM_RUN/tutorials/incompressible/pisoFoam/-ras$  directory (N.B: the **pisoFoam/ras** directory). Generate the mesh by running blockMesh as before. Mesh grading towards the wall is not necessary when using the standard  $k - \varepsilon$  model with wall functions since the flow in the near wall cell is modelled, rather than having to be resolved.

A range of wall function models is available in OpenFOAM that are applied as boundary conditions on individual patches. This enables different wall function models to be applied to different wall regions. The choice of wall function models are specified through the turbulent viscosity field,  $\nu_t$  in the 0/nut file:

```
17
                          [0 \ 2 \ -1 \ 0 \ 0 \ 0];
     dimensions
18
19
     internalField
                          uniform 0;
20
21
     boundaryField
22
23
     {
          movingWall
24
25
          {
               type
value
26
                                    nutkWallFunction;
                                    uniform 0;
27
28
          fixedWalls
29
30
               type
value
                                    nutkWallFunction;
31
                                    uniform 0;
32
33
          frontAndBack
34
35
          {
               type
                                    empty;
36
          }
37
```

38 } 39

This case uses standard wall functions, specified by the nutWallFunction type on the movingWall and fixedWalls patches. Other wall function models include the rough wall functions, specified though the nutRoughWallFunction keyword.

The user should now open the field files for k and  $\varepsilon$  (0/k and 0/epsilon) and examine their boundary conditions. For a wall boundary condition,  $\varepsilon$  is assigned a epsilonWall-Function boundary condition and a kqRwallFunction boundary condition is assigned to k. The latter is a generic boundary condition that can be applied to any field that are of a turbulent kinetic energy type, *e.g.* k, q or Reynolds Stress R. The initial values for k and  $\varepsilon$  are set using an estimated fluctuating component of velocity U' and a turbulent length scale, l. k and  $\varepsilon$  are defined in terms of these parameters as follows:

$$k = \frac{1}{2} \overline{\mathbf{U}' \cdot \mathbf{U}'}$$

$$C^{0.75} k^{1.5}$$

$$(2.8)$$

$$\varepsilon = \frac{C_{\mu} - \kappa}{l} \tag{2.9}$$

where  $C_{\mu}$  is a constant of the  $k - \varepsilon$  model equal to 0.09. For a Cartesian coordinate system, k is given by:

$$k = \frac{1}{2} (U'_x{}^2 + U'_y{}^2 + U'_z{}^2)$$
(2.10)

where  $U'_x{}^2$ ,  $U'_y{}^2$  and  $U'_z{}^2$  are the fluctuating components of velocity in the x, y and z directions respectively. Let us assume the initial turbulence is isotropic, *i.e.*  $U'_x{}^2 = U'_y{}^2 = U'_z{}^2$ , and equal to 5% of the lid velocity and that l, is equal to 20% of the box width, 0.1 m, then k and  $\varepsilon$  are given by:

$$U'_{x} = U'_{y} = U'_{z} = \frac{5}{100} 1 \text{ m s}^{-1}$$
(2.11)

$$\Rightarrow k = \frac{3}{2} \left(\frac{5}{100}\right)^2 \,\mathrm{m}^2 \,\mathrm{s}^{-2} = 3.75 \times 10^{-3} \,\mathrm{m}^2 \,\mathrm{s}^{-2} \tag{2.12}$$

$$\varepsilon = \frac{C_{\mu}^{0.75} k^{1.5}}{l} \approx 7.65 \times 10^{-4} \text{ m}^2 \text{s}^{-3}$$
(2.13)

These form the initial conditions for k and  $\varepsilon$ . The initial conditions for U and p are (0,0,0) and 0 respectively as before.

Turbulence modelling includes a range of methods, *e.g.* RAS or large-eddy simulation (LES), that are provided in OpenFOAM. In most transient solvers, the choice of turbulence modelling method is selectable at run-time through the **simulationType** keyword in *turbulenceProperties* dictionary. The user can view this file in the *constant* directory:

The options for simulationType are laminar, RASModel and LESModel. With RASModel selected in this case, the choice of RAS modelling is specified in a *RASProperties* file, also in the *constant* directory. The turbulence model is selected by the RASModel entry from a long list of available models that are listed in Table 3.9. The kEpsilon model should be selected which is is the standard  $k - \varepsilon$  model; the user should also ensure that turbulence calculation is switched on.

The coefficients for each turbulence model are stored within the respective code with a set of default values. Setting the optional switch called printCoeffs to on will make the default values be printed to standard output, *i.e.* the terminal, when the model is called at run time. The coefficients are printed out as a sub-dictionary whose name is that of the model name with the word Coeffs appended, *e.g.* kEpsilonCoeffs in the case of the kEpsilon model. The coefficients of the model, *e.g.* kEpsilon, can be modified by optionally including (copying and pasting) that sub-dictionary within the RASProperties dictionary and adjusting values accordingly.

The user should next set the laminar kinematic viscosity in the *transportProperties* dictionary. To achieve a Reynolds number of  $10^4$ , a kinematic viscosity of  $10^{-5}$  m is required based on the Reynolds number definition given in Equation 2.1.

Finally the user should set the startTime, stopTime, deltaT and the writeInterval in the *controlDict*. Set deltaT to 0.005 s to satisfy the Courant number restriction and the endTime to 10 s.

#### 2.1.8.2 Running the code

Execute pisoFoam by entering the case directory and typing "pisoFoam" in a terminal. In this case, where the viscosity is low, the boundary layer next to the moving lid is very thin and the cells next to the lid are comparatively large so the velocity at their centres are much less than the lid velocity. In fact, after  $\approx 100$  time steps it becomes apparent that the velocity in the cells adjacent to the lid reaches an upper limit of around  $0.2 \text{ m s}^{-1}$  hence the maximum Courant number does not rise much above 0.2. It is sensible to increase the solution time by increasing the time step to a level where the Courant number is much closer to 1. Therefore reset deltaT to 0.02 s and, on this occasion, set startFrom to latestTime. This instructs pisoFoam to read the start data from the latest time directory, *i.e.10.0*. The endTime should be set to 20 s since the run converges a lot slower than the laminar case. Restart the run as before and monitor the convergence of the solution converges to a steady-state or perhaps reaches some periodically oscillating state. In the latter case, convergence may never occur but this does not mean the results are inaccurate.

#### 2.1.9 Changing the case geometry

A user may wish to make changes to the geometry of a case and perform a new simulation. It may be useful to retain some or all of the original solution as the starting conditions for the new simulation. This is a little complex because the fields of the original solution are not consistent with the fields of the new case. However the mapFields utility can map fields that are inconsistent, either in terms of geometry or boundary types or both.

As an example, let us go to the cavityClipped case in the *icoFoam* directory which consists of the standard cavity geometry but with a square of length 0.04 m removed from the bottom right of the cavity, according to the *blockMeshDict* below:

```
convertToMeters 0.1;
17
18
    vertices
19
20
         21
22
23
24
25
26
27
28
29
```

```
(0 \ 0 \ 0.1) \\ (0.6 \ 0 \ 0.1) \\ (0 \ 0.4 \ 0.1) \\ (0.6 \ 0.4 \ 0.1) \\ (1 \ 0.4 \ 0.1) \\ (0 \ 1 \ 0.1) \\ (0.6 \ 1 \ 0.1) \\ (1 \ 1 \ 0 \ 1) \\ (1 \ 1 \ 0 \ 1) \\ (1 \ 1 \ 0 \ 1) \\ (1 \ 1 \ 0 \ 1) \\ (1 \ 1 \ 0 \ 1) \\ (1 \ 1 \ 0 \ 1) \\ (1 \ 1 \ 0 \ 1) \\ (1 \ 1 \ 0 \ 1) \\ (1 \ 1 \ 0 \ 1) \\ (1 \ 1 \ 0 \ 1) \\ (1 \ 1 \ 0 \ 1) \\ (1 \ 1 \ 0 \ 1) \\ (1 \ 1 \ 0 \ 1) \\ (1 \ 1 \ 0 \ 1) \\ (1 \ 1 \ 0 \ 1) \\ (1 \ 1 \ 0 \ 1) \\ (1 \ 1 \ 0 \ 1) \\ (1 \ 1 \ 0 \ 1) \\ (1 \ 1 \ 0 \ 1) \\ (1 \ 1 \ 0 \ 1) \\ (1 \ 1 \ 0 \ 1) \\ (1 \ 1 \ 0 \ 1) \\ (1 \ 1 \ 0 \ 1) \\ (1 \ 1 \ 0 \ 1) \\ (1 \ 1 \ 0 \ 1) \\ (1 \ 1 \ 0 \ 1) \\ (1 \ 1 \ 0 \ 1) \\ (1 \ 1 \ 0 \ 1) \\ (1 \ 1 \ 0 \ 1) \\ (1 \ 1 \ 0 \ 1) \\ (1 \ 1 \ 0 \ 1) \\ (1 \ 1 \ 0 \ 1) \\ (1 \ 1 \ 0 \ 1) \\ (1 \ 1 \ 0 \ 1) \\ (1 \ 1 \ 0 \ 1) \\ (1 \ 1 \ 0 \ 1) \\ (1 \ 1 \ 0 \ 1) \\ (1 \ 1 \ 0 \ 1) \\ (1 \ 1 \ 0 \ 1) \\ (1 \ 1 \ 0 \ 1) \\ (1 \ 1 \ 0 \ 1) \\ (1 \ 1 \ 0 \ 1) \\ (1 \ 1 \ 0 \ 1) \\ (1 \ 1 \ 0 \ 1) \\ (1 \ 1 \ 0 \ 1) \\ (1 \ 1 \ 0 \ 1) \\ (1 \ 1 \ 0 \ 1) \\ (1 \ 1 \ 0 \ 1) \\ (1 \ 1 \ 0 \ 1) \\ (1 \ 1 \ 0 \ 1) \\ (1 \ 1 \ 0 \ 1) \\ (1 \ 1 \ 0 \ 1) \\ (1 \ 1 \ 0 \ 1) \\ (1 \ 1 \ 0 \ 1) \\ (1 \ 1 \ 0 \ 1) \\ (1 \ 1 \ 0 \ 1) \\ (1 \ 1 \ 0 \ 1) \\ (1 \ 1 \ 0 \ 1) \\ (1 \ 1 \ 0 \ 1) \\ (1 \ 1 \ 0 \ 1) \\ (1 \ 1 \ 0 \ 1) \\ (1 \ 1 \ 0 \ 1) \ (1 \ 1 \ 0 \ 1) \ (1 \ 1 \ 0 \ 1) \ (1 \ 1 \ 0 \ 1) \ (1 \ 1 \ 1) \ (1 \ 1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1 \ 1) \ (1
30
31
32
33
34
35
36
                                                         1 0.1)
                                             (1)
37
38
                     );
39
40
                     blocks
41
42
                                          hex (0 1 3 2 8 9 11 10) (12 8 1) simpleGrading (1 1 1)
43
                                          hex (2 3 6 5 10 11 14 13) (12 12 1) simpleGrading (1 1 1)
44
                                          hex (3 4 7 6 11 12 15 14) (8 12 1) simpleGrading (1 1 1)
45
                     );
46
47
                     edges
48
                     (
);
49
50
51
                     boundary
52
53
                      (
                                          lid
{
54
55
                                                                type wall;
56
                                                               faces
57
58
                                                                (
                                                                                       (5 13 14 6)
59
                                                                                      (6 14 15 7)
60
                                                               );
61
62
                                          fixedWalls
63
64
                                                                type wall;
65
                                                                faces
66
                                                                (
67
                                                                                                  8 10 2)
10 13 5)
                                                                                       (0
68
                                                                                      (2
(7
69
                                                                                                   15
                                                                                                                  12 4)
70
                                                                                                   12 11 3)
11 9 1)
                                                                                       (4
71
                                                                                       (3
72
                                                                                       (1
                                                                                                  980)
73
                                                               );
74
75
                                           frontAndBack
76
77
                                                                type empty;
78
                                                                faces
79
                                                                                  (0 2 3 1)
(2 5 6 3)
(3 6 7 4)
(8 9 11 10)
(10 11 14 13)
(11 12 15 14)
80
81
82
83
84
85
86
                                                               );
87
                                          }
88
89
                     );
90
                     mergePatchPairs
91
92
                     );
93
94
                                                                                                                                                                                                                                                                                                       ***********************//
95
```

Generate the mesh with blockMesh. The patches are set accordingly as in previous cavity cases. For the sake of clarity in describing the field mapping process, the upper wall patch is renamed lid, previously the movingWall patch of the original cavity.

In an inconsistent mapping, there is no guarantee that all the field data can be mapped from the source case. The remaining data must come from field files in the target case itself. Therefore field data must exist in the time directory of the target case before mapping takes place. In the cavityClipped case the mapping is set to occur at time 0.5 s, since the startTime is set to 0.5 s in the controlDict. Therefore the user needs to copy initial field data to that directory, *e.g.* from time 0:

cd \$FOAM\_RUN/tutorials/incompressible/icoFoam/cavityClipped

cp -r 0 0.5

Before mapping the data, the user should view the geometry and fields at  $0.5~{\rm s.}$ 

Now we wish to map the velocity and pressure fields from cavity onto the new fields of cavityClipped. Since the mapping is inconsistent, we need to edit the *mapFieldsDict* dictionary, located in the *system* directory. The dictionary contains 2 keyword entries: patchMap and cuttingPatches. The patchMap list contains a mapping of patches from the source fields to the target fields. It is used if the user wishes a patch in the target field to inherit values from a corresponding patch in the source field. In cavityClipped, we wish to inherit the boundary values on the lid patch from movingWall in cavity so we must set the patchMap as:

```
patchMap
(
    lid movingWall
);
```

The cuttingPatches list contains names of target patches whose values are to be mapped from the source internal field through which the target patch cuts. In this case we will include the fixedWalls to demonstrate the interpolation process.

```
cuttingPatches
(
    fixedWalls
);
```

Now the user should run mapFields, from within the *cavityClipped* directory:

```
mapFields ../cavity
```

The user can view the mapped field as shown in Figure 2.13. The boundary patches have inherited values from the source case as we expected. Having demonstrated this, however, we actually wish to reset the velocity on the fixedWalls patch to (0,0,0). Edit the U field, go to the fixedWalls patch and change the field from nonuniform to uniform (0,0,0). The nonuniform field is a list of values that requires deleting in its entirety. Now run the case with icoFoam.

### 2.1.10 Post-processing the modified geometry

Velocity glyphs can be generated for the case as normal, first at time 0.5 s and later at time 0.6 s, to compare the initial and final solutions. In addition, we provide an outline of the geometry which requires some care to generate for a 2D case. The user should select Extract Block from the Filter menu and, in the Parameter panel, highlight the patches of interest, namely the lid and fixedWalls. On clicking Apply, these items of geometry can be displayed by selecting Wireframe in the Display panel. Figure 2.14 displays the patches in black and shows vortices forming in the bottom corners of the modified geometry.



Figure 2.13: cavity solution velocity field mapped onto cavityClipped.



Figure 2.14: cavityClipped solution for velocity field.

### 2.2 Stress analysis of a plate with a hole

This tutorial describes how to pre-process, run and post-process a case involving linearelastic, steady-state stress analysis on a square plate with a circular hole at its centre. The plate dimensions are: side length 4 m and radius R = 0.5 m. It is loaded with a uniform traction of  $\sigma = 10$  kPa over its left and right faces as shown in Figure 2.15. Two symmetry planes can be identified for this geometry and therefore the solution domain need only cover a quarter of the geometry, shown by the shaded area in Figure 2.15.



Figure 2.15: Geometry of the plate with a hole.

The problem can be approximated as 2-dimensional since the load is applied in the plane of the plate. In a Cartesian coordinate system there are two possible assumptions to take in regard to the behaviour of the structure in the third dimension: (1) the plane stress condition, in which the stress components acting out of the 2D plane are assumed to be negligible; (2) the plane strain condition, in which the strain components out of the 2D plane are assumed negligible. The plane stress condition is appropriate for solids whose third dimension is thin as in this case; the plane strain condition is applicable for solids where the third dimension is thick.

An analytical solution exists for loading of an infinitely large, thin plate with a circular hole. The solution for the stress normal to the vertical plane of symmetry is

$$(\sigma_{xx})_{x=0} = \begin{cases} \sigma \left( 1 + \frac{R^2}{2y^2} + \frac{3R^4}{2y^4} \right) & \text{for } |y| \ge R \\ 0 & \text{for } |y| < R \end{cases}$$
(2.14)

Results from the simulation will be compared with this solution. At the end of the tutorial, the user can: investigate the sensitivity of the solution to mesh resolution and mesh grading; and, increase the size of the plate in comparison to the hole to try to estimate the error in comparing the analytical solution for an infinite plate to the solution of this problem of a finite plate.

#### 2.2.1 Mesh generation

The domain consists of four blocks, some of which have arc-shaped edges. The block structure for the part of the mesh in the x - y plane is shown in Figure 2.16. As already mentioned in section 2.1.1.1, all geometries are generated in 3 dimensions in OpenFOAM even if the case is to be as a 2 dimensional problem. Therefore a dimension of the block in the z direction has to be chosen; here, 0.5 m is selected. It does not affect the solution since the traction boundary condition is specified as a stress rather than a force, thereby making the solution independent of the cross-sectional area.



Figure 2.16: Block structure of the mesh for the plate with a hole.

The user should change into the plateHole case in the *\$FOAM\_RUN/tutorials/stress-Analysis/solidDisplacementFoam* directory and open the *constant/polyMesh/blockMeshDict* file in an editor, as listed below

```
convertToMeters 1;
17
                    (0.5 0 0)
(1 0 0)
(2 0 0)
(2 0.707107 0)
(0.707107 0.707107 0)
(0.353553 0.353553 0)
2 2 0)
0.707107 2 0)
0 2 0)
0 1 0)
) 0.5 0)
1.5 0 0.5)
0 0.5)
18
          vertices
19
20
21
22
23
24
25
26
27
28
29
30
31
32
33
                               0.5)
0.5)
.707107 0.5)
07107 0.707107 0.5)
                      (2
(2
                            0
34
                            0
35
                            707107
36
```

```
(0.353553 0.353553 0.5)
37
              (2 2 0.5)
(0.707107 2 0.5)
38
39
              (0\ 2\ 0.5)
(0\ 1\ 0.5)
(0\ 0.5\ 0.5)
40
41
42
        );
43
44
       blocks
45
46
        (
              hex (5 4 9 10 16 15 20 21) (10 10 1) simpleGrading (1 1 1)
47
              hex (0 1 4 5 11 12 15 16) (10 10 1) simpleGrading (1 1 1)
48
              hex (1 2 3 4 12 13 14 15) (20 10 1) simpleGrading (1 1 1)
49
              hex (4 3 6 7 15 14 17 18) (20 20 1) simpleGrading (1 1 1)
50
51
              hex (9 4 7 8 20 15 18 19) (10 20 1) simpleGrading (1 1 1)
       );
52
53
        edges
54
        (
55
             arc 0 5 (0.469846 0.17101 0)
arc 5 10 (0.17101 0.469846 0)
arc 1 4 (0.939693 0.34202 0)
arc 4 9 (0.34202 0.939693 0)
arc 11 16 (0.469846 0.17101 0.5)
arc 16 21 (0.17101 0.469846 0.5)
arc 12 15 (0.939693 0.34202 0.5)
arc 15 20 (0.34202 0.939693 0.5)
56
57
 58
59
60
61
62
63
       );
64
65
        boundary
66
        (
67
              left
68
              {
69
                     type symmetryPlane;
 70
                     faces
71
                     (
 72
                            (8 9 20 19)
73
                            (9 10 21 20)
74
                     );
75
              }
 76
              right
77
 78
              {
 79
                     type patch;
                     faces
 80
 81
                     (
                           (2 \ 3 \ 14 \ 13)
(3 \ 6 \ 17 \ 14)
 82
 83
 84
                     );
              }
 85
              down
 86
 87
              {
                     type symmetryPlane;
 88
 89
                     faces
90
                     (
                           91
92
                    );
93
              }
94
95
              up
              {
96
97
                     type patch;
                     faces
98
99
                     (
                           (7 8 19 18)
(6 7 18 17)
100
101
                     );
102
              }
103
              hole
104
105
              {
                     type patch;
106
                     faces
107
                     (
108
                           (10 5 16 21)
(5 0 11 16)
109
110
                     );
111
112
              frontAndBack
113
114
              {
                     type empty;
115
116
                     faces
117
                     (
                           \begin{array}{ccccc} (10 & 9 & 4 & 5) \\ (5 & 4 & 1 & 0) \\ (1 & 4 & 3 & 2) \\ (4 & 7 & 6 & 3) \end{array}
118
119
120
121
```

122	(4 9 8 7)
	$(21 \ 16 \ 15 \ 20)$
123	
124	$(16 \ 11 \ 12 \ 15)$
125	(12 13 14 15)
126	(15 14 17 18)
127	(15 18 19 20)
128	);
129	
130	);
	<i>,</i> ,
131	
132	mergePatchPairs
133	
134	);
135	/ )
	//
136	// ************************************

Until now, we have only specified straight edges in the geometries of previous tutorials but here we need to specify curved edges. These are specified under the edges keyword entry which is a list of non-straight edges. The syntax of each list entry begins with the type of curve, including arc, simpleSpline, polyLine *etc.*, described further in section 5.3.1. In this example, all the edges are circular and so can be specified by the arc keyword entry. The following entries are the labels of the start and end vertices of the arc and a point vector through which the circular arc passes.

The blocks in this *blockMeshDict* do not all have the same orientation. As can be seen in Figure 2.16 the  $x_2$  direction of block 0 is equivalent to the  $-x_1$  direction for block 4. This means care must be taken when defining the number and distribution of cells in each block so that the cells match up at the block faces.

6 patches are defined: one for each side of the plate, one for the hole and one for the front and back planes. The left and down patches are both a symmetry plane. Since this is a *geometric* constraint, it is included in the definition of the *mesh*, rather than being purely a specification on the boundary condition of the fields. Therefore they are defined as such using a special symmetryPlane type as shown in the *blockMeshDict*.

The frontAndBack patch represents the plane which is ignored in a 2D case. Again this is a geometric constraint so is defined within the mesh, using the empty type as shown in the *blockMeshDict*. For further details of boundary types and geometric constraints, the user should refer to section 5.2.1.

The remaining patches are of the regular **patch** type. The mesh should be generated using **blockMesh** and can be viewed in **paraFoam** as described in section 2.1.2. It should appear as in Figure 2.17.



Figure 2.17: Mesh of the hole in a plate problem.

#### 2.2.1.1 Boundary and initial conditions

Once the mesh generation is complete, the initial field with boundary conditions must be set. For a stress analysis case without thermal stresses, only displacement D needs to be set. The 0/D is as follows:

```
[0 1 0 0 0 0 0];
    dimensions
17
18
                      uniform (0 \ 0 \ 0):
    internalField
19
20
    boundaryField
21
22
    ſ
         left
23
24
         {
25
             type
                               symmetryPlane;
         }
26
         right
27
         ł
28
                               tractionDisplacement;
             type
29
                               uniform ( 10000 0 0 );
uniform 0;
             traction
30
31
             pressure
             value
                               uniform (0 \ 0 \ 0);
32
         }
33
         down
34
         {
35
                               symmetryPlane;
             type
36
         }
37
38
         up
         ſ
39
                               tractionDisplacement;
40
             type
41
             traction
                               uniform (000);
             pressure
                               uniform 0;
42
                               uniform (0 0 0);
43
             value
44
         hole
45
46
                               tractionDisplacement;
47
             type
             traction
                               uniform (000);
48
                               uniform Ò;
             pressure
49
                               uniform (0 0 0);
             value
50
51
         frontAndBack
52
53
54
             type
                               empty;
         }
55
    }
56
57
                                              58
```

Firstly, it can be seen that the displacement initial conditions are set to (0,0,0) m. The left and down patches must be both of symmetryPlane type since they are specified as such in the mesh description in the *constant/polyMesh/boundary* file. Similarly the frontAndBack patch is declared empty.

The other patches are traction boundary conditions, set by a specialist traction boundary type. The traction boundary conditions are specified by a linear combination of: (1) a boundary traction vector under keyword traction; (2) a pressure that produces a traction normal to the boundary surface that is defined as negative when pointing out of the surface, under keyword pressure. The up and hole patches are zero traction so the boundary traction and pressure are set to zero. For the right patch the traction should be (1e4, 0, 0) Pa and the pressure should be 0 Pa.

#### 2.2.1.2 Mechanical properties

The physical properties for the case are set in the *mechanicalProperties* dictionary in the *constant* directory. For this problem, we need to specify the mechanical properties of steel given in Table 2.1. In the mechanical properties dictionary, the user must also set planeStress to yes.

Property	Units	Keyword	Value
Density	${ m kgm^{-3}}$	rho	7854
Young's modulus	Pa	E	$2 \times 10^{11}$
Poisson's ratio		nu	0.3

 Table 2.1: Mechanical properties for steel

#### 2.2.1.3 Thermal properties

The temperature field variable T is present in the solidDisplacementFoam solver since the user may opt to solve a thermal equation that is coupled with the momentum equation through the thermal stresses that are generated. The user specifies at run time whether OpenFOAM should solve the thermal equation by the thermalStress switch in the *thermalProperties* dictionary. This dictionary also sets the thermal properties for the case, *e.g.* for steel as listed in Table 2.2.

Property	Units	Keyword	Value
Specific heat capacity	$\rm Jkg^{-1}K^{-1}$	С	434
Thermal conductivity	$\mathrm{Wm^{-1}K^{-1}}$	k	60.5
Thermal expansion coeff.	$\mathrm{K}^{-1}$	alpha	$1.1 \times 10^{-5}$

Table 2.2: Thermal properties for steel

In this case we do not want to solve for the thermal equation. Therefore we must set the thermalStress keyword entry to no in the *thermalProperties* dictionary.

#### 2.2.1.4 Control

As before, the information relating to the control of the solution procedure are read in from the *controlDict* dictionary. For this case, the **startTime** is 0 s. The time step is not important since this is a steady state case; in this situation it is best to set the time step **deltaT** to 1 so it simply acts as an iteration counter for the steady-state case. The **endTime**, set to 100, then acts as a limit on the number of iterations. The **writeInterval** can be set to 20.

The *controlDict* entries are as follows:

17		
18	application	<pre>solidDisplacementFoam;</pre>
19 20	startFrom	<pre>startTime;</pre>
21 22	startTime	0;
23 24	stopAt	endTime;
25 26	endTime	100;
27 28	deltaT	1;
29 30	writeControl	<pre>timeStep;</pre>
31 32	writeInterval	20;
33 34	purgeWrite	0;
35 36	writeFormat	ascii;
37 38	writePrecision	6;
39 40	writeCompression	
41	timeFormat	
42	LIMEFOLMAT	general;

#### 2.2.1.5 Discretisation schemes and linear-solver control

Let us turn our attention to the *fvSchemes* dictionary. Firstly, the problem we are analysing is steady-state so the user should select SteadyState for the time derivatives in timeScheme. This essentially switches off the time derivative terms. Not all solvers, especially in fluid dynamics, work for both steady-state and transient problems but solid-DisplacementFoam does work, since the base algorithm is the same for both types of simulation.

The momentum equation in linear-elastic stress analysis includes several explicit terms containing the gradient of displacement. The calculations benefit from accurate and smooth evaluation of the gradient. Normally, in the finite volume method the discretisation is based on Gauss's theorem The Gauss method is sufficiently accurate for most purposes but, in this case, the least squares method will be used. The user should therefore open the fvSchemes dictionary in the *system* directory and ensure the leastSquares method is selected for the grad(U) gradient discretisation scheme in the gradSchemes sub-dictionary:

```
17
     d2dt2Schemes
18
     ſ
19
         default
                             steadyState;
20
     }
21
22
     ddtSchemes
23
^{24}
         default
                             Euler;
25
     }
26
27
    gradSchemes
^{28}
29
         default
                             leastSquares;
30
         grad(D)
                             leastSquares;
31
                             leastSquares;
         grad(T)
32
     }
33
34
35
     divSchemes
36
          default
                             none;
37
         div(sigmaD)
                             Gauss linear;
38
    }
39
40
    laplacianSchemes
41
42
     ſ
         default
                            none;
43
         laplacian(DD,D) Gauss linear corrected;
44
         laplacian(DT,T) Gauss linear corrected;
45
    }
46
47
    interpolationSchemes
48
49
    ſ
         default
                             linear;
50
     }
51
52
     snGradSchemes
53
54
         default
55
                             none;
     }
56
57
    fluxRequired
58
     ſ
59
         default
60
                             no;
         D
61
                             yes;
         T
                             ňo;
62
```

66

} 63 64 65 11

The *fvSolution* dictionary in the *system* directory controls the linear equation solvers and algorithms used in the solution. The user should first look at the *solvers* sub-dictionary and notice that the choice of **solver** for D is **GAMG**. The solver **tolerance** should be set to  $10^{-6}$  for this problem. The solver relative tolerance, denoted by relTol, sets the required reduction in the residuals within each iteration. It is uneconomical to set a tight (low) relative tolerance within each iteration since a lot of terms in each equation are explicit and are updated as part of the segregated iterative procedure. Therefore a reasonable value for the relative tolerance is 0.01, or possibly even higher, say 0.1, or in some cases even 0.9 (as in this case).

```
17
   solvers
18
19
        "(D|T)"
20
        {
21
                           GAMG
            solver
22
                           1e-06;
            tolerance
23
                           0.9;
           relTol
24
                           GaussSeidel;
           smoother
25
           cacheAgglomeration true;
nCellsInCoarsestLevel 20
26
27
            agglomerator
                           faceAreaPair;
28
           mĕrgeLevels
                           1;
29
        }
30
   }
31
32
    stressAnalysis
33
34
    {
        compactNormalStress yes;
35
36
        nCorrectors
                       1;
1e-06;
37
   }
38
39
40
    41
```

The *fvSolution* dictionary contains a sub-dictionary, *stressAnalysis* that contains some control parameters specific to the application solver. Firstly there is nCorrectors which specifies the number of outer loops around the complete system of equations, including traction boundary conditions within each time step. Since this problem is steady-state, we are performing a set of iterations towards a converged solution with the 'time step' acting as an iteration counter. We can therefore set **nCorrectors** to 1.

The D keyword specifies a convergence tolerance for the outer iteration loop, *i.e.* sets a level of initial residual below which solving will cease. It should be set to the desired solver tolerance specified earlier,  $10^{-6}$  for this problem.

#### 2.2.2Running the code

The user should run the code here in the background from the command line as specified below, so he/she can look at convergence information in the log file afterwards.

#### cd \$FOAM\_RUN/tutorials/stressAnalysis/solidDisplacementFoam/plateHole solidDisplacementFoam > log &

The user should check the convergence information by viewing the generated log file which shows the number of iterations and the initial and final residuals of the displacement in each direction being solved. The final residual should always be less than 0.9 times the initial residual as this iteration tolerance set. Once both initial residuals have dropped below the convergence tolerance of  $10^{-6}$  the run has converged and can be stopped by killing the batch job.

### 2.2.3 Post-processing

Post processing can be performed as in section 2.1.4. The solidDisplacementFoam solver outputs the stress field  $\sigma$  as a symmetric tensor field sigma. This is consistent with the way variables are usually represented in OpenFOAM solvers by the mathematical symbol by which they are represented; in the case of Greek symbols, the variable is named phonetically.

For post-processing individual scalar field components,  $\sigma_{xx}$ ,  $\sigma_{xy}$  etc., can be generated by running the foamCalc utility as before in section 2.1.5.7, this time on sigma:

#### foamCalc components sigma

Components named sigmaxx, sigmaxy *etc.* are written to time directories of the case. The  $\sigma_{xx}$  stresses can be viewed in paraFoam as shown in Figure 2.18.



Figure 2.18:  $\sigma_{xx}$  stress field in the plate with hole.

We would like to compare the analytical solution of Equation 2.14 to our solution. We therefore must output a set of data of  $\sigma_{xx}$  along the left edge symmetry plane of our domain. The user may generate the required graph data using the sample utility. The utility uses a *sampleDict* dictionary located in the *system* directory, whose entries are summarised in Table 6.3. The sample line specified in sets is set between (0.0, 0.5, 0.25) and (0.0, 2.0, 0.25), and the fields are specified in the fields list:

```
17
   interpolationScheme cellPoint;
18
19
   setFormat
20
                  raw:
^{21}
22
   sets
^{23}
       leftPatch
24
25
26
          type
                  uniform;
27
          axis
                  y;
(
                   0 0.5 0.25 );
          start
28
                   0 2 0.25);
29
          end
                 100;
          nPoints
30
       }
31
   );
32
33
   fields
                  ( sigmaxx );
34
35
36
   37
```



Figure 2.19: Normal stress along the vertical symmetry  $(\sigma_{xx})_{x=0}$ 

The user should execute sample as normal. The writeFormat is raw 2 column format. The data is written into files within time subdirectories of a sets directory, e.g. the data at t = 100 s is found within the file sets/100/leftPatch\_sigmaxx.xy. In an application such as GnuPlot, one could type the following at the command prompt would be sufficient to plot both the numerical data and analytical solution:

An example plot is shown in Figure 2.19.

### 2.2.4 Exercises

The user may wish to experiment with solidDisplacementFoam by trying the following exercises:

#### 2.2.4.1 Increasing mesh resolution

Increase the mesh resolution in each of the x and y directions. Use mapFields to map the final coarse mesh results from section 2.2.3 to the initial conditions for the fine mesh.

#### 2.2.4.2 Introducing mesh grading

Grade the mesh so that the cells near the hole are finer than those away from the hole. Design the mesh so that the ratio of sizes between adjacent cells is no more than 1.1 and so that the ratio of cell sizes between blocks is similar to the ratios within blocks. Mesh grading is described in section 2.1.6. Again use mapFields to map the final coarse mesh results from section 2.2.3 to the initial conditions for the graded mesh. Compare the results with those from the analytical solution and previous calculations. Can this solution be improved upon using the same number of cells with a different solution?

The analytical solution is for an infinitely large plate with a finite sized hole in it. Therefore this solution is not completely accurate for a finite sized plate. To estimate the error, increase the plate size while maintaining the hole size at the same value.

# 2.3 Breaking of a dam

In this tutorial we shall solve a problem of simplified dam break in 2 dimensions using the interFoam. The feature of the problem is a transient flow of two fluids separated by a sharp interface, or free surface. The two-phase algorithm in interFoam is based on the volume of fluid (VOF) method in which a specie transport equation is used to determine the relative volume fraction of the two phases, or phase fraction  $\alpha_1$ , in each computational cell. Physical properties are calculated as weighted averages based on this fraction. The nature of the VOF method means that an interface between the species is not explicitly computed, but rather emerges as a property of the phase fraction field. Since the phase fraction can have any value between 0 and 1, the interface is never sharply defined, but occupies a volume around the region where a sharp interface should exist.

The test setup consists of a column of water at rest located behind a membrane on the left side of a tank. At time t = 0 s, the membrane is removed and the column of water collapses. During the collapse, the water impacts an obstacle at the bottom of the tank and creates a complicated flow structure, including several captured pockets of air. The geometry and the initial setup is shown in Figure 2.20.



Figure 2.20: Geometry of the dam break.

### 2.3.1 Mesh generation

The user should go to the *damBreak* case in their *\$FOAM\_RUN/tutorials/multiphase/inter-Foam/laminar* directory. Generate the mesh running blockMesh as described previously. The damBreak mesh consist of 5 blocks; the *blockMeshDict* entries are given below.

```
convertToMeters 0.146;
17
18
      vertices
19
20
      (
            21
22
23
24
25
26
27
28
29
30
31
            32
33
34
35
36
37
38
39
            \begin{array}{c} (2.10438 \ 0.32876 \ 0.1) \\ (4 \ 0.32876 \ 0.1) \\ (0 \ 4 \ 0.1) \\ (2 \ 4 \ 0.1) \\ (2.16438 \ 4 \ 0.1) \end{array}
40
41
42
43
            (4 \ 4 \ 0.1)
44
      );
45
46
      blocks
47
48
            hex (0 1 5 4 12 13 17 16) (23 8 1) simpleGrading (1 1 1)
49
            hex (2 3 7 6 14 15 19 18) (19 8 1) simpleGrading (1 1 1)
50
            hex (4 5 9 8 16 17 21 20) (23 42 1) simpleGrading (1 1 1)
51
            hex (5 6 10 9 17 18 22 21) (4 42 1) simpleGrading (1 1 1)
52
            hex (6 7 11 10 18 19 23 22) (19 42 1) simpleGrading (1 1 1)
53
      );
54
55
      edges
56
57
      (
      );
58
 59
      boundary
60
61
      (
            leftWall
62
            {
63
                  type wall;
64
                  faces
65
66
                  (
                       (0 12 16 4)
(4 16 20 8)
67
68
                  );
69
            }
70
            rightWall
71
            {
72
73
                  type wall;
74
                  faces
                  (
75
                       (7 19 15 3)
(11 23 19 7)
76
77
                 );
78
79
            ĺowerWall
{
80
81
                  type wall;
82
                 faces
83
                  (
84
                       (0 1 13 12)
(1 5 17 13)
(5 6 18 17)
(2 14 18 6)
85
86
87
                       (2 \ 14 \ 18 \ 6)
(2 \ 3 \ 15 \ 14)
88
89
                 );
90
            }
91
            atmosphere
92
93
            {
 94
                  type patch;
 95
                  faces
 96
                  (
                        (8 20 21 9)
(9 21 22 10)
97
98
                        (10 22 23 11)
99
                 );
100
            }
101
```

### 2.3.2 Boundary conditions

The user can examine the boundary geometry generated by blockMesh by viewing the boundary file in the constant/polyMesh directory. The file contains a list of 5 boundary patches: leftWall, rightWall, lowerWall, atmosphere and defaultFaces. The user should notice the type of the patches. The atmosphere is a standard patch, *i.e.* has no special attributes, merely an entity on which boundary conditions can be specified. The defaultFaces patch is empty since the patch normal is in the direction we will not solve in this 2D case. The leftWall, rightWall and lowerWall patches are each a wall. Like the plain patch, the wall type contains no geometric or topological information about the mesh and only differs from the plain patch in that it identifies the patch as a wall, should an application need to know, *e.g.* to apply special wall surface modelling.

A good example is that the interFoam solver includes modelling of surface tension at the contact point between the interface and wall surface. The models are applied by specifying the alphaContactAngle boundary condition on the alpha1 ( $\alpha_1$ ) field. With it, the user must specify the following: a static contact angle, theta0  $\theta_0$ ; leading and trailing edge dynamic contact angles, thetaA  $\theta_A$  and thetaR  $\theta_R$  respectively; and a velocity scaling function for dynamic contact angle, uTheta.

In this tutorial we would like to ignore surface tension effects between the wall and interface. We can do this by setting the static contact angle,  $\theta_0 = 90^\circ$  and the velocity scaling function to 0. However, the simpler option which we shall choose here is to specify a zeroGradient type on alpha1, rather than use the alphaContactAngle boundary condition.

The top boundary is free to the atmosphere so needs to permit both outflow and inflow according to the internal flow. We therefore use a combination of boundary conditions for pressure and velocity that does this while maintaining stability. They are:

- totalPressure which is a fixedValue condition calculated from specified total pressure p0 and local velocity U;
- pressureInletOutletVelocity, which applies zeroGradient on all components, except where there is inflow, in which case a fixedValue condition is applied to the *tangential* component;
- inletOutlet, which is a zeroGradient condition when flow outwards, fixedValue when flow is inwards.

At all wall boundaries, the **buoyantPressure** boundary condition is applied to the pressure field, which calculates the normal gradient from the local density gradient.

The defaultFaces patch representing the front and back planes of the 2D problem, is, as usual, an empty type.

### 2.3.3 Setting initial field

Unlike the previous cases, we shall now specify a non-uniform initial condition for the phase fraction  $\alpha_1$  where

$$\alpha_1 = \begin{cases} 1 & \text{for the liquid phase} \\ 0 & \text{for the gas phase} \end{cases}$$
(2.15)

This will be done by running the setFields utility. It requires a *setFieldsDict* dictionary, located in the *system* directory, whose entries for this case are shown below.

```
17
    defaultFieldValues
18
19
        volScalarFieldValue alpha1 0
20
    );
21
22
    regions
23
24
    (
25
        boxToCell
26
            box (0 0 -1) (0.1461 0.292 1);
27
            fieldValues
28
29
                 volScalarFieldValue alpha1 1
30
            );
31
        }
32
    );
33
34
35
                              *****
36
```

The defaultFieldValues sets the default value of the fields, *i.e.* the value the field takes unless specified otherwise in the regions sub-dictionary. That sub-dictionary contains a list of subdictionaries containing fieldValues that override the defaults in a specified region. The region is expressed in terms of a topoSetSource that creates a set of points, cells or faces based on some topological constraint. Here, boxToCell creates a bounding box within a vector minimum and maximum to define the set of cells of the liquid region. The phase fraction  $\alpha_1$  is defined as 1 in this region.

The setFields utility reads fields from file and, after re-calculating those fields, will write them back to file. Because the files are then overridden, it is recommended that a backup is made before setFields is executed. In the damBreak tutorial, the alpha1 field is initially stored as a backup *only*, named alpha1.org. Before running setFields, the user first needs to copy alpha1.org to alpha1, *e.g.* by typing:

```
cp 0/alpha1.org 0/alpha1
```

The user should then execute setFields as any other utility is executed. Using paraFoam, check that the initial alpha1 field corresponds to the desired distribution as in Figure 2.21.



Figure 2.21: Initial conditions for phase fraction alpha1.

### 2.3.4 Fluid properties

Let us examine the *transportProperties* file in the *constant* directory. The dictionary contains the material properties for each fluid, separated into two subdictionaries *phase1* and *phase2*. The transport model for each phase is selected by the transportModel keyword. The user should select Newtonian in which case the kinematic viscosity is single valued and specified under the keyword nu. The viscosity parameters for the other models, *e.g.*CrossPowerLaw, are specified within subdictionaries with the generic name <*model*>*Coeffs*, *i.e.CrossPowerLawCoeffs* in this example. The density is specified under the keyword rho.

The surface tension between the two phases is specified under the keyword sigma. The values used in this tutorial are listed in Table 2.3.

phase1 properties					
Kinematic viscosity	$\mathrm{m}^2\mathrm{s}^{-1}$	nu	$1.0 \times 10^{-6}$		
Density	${\rm kg}{\rm m}^{-3}$	rho	$1.0  imes 10^3$		
phase2 properties					
Kinematic viscosity	$\mathrm{m}^2\mathrm{s}^{-1}$	nu	$1.48 \times 10^{-5}$		
Density	${\rm kg}{\rm m}^{-3}$	rho	1.0		
Properties of both phases					
Surface tension	${ m N}{ m m}^{-1}$	sigma	0.07		

Table 2.3: Fluid properties for the damBreak tutorial

Gravitational acceleration is uniform across the domain and is specified in a file named g in the *constant* directory. Unlike a normal field file, *e.g.* U and p, g is a uniformDimensionedVectorField and so simply contains a set of dimensions and a value that represents  $(0, 9.81, 0) \text{ m s}^{-2}$  for this tutorial:

### 2.3.5 Turbulence modelling

As in the cavity example, the choice of turbulence modelling method is selectable at runtime through the simulationType keyword in *turbulenceProperties* dictionary. In this example, we wish to run without turbulence modelling so we set laminar:

### 2.3.6 Time step control

Time step control is an important issue in free surface tracking since the surface-tracking algorithm is considerably more sensitive to the Courant number Co than in standard fluid flow calculations. Ideally, we should not exceed an upper limit  $Co \approx 0.5$  in the region of the interface. In some cases, where the propagation velocity is easy to predict, the

user should specify a fixed time-step to satisfy the *Co* criterion. For more complex cases, this is considerably more difficult. interFoam therefore offers automatic adjustment of the time step as standard in the *controlDict*. The user should specify adjustTimeStep to be on and the the maximum *Co* for the phase fields, maxAlphaCo, and other fields, maxCo, to be 0.5. The upper limit on time step maxDeltaT can be set to a value that will not be exceeded in this simulation, *e.g.* 1.0.

By using automatic time step control, the steps themselves are never rounded to a convenient value. Consequently if we request that OpenFOAM saves results at a fixed number of time step intervals, the times at which results are saved are somewhat arbitrary. However even with automatic time step adjustment, OpenFOAM allows the user to specify that results are written at fixed times; in this case OpenFOAM forces the automatic time stepping procedure to adjust time steps so that it 'hits' on the exact times specified for write output. The user selects this with the adjustableRunTime option for writeControl in the controlDict dictionary. The controlDict dictionary entries should be:

```
17
    application
                    interFoam;
18
19
    startFrom
                    startTime;
20
21
22
    startTime
                    0;
23
    stopAt
                    endTime;
24
25
26
    endTime
                    1;
27
28
    deltaT
                    0.001;
29
30
    writeControl
                    adjustableRunTime;
31
    writeInterval
                    0.05;
32
33
34
    purgeWrite
                    0;
35
    writeFormat
                    ascii;
36
37
38
    writePrecision
                   6;
39
    writeCompression uncompressed;
40
41
    timeFormat
                    general;
42
43
    timePrecision
                    6;
44
45
    runTimeModifiable yes;
46
47
    adjustTimeStep yes;
48
49
    maxCo
                    0.5;
0.5:
50
    maxAlphaCo
51
52
    maxDeltaT
                    1;
53
54
55
    56
```

## 2.3.7 Discretisation schemes

The interFoam solver uses the multidimensional universal limiter for explicit solution (MULES) method, created by OpenCFD, to maintain boundedness of the phase fraction independent of underlying numerical scheme, mesh structure, *etc.* The choice of schemes for convection are therfore not restricted to those that are strongly stable or bounded, *e.g.* upwind differencing.

The convection schemes settings are made in the *divSchemes* sub-dictionary of the *fvSchemes* dictionary. In this example, the convection term in the momentum equation  $(\nabla \cdot (\rho UU))$ , denoted by the div(rho\*phi,U) keyword, uses Gauss limitedLinearV 1.0 to produce good accuracy. The limited linear schemes require a coefficient  $\phi$  as described in section 4.4.1. Here, we have opted for best stability with  $\phi = 1.0$ . The

 $\nabla \cdot (\mathbf{U}\alpha_1)$  term, represented by the div(phi,alpha) keyword uses the vanLeer scheme. The  $\nabla \cdot (\mathbf{U}_{rb}\alpha_1)$  term, represented by the div(phirb,alpha) keyword, can similarly use the vanLeer scheme, but generally produces smoother interfaces using the specialised interfaceCompression scheme.

The other discretised terms use commonly employed schemes so that the fvSchemes dictionary entries should therefore be:

```
17
    ddtSchemes
18
19
    {
        default
                        Euler;
20
    }
^{21}
22
    gradSchemes
23
24
        default
                        Gauss linear;
25
    }
26
27
    divSchemes
28
29
        div(rho*phi,U)
                        Gauss limitedLinearV 1;
30
31
        div(phi,alpha) Gauss vanLeer;
        div(phirb,alpha) Gauss interfaceCompression;
32
    }
33
34
    laplacianSchemes
35
36
    {
                        Gauss linear corrected;
        default
37
    }
38
39
    interpolationSchemes
40
41
    {
        default
                        linear;
42
    }
43
44
    snGradSchemes
45
46
    ſ
47
        default
                        corrected;
    }
48
49
50
    fluxRequired
51
    Ł
        default
                        no:
52
53
        p_rgh;
54
        pcorr
        alpha1;
55
    }
56
57
58
```

### 2.3.8 Linear-solver control

In the *fvSolution*, the *PISO* sub-dictionary contains elements that are specific to interFoam. There are the usual correctors to the momentum equation but also correctors to a PISO loop around the  $\alpha_1$  phase equation. Of particular interest are the nAlphaSubCycles and cAlpha keywords. nAlphaSubCycles represents the number of sub-cycles within the  $\alpha_1$  equation; sub-cycles are additional solutions to an equation within a given time step. It is used to enable the solution to be stable without reducing the time step and vastly increasing the solution time. Here we specify 2 sub-cycles, which means that the  $\alpha_1$  equation is solved in 2× half length time steps within each actual time step.

The cAlpha keyword is a factor that controls the compression of the interface where: 0 corresponds to no compression; 1 corresponds to conservative compression; and, anything larger than 1, relates to enhanced compression of the interface. We generally recommend a value of 1.0 which is employed in this example.

### 2.3.9 Running the code

Running of the code has been described in detail in previous tutorials. Try the following, that uses **tee**, a command that enables output to be written to both standard output and files:

cd \$FOAM\_RUN/tutorials/multiphase/interFoam/laminar/damBreak interFoam | tee log

The code will now be run interactively, with a copy of output stored in the log file.

### 2.3.10 Post-processing

Post-processing of the results can now be done in the usual way. The user can monitor the development of the phase fraction alpha1 in time, e.g. see Figure 2.22.

### 2.3.11 Running in parallel

The results from the previous example are generated using a fairly coarse mesh. We now wish to increase the mesh resolution and re-run the case. The new case will typically take a few hours to run with a single processor so, should the user have access to multiple processors, we can demonstrate the parallel processing capability of OpenFOAM.

The user should first make a copy of the damBreak case, e.g. by

```
cd $FOAM_RUN/tutorials/multiphase/interFoam/laminar
mkdir damBreakFine
cp -r damBreak/0 damBreakFine
cp -r damBreak/system damBreakFine
cp -r damBreak/constant damBreakFine
```

Enter the new case directory and change the  $\verb+blocks+$  description in the blockMeshDict dictionary to

Here, the entry is presented as printed from the *blockMeshDict* file; in short the user must change the mesh densities, *e.g.* the 46 10 1 entry, and some of the mesh grading entries to 1 2 1. Once the dictionary is correct, generate the mesh.

As the mesh has now changed from the damBreak example, the user must re-initialise the phase field alpha1 in the 0 time directory since it contains a number of elements that is inconsistent with the new mesh. Note that there is no need to change the U and p\_rgh fields since they are specified as uniform which is independent of the number of elements in the field. We wish to initialise the field with a sharp interface, *i.e.* it elements would have  $\alpha_1 = 1$  or  $\alpha_1 = 0$ . Updating the field with mapFields may produce interpolated



Figure 2.22: Snapshots of phase  $\alpha_1$ .

values  $0 < \alpha_1 < 1$  at the interface, so it is better to rerun the setFields utility. There is a backup copy of the initial uniform  $\alpha_1$  field named 0/alpha1.org that the user should copy to 0/alpha1 before running setFields:

cd \$FOAM\_RUN/tutorials/multiphase/interFoam/laminar/damBreakFine cp -r 0/alpha1.org 0/alpha1 setFields

The method of parallel computing used by OpenFOAM is known as domain decomposition, in which the geometry and associated fields are broken into pieces and allocated to separate processors for solution. The first step required to run a parallel case is therefore to decompose the domain using the decomposePar utility. There is a dictionary associated with decomposePar named decomposeParDict which is located in the system directory of the tutorial case; also, like with many utilities, a default dictionary can be found in the directory of the source code of the specific utility, *i.e.* in *\$FOAM\_UTILITIES/parallelProcessing/decomposePar* for this case.

The first entry is numberOfSubdomains which specifies the number of subdomains into which the case will be decomposed, usually corresponding to the number of processors available for the case.

In this tutorial, the method of decomposition should be simple and the corresponding simpleCoeffs should be edited according to the following criteria. The domain is split into pieces, or subdomains, in the x, y and z directions, the number of subdomains in each direction being given by the vector  $\mathbf{n}$ . As this geometry is 2 dimensional, the 3rd direction, z, cannot be split, hence  $n_z$  must equal 1. The  $n_x$  and  $n_y$  components of  $\mathbf{n}$  split the domain in the x and y directions and must be specified so that the number of subdomains specified by  $n_x$  and  $n_y$  equals the specified numberOfSubdomains, *i.e.*  $n_x n_y =$ numberOfSubdomains. It is beneficial to keep the number of cell faces adjoining the subdomains to a minimum so, for a square geometry, it is best to keep the split between the x and y directions should be fairly even. The delta keyword should be set to 0.001.

For example, let us assume we wish to run on 4 processors. We would set number-OfSubdomains to 4 and n = (2, 2, 1). When running decomposePar, we can see from the screen messages that the decomposition is distributed fairly even between the processors.

The user should consult section 3.4 for details of how to run a case in parallel; in this tutorial we merely present an example of running in parallel. We use the openMPI implementation of the standard message-passing interface (MPI). As a test here, the user can run in parallel on a single node, the local host only, by typing:

#### mpirun -np 4 interFoam -parallel > log &

The user may run on more nodes over a network by creating a file that lists the host names of the machines on which the case is to be run as described in section 3.4.2. The case should run in the background and the user can follow its progress by monitoring the *log* file as usual.

### 2.3.12 Post-processing a case run in parallel

Once the case has completed running, the decomposed fields and mesh must be reassembled for post-processing using the reconstructPar utility. Simply execute it from the command line. The results from the fine mesh are shown in Figure 2.24. The user can see that the resolution of interface has improved significantly compared to the coarse mesh.



Figure 2.23: Mesh of processor 2 in parallel processed case.

The user may also post-process a segment of the decomposed domain individually by simply treating the individual processor directory as a case in its own right. For example if the user starts paraFoam by

#### paraFoam -case processor1

then processor1 will appear as a case module in ParaView. Figure 2.23 shows the mesh from processor 1 following the decomposition of the domain using the simple method.



Figure 2.24: Snapshots of phase  $\alpha_1$  with refined mesh.

# Chapter 3

# **Applications and libraries**

We should reiterate from the outset that OpenFOAM is a C++ library used primarily to create executables, known as *applications*. OpenFOAM is distributed with a large set of precompiled applications but users also have the freedom to create their own or modify existing ones. Applications are split into two main categories:

- **solvers** that are each designed to solve a specific problem in computational continuum mechanics;
- utilities that perform simple pre-and post-processing tasks, mainly involving data manipulation and algebraic calculations.

OpenFOAM is divided into a set of precompiled libraries that are dynamically linked during compilation of the solvers and utilities. Libraries such as those for physical models are supplied as source code so that users may conveniently add their own models to the libraries. This chapter gives an overview of solvers, utilities and libraries, their creation, modification, compilation and execution.

### 3.1 The programming language of OpenFOAM

In order to understand the way in which the OpenFOAM library works, some background knowledge of C++, the base language of OpenFOAM, is required; the necessary information will be presented in this chapter. Before doing so, it is worthwhile addressing the concept of language in general terms to explain some of the ideas behind object-oriented programming and our choice of C++ as the main programming language of OpenFOAM.

#### 3.1.1 Language in general

The success of verbal language and mathematics is based on efficiency, especially in expressing abstract concepts. For example, in fluid flow, we use the term "velocity field", which has meaning without any reference to the nature of the flow or any specific velocity data. The term encapsulates the idea of movement with direction and magnitude and relates to other physical properties. In mathematics, we can represent velocity field by a single symbol, *e.g.* U, and express certain concepts using symbols, *e.g.* "the field of velocity magnitude" by  $|\mathbf{U}|$ . The advantage of mathematics over verbal language is its greater efficiency, making it possible to express complex concepts with extreme clarity.

The problems that we wish to solve in continuum mechanics are not presented in terms of intrinsic entities, or types, known to a computer, *e.g.* bits, bytes, integers. They are usually presented first in verbal language, then as partial differential equations in 3

dimensions of space and time. The equations contain the following concepts: scalars, vectors, tensors, and fields thereof; tensor algebra; tensor calculus; dimensional units. The solution to these equations involves discretisation procedures, matrices, solvers, and solution algorithms.

### 3.1.2 Object-orientation and C++

Progamming languages that are object-oriented, such as C++, provide the mechanism — *classes* — to declare types and associated operations that are part of the verbal and mathematical languages used in science and engineering. Our velocity field introduced earlier can be represented in programming code by the symbol U and "the field of velocity magnitude" can be mag(U). The velocity is a vector field for which there should exist, in an object-oriented code, a vectorField class. The velocity field U would then be an instance, or *object*, of the vectorField class; hence the term object-oriented.

The clarity of having objects in programming that represent physical objects and abstract entities should not be underestimated. The class structure concentrates code development to contained regions of the code, *i.e.* the classes themselves, thereby making the code easier to manage. New classes can be derived or inherit properties from other classes, *e.g.* the vectorField can be derived from a vector class and a Field class. C++ provides the mechanism of *template classes* such that the template class Field<Type> can represent a field of any <Type>, *e.g.* scalar, vector, tensor. The general features of the template class are passed on to any class created from the template. Templating and inheritance reduce duplication of code and create class hierarchies that impose an overall structure on the code.

### 3.1.3 Equation representation

A central theme of the OpenFOAM design is that the solver applications, written using the OpenFOAM classes, have a syntax that closely resembles the partial differential equations being solved. For example the equation

$$\frac{\partial \rho \mathbf{U}}{\partial t} + \nabla \boldsymbol{\cdot} \boldsymbol{\phi} \mathbf{U} - \nabla \boldsymbol{\cdot} \boldsymbol{\mu} \nabla \mathbf{U} = -\nabla p$$

is represented by the code

```
solve
(
    fvm::ddt(rho, U)
    + fvm::div(phi, U)
    - fvm::laplacian(mu, U)
    ==
    - fvc::grad(p)
);
```

This and other requirements demand that the principal programming language of Open-FOAM has object-oriented features such as inheritance, template classes, virtual functions and operator overloading. These features are not available in many languages that purport to be object-orientated but actually have very limited object-orientated capability, such as FORTRAN-90. C++, however, possesses all these features while having the additional advantage that it is widely used with a standard specification so that reliable compilers are available that produce efficient executables. It is therefore the primary language of OpenFOAM.

### 3.1.4 Solver codes

Solver codes are largely procedural since they are a close representation of solution algorithms and equations, which are themselves procedural in nature. Users do not need a deep knowledge of object-orientation and C++ programming to write a solver but should know the principles behind object-orientation and classes, and to have a basic knowledge of some C++ code syntax. An understanding of the underlying equations, models and solution method and algorithms is far more important.

There is often little need for a user to immerse themselves in the code of any of the OpenFOAM classes. The essence of object-orientation is that the user should not have to; merely the knowledge of the class' existence and its functionality are sufficient to use the class. A description of each class, its functions *etc.* is supplied with the OpenFOAM distribution in HTML documentation generated with Doxygen at \$WM\_PROJECT\_DIR/-doc/Doxygen/html/index.html.

## 3.2 Compiling applications and libraries

Compilation is an integral part of application development that requires careful management since every piece of code requires its own set instructions to access dependent components of the OpenFOAM library. In UNIX/Linux systems these instructions are often organised and delivered to the compiler using the standard UNIXmake utility. Open-FOAM, however, is supplied with the wmake compilation script that is based on make but is considerably more versatile and easier to use; wmake can, in fact, be used on any code, not simply the OpenFOAM library. To understand the compilation process, we first need to explain certain aspects of C++ and its file structure, shown schematically in Figure 3.1. A class is defined through a set of instructions such as object construction, data storage and class member functions. The file containing the class definition takes a .C extension, e.g. a class nc would be written in the file nc.C. This file can be compiled independently of other code into a binary executable library file known as a shared object library with the .so file extension, *i.e.nc.so*. When compiling a piece of code, say newApp.C, that uses the nc class, nc.C need not be recompiled, rather newApp.C calls nc.so at runtime. This is known as dynamic linking.



Figure 3.1: Header files, source files, compilation and linking

### 3.2.1 Header .H files

As a means of checking errors, the piece of code being compiled must know that the classes it uses and the operations they perform actually exist. Therefore each class requires a class *declaration*, contained in a header file with a *.H* file extension, *e.g.nc.H*, that includes the names of the class and its functions. This file is included at the beginning of any piece of code using the class, including the class declaration code itself. Any piece of *.C* code can resource any number of classes and must begin with all the *.H* files required to declare these classes. The classes in turn can resource other classes and begin with the relevant *.H* files. By searching recursively down the class hierarchy we can produce a complete list of header files for all the classes on which the top level *.C* code ultimately depends; these *.H* files are known as the *dependencies*. With a dependency list, a compiler can check whether the source files have been updated since their last compilation and selectively compile only those that need to be.

Header files are included in the code using **#** include statements, *e.g.* 

```
# include "otherHeader.H";
```

causes the compiler to suspend reading from the current file to read the file specified. Any self-contained piece of code can be put into a header file and included at the relevant location in the main code in order to improve code readability. For example, in most OpenFOAM applications the code for creating fields and reading field input data is included in a file *createFields.H* which is called at the beginning of the code. In this way, header files are not solely used as class declarations. It is **wmake** that performs the task of maintaining file dependency lists amongst other functions listed below.

- Automatic generation and maintenance of file dependency lists, *i.e.* lists of files which are included in the source files and hence on which they depend.
- Multi-platform compilation and linkage, handled through appropriate directory structure.
- Multi-language compilation and linkage, e.g. C, C++, Java.
- Multi-option compilation and linkage, *e.g.* debug, optimised, parallel and profiling.
- Support for source code generation programs, *e.g.* lex, yacc, IDL, MOC.
- Simple syntax for source file lists.
- Automatic creation of source file lists for new codes.
- Simple handling of multiple shared or static libraries.
- Extensible to new machine types.
- Extremely portable, works on any machine with: make; sh, ksh or csh; lex, cc.
- Has been tested on Apollo, SUN, SGI, HP (HPUX), Compaq (DEC), IBM (AIX), Cray, Ardent, Stardent, PC Linux, PPC Linux, NEC, SX4, Fujitsu VP1000.
# 3.2.2 Compiling with wmake

OpenFOAM applications are organised using a standard convention that the source code of each application is placed in a directory whose name is that of the application. The top level source file takes the application name with the .C extension. For example, the source code for an application called newApp would reside is a directory newApp and the top level file would be newApp.C as shown in Figure 3.2. The directory must also contain



Figure 3.2: Directory structure for an application

a  ${\it Make}$  subdirectory containing 2 files,  ${\it options}$  and  ${\it files},$  that are described in the following sections.

## 3.2.2.1 Including headers

The compiler searches for the included header files in the following order, specified with the  $\neg I$  option in <code>wmake</code>:

- 1. the **\$WM\_PROJECT\_DIR/src/OpenFOAM/InInclude** directory;
- 2. a local *InInclude* directory, *i.e.newApp/InInclude*;
- 3. the local directory, *i.e.newApp*;
- 4. platform dependent paths set in files in the \$WM\_PROJECT\_DIR/wmake/rules/-\$WM\_ARCH/ directory, e.g./usr/X11/include and \$(MPICH\_ARCH\_PATH)/include;
- 5. other directories specified explicitly in the *Make/options* file with the -I option.

The  ${\it Make/options}$  file contains the full directory paths to locate header files using the syntax:

```
EXE_INC = \
    -I<directoryPath1> \
    -I<directoryPath2> \
    ... \
    -I<directoryPathN>
```

Notice first that the directory names are preceded by the -I flag and that the syntax uses the  $\$  to continue the EXE\_INC across several lines, with no  $\$  after the final entry.

## 3.2.2.2 Linking to libraries

The compiler links to shared object library files in the following directory paths, specified with the -L option in wmake:

- 1. the **\$FOAM\_LIBBIN** directory;
- 2. platform dependent paths set in files in the \$WM\_DIR/rules/\$WM\_ARCH/ directory, e.g./usr/X11/lib and \$(MPICH\_ARCH\_PATH)/lib;
- 3. other directories specified in the *Make/options* file.

The actual library **files** to be linked must be specified using the **-1** option and removing the **lib** prefix and **.so** extension from the library file name, *e.g.*libnew.so is included with the flag **-lnew**. By default, wmake loads the following libraries:

- 1. the libOpenFOAM.so library from the *\$FOAM\_LIBBIN* directory;
- 2. platform dependent libraries specified in set in files in the \$WM\_DIR/rules/\$WM\_ARCH/ directory, e.g.libm.so from /usr/X11/lib and liblam.so from \$(LAM\_ARCH\_PATH)/lib;
- 3. other libraries specified in the *Make/options* file.

The *Make/options* file contains the full directory paths and library names using the syntax:

```
EXE_LIBS = \
    -L<libraryPath1> \
    -L<libraryPath2> \
    ... \
    -L<libraryPathN> \
    -l<library1> \
    ... \
    -l<library2> \
    ... \
    -l<libraryN>
```

Let us reiterate that the directory paths are preceeded by the -L flag, the library names are preceeded by the -l flag.

## 3.2.2.3 Source files to be compiled

The compiler requires a list of .C source files that must be compiled. The list must contain the main .C file but also any other source files that are created for the specific application but are not included in a class library. For example, users may create a new class or some new functionality to an existing class for a particular application. The full list of .C source files must be included in the *Make/files* file. As might be expected, for many applications the list only includes the name of the main .C file, *e.g.newApp.C* in the case of our earlier example.

The Make/files file also includes a full path and name of the compiled executable, specified by the EXE = syntax. Standard convention stipulates the name is that of the application, *i.e.*newApp in our example. The OpenFOAM release offers two useful choices for path: standard release applications are stored in *\$FOAM\_APPBIN*; applications developed by the user are stored in *\$FOAM\_USER\_APPBIN*.

If the user is developing their own applications, we recommend they create an applications subdirectory in their  $\mbox{\it SWM\_PROJECT\_USER\_DIR}$  directory containing the source

code for personal OpenFOAM applications. As with standard applications, the source code for each OpenFOAM application should be stored within its own directory. The only difference between a user application and one from the standard release is that the Make/files file should specify that the user's executables are written into their  $FOAM_-USER_APPBIN$  directory. The Make/files file for our example would appear as follows:

newApp.C

EXE = \$(FOAM\_USER\_APPBIN)/newApp

## 3.2.2.4 Running wmake

The wmake script is executed by typing:

wmake <optionalArguments> <optionalDirectory>

The <optionalDirectory> is the directory path of the application that is being compiled. Typically, wmake is executed from within the directory of the application being compiled, in which case <optionalDirectory> can be omitted.

If a user wishes to build an application executable, then no <optionalArguments> are required. However <optionalArguments> may be specified for building libraries *etc.* as described in Table 3.1.

Argument	Type of compilation
lib	Build a statically-linked library
libso	Build a dynamically-linked library
libo	Build a statically-linked object file library
jar	Build a JAVA archive
exe	Build an application independent of the specified project

Table 3.1: Optional compilation arguments to wmake.

## 3.2.2.5 wmake environment variables

For information, the environment variable settings used by wmake are listed in Table 3.2.

# 3.2.3 Removing dependency lists: wclean and rmdepall

On execution, wmake builds a dependency list file with a .dep file extension, e.g.newApp.dep in our example, and a list of files in a Make/\$WM\_OPTIONS directory. If the user wishes to remove these files, perhaps after making code changes, the user can run the wclean script by typing:

```
wclean <optionalArguments> <optionalDirectory>
```

Again, the <optionalDirectory> is a path to the directory of the application that is being compiled. Typically, wclean is executed from within the directory of the application, in which case the path can be omitted.

Main paths					
\$WM_PROJECT_INST_DIR	Full	path	to	installation	directory,
	e.g.\$HON	ME/Open	FOAM		
\$WM_PROJECT	Name of	the proje	ect being	compiled: OpenH	MAO
\$WM_PROJECT_VERSION	Version c	of the pro	ject bein	g compiled: 2.1	. 0
\$WM_PROJECT_DIR	Full path	n to loca	te binary	v executables of	OpenFOAM
	release, $e$	e.g. <b>\$HOM</b>	E/OpenF	OAM/OpenFOA	M-2.1.0
\$WM_PROJECT_USER_DIR	Full pat	h to loc	ate bina	ary executables	of the user
	e.g.\$HON	ME/Open	FOAM/\$	{USER}-2.1.0	
Other paths/settings					
\$WM_ARCH	Machine	architect	ure: Lin	ux, SunOS	
\$WM_ARCH_OPTION	32 or 64			7	
\$WM_COMPILER	Compiler	being us	sed: Gcc4	43 - gcc 4.4.3, IC	C - Intel
\$WM_COMPILER_DIR	Compiler	-		•	
\$WM_COMPILER_BIN	-			ries <b>\$WM_COMP</b>	LER_BIN/bin
\$WM_COMPILER_LIB	-			ries <b>\$WM_COMP</b>	,
\$WM_COMPILE_OPTION	-			g - debugging, O	,
	tion.	I		0 00 0	
\$WM_DIR	Full path	of the w	<i>make</i> dir	ectory	
\$WM_MPLIB	-			brary: LAM, MPI,	MPICH, PVM
\$WM_OPTIONS	$=$ \$WM_			· · · ·	,
		.\$WM_CO	OMPILE_	OPTION\$WM_M	PLIB
	e.g.linux	kGcc30pt	MPICH		
\$WM_PRECISION_OPTION	0	-		oinares, SP, single	e precision or
	DP, doub	le precisio	on		

Table 3.2: Environment variable settings for wmake.

If a user wishes to remove the dependency files and files from the *Make* directory, then no <optionalArguments> are required. However if lib is specified in <optionalArguments> a local *lnlnclude* directory will be deleted also.

An additional script, **rmdepall** removes all dependency *.dep* files recursively down the directory tree from the point at which it is executed. This can be useful when updating OpenFOAM libraries.

# 3.2.4 Compilation example: the pisoFoam application

The source code for application pisoFoam is in the *\$FOAM\_APP/solvers/incompressible/pisoFoam* directory and the top level source file is named *pisoFoam.C*. The *pisoFoam.C* source code is:

```
----*\
1
        =========
2
                                             OpenFOAM: The Open Source CFD Toolbox
                      F ield
3
                      0 peration
4
                                           | Copyright (C) 2011 OpenFOAM Foundation
5
                      A nd
                      M anipulation
6
            \langle \rangle
                                           7
     License
8
          This file is part of OpenFOAM.
9
10
          OpenFOAM is free software: you can redistribute it and/or modify it under the terms of the GNU General Public License as published by
11
12
```

the Free Software Foundation, either version 3 of the License, or (at your option) any later version. OpenFOAM is distributed in the hope that it will be useful, but WITHOUT ANY WARRANTY; without even the implied warranty of MERCHANTABILITY or FITNESS FOR A PARTICULAR PURPOSE. See the GNU General Public License for more details. You should have received a copy of the GNU General Public License along with OpenFOAM. If not, see <http://www.gnu.org/licenses/>. Application pisoFoam Description Transient solver for incompressible flow. Turbulence modelling is generic, i.e. laminar, RAS or LES may be selected. \\*-----\*/ #include "fvCFD.H"
#include "singlePhaseTransportModel.H" #include "turbulenceModel.H" int main(int argc, char \*argv[]) ł #include "setRootCase.H" #include "createTime.H" #include "createMesh.H"
#include "createFields.H" #include "initContinuityErrs.H" // \* // Info<< "\nStarting time loop\n" << endl;</pre> while (runTime.loop()) Info<< "Time = " << runTime.timeName() << nl << endl;</pre> #include "readPISOControls.H"
#include "CourantNo.H" // Pressure-velocity PISO corrector { // Momentum predictor fvVectorMatrix UEqn fvm::ddt(U) + fvm::div(phi, U) + turbulence->divDevReff(U) ): UEqn.relax(); if (momentumPredictor) { solve(UEqn == -fvc::grad(p)); } // --- PISO loop for (int corr=0; corr<nCorr; corr++)</pre> volScalarField rAU(1.0/UEqn.A()); U = rAU \* UEqn.H();phi = (fvc::interpolate(U) & mesh.Sf()) + fvc::ddtPhiCorr(rAU, U, phi); adjustPhi(phi, U, p); // Non-orthogonal pressure corrector loop for (int nonOrth=0; nonOrth<=nNonOrthCorr; nonOrth++)</pre> // Pressure corrector fvScalarMatrix pEqn (

```
fvm::laplacian(rAU, p) == fvc::div(phi)
97
                          );
98
99
                          pEqn.setReference(pRefCell, pRefValue);
100
101
                          if
102
                           (
103
                               corr == nCorr-1
104
                           && nonOrth == nNonOrthCorr
105
106
107
                               pEqn.solve(mesh.solver("pFinal"));
108
                          }
109
                          else
{
110
111
                               pEqn.solve();
112
                          }
113
114
                             (nonOrth == nNonOrthCorr)
                          if
115
116
                               phi -= pEqn.flux();
117
                          }
118
                      }
119
120
                      #include "continuityErrs.H"
121
122
                      U -= rAU*fvc::grad(p);
123
                      U.correctBoundaryConditions();
124
                  }
125
             }
126
127
128
             turbulence->correct();
129
             runTime.write();
130
131
             Info<< "ExecutionTime = " << runTime.elapsedCpuTime() << " s"</pre>
132
                  << " ClockTime = " << runTime.elapsedClockTime() << " s"
133
                  << nl << endl;
134
         }
135
136
         Info<< "End\n" << endl;</pre>
137
138
         return 0;
139
    }
140
141
142
                                  143
```

The code begins with a brief description of the application contained within comments over 1 line (//) and multiple lines (/\*...\*/). Following that, the code contains several **#** include statements, *e.g.***#** include "fvCFD.H", which causes the compiler to suspend reading from the current file, *pisoFoam.C* to read the *fvCFD.H*.

pisoFoam resources the incompressibleRASModels, incompressibleLESModels and incompressibleTransportModels libraries and therefore requires the necessary header files, specified by the EXE\_INC = -I... option, and links to the libraries with the EXE\_LIBS = -1... option. The *Make/options* therefore contains the following:

```
EXE_INC = \setminus
1
         -I$(LIB_SRC)/turbulenceModels/incompressible/turbulenceModel \
2
        -I$(LIB_SRC)/transportModels \
3
        -I$(LIB_SRC)/transportModels/incompressible/singlePhaseTransportModel \
4
        -I$(LIB_SRC)/finiteVolume/lnInclude
5
6
    EXE\_LIBS = \setminus
7
         -lincompressibleTurbulenceModel \
8
9
        -lincompressibleRASModels \
        -lincompressibleLESModels \
10
        -lincompressibleTransportModels \
11
         -lfiniteVolume \
12
        -lmeshTools
13
```

pisoFoam contains only the *pisoFoam*.*C* source and the executable is written to the *\$FOAM\_APPBIN* directory as all standard applications are. The *Make/files* therefore contains:

```
1 pisoFoam.C
```

```
3 EXE = $(FOAM_APPBIN)/pisoFoam
```

The user can compile **pisoFoam** by going to the *\$FOAM\_SOLVERS/incompressible/pisoFoam* directory and typing:

#### wmake

The code should compile and produce a message similar to the following

```
Making dependency list for source file pisoFoam.C
SOURCE_DIR=.
SOURCE=pisoFoam.C ;
g++ -DFOAM_EXCEPTION -Dlinux -DlinuxOptMPICH
-DscalarMachine -DoptSolvers -DPARALLEL -DUSEMPI -Wall -O2 -DNoRepository
-ftemplate-depth-17 -I.../OpenFOAM/OpenFOAM-2.1.0/src/OpenFOAM/lnInclude
-IlnInclude
-I.
.....
-Impich -L/usr/X11/lib -lm
-o .../OpenFOAM/OpenFOAM-2.1.0/applications/bin/linuxOptMPICH/pisoFoam
```

The user can now try recompiling and will receive a message similar to the following to say that the executable is up to date and compiling is not necessary:

```
make: Nothing to be done for `allFiles'.
make: `Make/linuxOptMPICH/dependencies' is up to date.
make: `.../OpenFOAM/OpenFOAM-2.1.0/applications/bin/linuxOptMPICH/pisoFoam'
is up to date.
```

The user can compile the application from scratch by removing the dependency list with

wclean

and running wmake.

## 3.2.5 Debug messaging and optimisation switches

OpenFOAM provides a system of messaging that is written during runtime, most of which are to help debugging problems encountered during running of a OpenFOAM case. The switches are listed in the *\$WM\_PROJECT\_DIR/etc/controlDict* file; should the user wish to change the settings they should make a copy to their *\$HOME* directory, *i.e.\$HOME/.OpenFOAM/2.1.0/controlDict* file. The list of possible switches is extensive and can be viewed by running the foamDebugSwitches application. Most of the switches correspond to a class or range of functionality and can be switched on by their inclusion in the *controlDict* file, and by being set to 1. For example, OpenFOAM can perform the checking of dimensional units in all calculations by setting the dimensionSet switch to 1. There are some switches that control messaging at a higher level than most, listed in Table 3.3.

In addition, there are some switches that control certain operational and optimisation issues. These switches are also listed in Table 3.3. Of particular importance is fileModificationSkew. OpenFOAM scans the write time of data files to check for modification. When running over a NFS with some disparity in the clock settings on different machines, field data files appear to be modified ahead of time. This can cause a problem if OpenFOAM views the files as newly modified and attempting to re-read this data. The fileModificationSkew keyword is the time in seconds that OpenFOAM will subtract from the file write time when assessing whether the file has been newly modified.

High level of	lebugging switches - sub-dictionary DebugSwitches
level	Overall level of debugging messaging for OpenFOAM 3 levels 0,

	• • • • • • • • • • • • • • • • • • •	
	1, 2	
lduMatrix	Messaging for solver convergence during a run - 3 levels 0, 1, 2	

Optimisation switches - sub-dictionary OptimisationSwitches

fileModific-	A time in seconds that should be set higher than the maximum	
ationSkew	delay in NFS updates and clock difference for running OpenFOAM	
	over a NFS.	
fileModific-	Method of checking whether files have been modified during a	
ationChecking	simulation, either reading the timeStamp or using inotify; ver-	
-	sions that read only master-node data exist, timeStampMaster,	
	inotifyMaster.	
commsType	Parallel communications type: nonBlocking, scheduled,	
	blocking.	
floatTransfer	If 1, will compact numbers to float precision before transfer; de-	
	fault is 0	
nProcsSimpleSum	m Optimises global sum for parallel processing; sets number of pro-	
-	cessors above which hierarchical sum is performed rather than a	
	linear sum (default 16)	

Table 3.3: Runtime message switches.

## 3.2.6 Linking new user-defined libraries to existing applications

The situation may arise that a user creates a new library, say **new**, and wishes the features within that library to be available across a range of applications. For example, the user may create a new boundary condition, compiled into **new**, that would need to be recognised by a range of solver applications, pre- and post-processing utilities, mesh tools, *etc.* Under normal circumstances, the user would need to recompile every application with the **new** linked to it.

Instead there is a simple mechanism to link one or more shared object libraries dynamically at run-time in OpenFOAM. Simply add the optional keyword entry libs to the *controlDict* file for a case and enter the full names of the libraries within a list (as quoted string entries). For example, if a user wished to link the libraries new1 and new2 at run-time, they would simply need to add the following to the case *controlDict* file:

```
libs
(
    "libnew1.so"
    "libnew2.so"
);
```

# 3.3 Running applications

Each application is designed to be executed from a terminal command line, typically reading and writing a set of data files associated with a particular case. The data files for a case are stored in a directory named after the case as described in section 4.1; the directory name with full path is here given the generic name < caseDir.

For any application, the form of the command line entry for any can be found by simply entering the application name at the command line with the -help option, *e.g.* typing

```
blockMesh -help
```

returns the usage

```
Usage: blockMesh [-region region name] [-case dir] [-blockTopology]
       [-help] [-doc] [-srcDoc]
```

The arguments in square brackets, [], are optional flags. If the application is executed from within a case directory, it will operate on that case. Alternatively, the -case <caseDir> option allows the case to be specified directly so that the application can be executed from anywhere in the filing system.

Like any UNIX/Linux executable, applications can be run as a background process, *i.e.* one which does not have to be completed before the user can give the shell additional commands. If the user wished to run the blockMesh example as a background process and output the case progress to a *log* file, they could enter:

blockMesh > log &

# 3.4 Running applications in parallel

This section describes how to run OpenFOAM in parallel on distributed processors. The method of parallel computing used by OpenFOAM is known as domain decomposition, in which the geometry and associated fields are broken into pieces and allocated to separate processors for solution. The process of parallel computation involves: decomposition of mesh and fields; running the application in parallel; and, post-processing the decomposed case as described in the following sections. The parallel running uses the public domain openMPI implementation of the standard message passing interface (MPI).

## 3.4.1 Decomposition of mesh and initial field data

The mesh and fields are decomposed using the decomposePar utility. The underlying aim is to break up the domain with minimal effort but in such a way to guarantee a fairly economic solution. The geometry and fields are broken up according to a set of parameters specified in a dictionary named *decomposeParDict* that must be located in the *system* directory of the case of interest. An example *decomposeParDict* dictionary can be copied from the interFoam/damBreak tutorial if the user requires one; the dictionary entries within it are reproduced below:

```
17
     numberOfSubdomains 4;
18
19
    method
                        simple;
20
21
     simpleCoeffs
22
     {
23
                             (221);
0.001;
24
         n
         delta
25
     }
26
27
    hierarchicalCoeffs
28
29
                             (111);
         n
30
```

```
0.001;
31
      delta
32
      order
                   xyz;
   }
33
34
   manualCoeffs
35
36
                   "":
      dataFile
37
   }
38
39
   distributed
                no;
40
41
                ();
42
   roots
43
44
                       45
```

The user has a choice of four methods of decomposition, specified by the **method** keyword as described below.

- simple Simple geometric decomposition in which the domain is split into pieces by direction, *e.g.* 2 pieces in the x direction, 1 in y *etc.*
- hierarchical Hierarchical geometric decomposition which is the same as simple except the user specifies the order in which the directional split is done, e.g. first in the y-direction, then the x-direction etc.
- scotch Scotch decomposition which requires no geometric input from the user and attempts to minimise the number of processor boundaries. The user can specify a weighting for the decomposition between processors, through an optional processorWeights keyword which can be useful on machines with differing performance between processors. There is also an optional keyword entry strategy that controls the decomposition strategy through a complex string supplied to Scotch. For more information, see the source code file: \$FOAM\_SRC/decompositionMethods/decompositionMethods/scotchDecomp/scotchDecomp.C
- manual Manual decomposition, where the user directly specifies the allocation of each cell to a particular processor.

For each method there are a set of coefficients specified in a sub-dictionary of *decompositionDict*, named *<method>Coeffs* as shown in the dictionary listing. The full set of keyword entries in the *decomposeParDict* dictionary are explained in Table 3.4.

The decomposePar utility is executed in the normal manner by typing

## decomposePar

On completion, a set of subdirectories will have been created, one for each processor, in the case directory. The directories are named *processorN* where  $N = 0, 1, \ldots$  represents a processor number and contains a time directory, containing the decomposed field descriptions, and a *constant/polyMesh* directory containing the decomposed mesh description.

# 3.4.2 Running a decomposed case

A decomposed OpenFOAM case is run in parallel using the  $\mathsf{openMPI}$  implementation of MPI.

<code>openMPI</code> can be run on a local multiprocessor machine very simply but when running on machines across a network, a file must be created that contains the host names of the machines. The file can be given any name and located at any path. In the following description we shall refer to such a file by the generic name, including full path, <machines>.

Compulsory entries			
numberOfSubdomains	Total number of subdomains	N	
method	Method of decomposition	simple/	
		${\tt hierarchical}/{}$	
		scotch/ metis/	
		manual/	
simpleCoeffs $\operatorname{entrie}$	S		
n	Number of subdomains in $x, y, z$	$(n_x n_y n_z)$	
delta	Cell skew factor	Typically, $10^{-3}$	
hierarchicalCoeffs	entries		
n	Number of subdomains in $x, y, z$	$(n_x n_y n_z)$	
delta	Cell skew factor	Typically, $10^{-3}$	
order	Order of decomposition	xyz/xzy/yxz	
scotchCoeffs entrie	S		
processorWeights	List of weighting factors for allocation	( <wt1><wtn>)</wtn></wt1>	
(optional)	of cells to processors; $$ is the		
	weighting factor for processor 1, etc.;		
	weights are normalised so can take any		
	range of values.		
strategy	Decomposition strategy (optional); de-		
	faults to "b"		
manualCoeffs entries			
dataFile	Name of file containing data of alloca-	" <filename>"</filename>	
	tion of cells to processors		
<b>Distributed data entries</b> (optional) — see section $3.4.3$			
distributed	Is the data distributed across several	yes/no	
	disks?		
roots	Root paths to case directories; <rt1></rt1>	$(< \texttt{rt1} > \ldots < \texttt{rtN} >)$	
	is the root path for node 1, <i>etc.</i>		
Table	Table 3.4: Keywords in <i>decompositionDict</i> dictionary		

Table 3.4: Keywords in *decompositionDict* dictionary.

The <machines> file contains the names of the machines listed one machine per line. The names must correspond to a fully resolved hostname in the /etc/hosts file of the machine on which the openMPI is run. The list must contain the name of the machine running the openMPI. Where a machine node contains more than one processor, the node name may be followed by the entry cpu=n where n is the number of processors openMPI should run on that node.

For example, let us imagine a user wishes to run openMPI from machine aaa on the following machines: aaa; bbb, which has 2 processors; and ccc. The *<machines*> would contain:

aaa bbb cpu=2 ccc An application is run in parallel using mpirun.

```
mpirun --hostfile <machines> -np <nProcs>
        <foamExec> <otherArgs> -parallel > log &
```

where: <nProcs> is the number of processors; <foamExec> is the executable, *e.g.*icoFoam; and, the output is redirected to a file named log. For example, if icoFoam is run on 4 nodes, specified in a file named *machines*, on the cavity tutorial in the \$FOAM\_RUN/-*tutorials/incompressible/icoFoam* directory, then the following command should be executed:

```
mpirun --hostfile machines -np 4 icoFoam -parallel > log &
```

## 3.4.3 Distributing data across several disks

Data files may need to be distributed if, for example, if only local disks are used in order to improve performance. In this case, the user may find that the root path to the case directory may differ between machines. The paths must then be specified in the *decomposeParDict* dictionary using distributed and roots keywords. The distributed entry should read

distributed yes;

and the roots entry is a list of root paths, <root0>, <root1>, ..., for each node

```
roots
<nRoots>
(
    "<root0>"
    "<root1>"
    ...
);
```

where <nRoots> is the number of roots.

Each of the *processorN* directories should be placed in the case directory at each of the root paths specified in the *decomposeParDict* dictionary. The *system* directory and *files* within the *constant* directory must also be present in each case directory. Note: the files in the *constant* directory are needed, but the *polyMesh* directory is not.

## 3.4.4 Post-processing parallel processed cases

When post-processing cases that have been run in parallel the user has two options:

- reconstruction of the mesh and field data to recreate the complete domain and fields, which can be post-processed as normal;
- post-processing each segment of decomposed domain individually.

#### Reconstructing mesh and data 3.4.4.1

After a case has been run in parallel, it can be reconstructed for post-processing. The case is reconstructed by merging the sets of time directories from each *processorN* directory into a single set of time directories. The reconstructPar utility performs such a reconstruction by executing the command:

## reconstructPar

When the data is distributed across several disks, it must be first copied to the local case directory for reconstruction.

#### Post-processing decomposed cases 3.4.4.2

The user may post-process decomposed cases using the paraFoam post-processor, described in section 6.1. The whole simulation can be post-processed by reconstructing the case or alternatively it is possible to post-process a segment of the decomposed domain individually by simply treating the individual processor directory as a case in its own right.

#### 3.5Standard solvers

The solvers with the OpenFOAM distribution are in the **\$FOAM\_SOLVERS** directory, reached quickly by typing app at the command line. This directory is further subdivided into several directories by category of continuum mechanics, e.g. incompressible flow, combustion and solid body stress analysis. Each solver is given a name that is reasonably descriptive, e.g. icoFoam solves incompressible, laminar flow. The current list of solvers distributed with OpenFOAM is given in Table 3.5.

'Basic' CFD codes	
laplacianFoam	Solves a simple Laplace equation, e.g. for thermal diffusion
	in a solid
potentialFoam	Simple potential flow solver which can be used to generate
	starting fields for full Navier-Stokes codes
scalarTransportFoam	Solves a transport equation for a passive scalar
Incompressible flow	
adjointShapeOptimiz-	Steady-state solver for incompressible, turbulent flow of non-
ationFoam	Newtonian fluids with optimisation of duct shape by applying
	"blockage" in regions causing pressure loss as estimated using
	an adjoint formulation
boundaryFoam	Steady-state solver for incompressible, 1D turbulent flow, typ-
	ically to generate boundary layer conditions at an inlet, for
	use in a simulation
channelFoam	Incompressible LES solver for flow in a channel
icoFoam	Transient solver for incompressible, laminar flow of Newtonian
	fluids
MRFSimpleFoam	Steady-state solver for incompressible, turbulent flow of non-
·	Newtonian fluids with MRF regions
	Continued on next page

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Continued from previous	s page
nonNewtonianIcoFoam	Transient solver for incompressible, laminar flow of non-
	Newtonian fluids
pimpleDyMFoam	Transient solver for incompressible, flow of Newtonian flu-
	ids on a moving mesh using the PIMPLE (merged PISO-
	SIMPLE) algorithm
pimpleFoam	Large time-step transient solver for incompressible, flow using
. –	the PIMPLE (merged PISO-SIMPLE) algorithm
pisoFoam	Transient solver for incompressible flow
porousSimpleFoam	Steady-state solver for incompressible, turbulent flow with im-
	plicit or explicit porosity treatment
shallowWaterFoam	Transient solver for inviscid shallow-water equations with ro-
	tation
simpleFoam	Steady-state solver for incompressible, turbulent flow
SRFSimpleFoam	Steady-state solver for incompressible, turbulent flow of non-
windSimpleFoam	Newtonian fluids in a single rotating frame Steady-state solver for incompressible, turbulent flow with ex-
	ternal source in the momentum equation
	ternal source in the momentum equation
Compressible flow	
rhoCentralFoam	Density-based compressible flow solver based on central-
	upwind schemes of Kurganov and Tadmor
rhoCentralDyMFoam	Density-based compressible flow solver based on central-
	upwind schemes of Kurganov and Tadmor with moving mesh
	capability and turbulence modelling
rhoPimpleFoam	Transient solver for laminar or turbulent flow of compressible
	fluids for HVAC and similar applications
rhoPorousMRFLTS-	Transient solver for laminar or turbulent flow of compressible
PimpleFoam	fluids with support for porous media and MRF for HVAC
	and similar applications, with local time-stepping for efficient
	steady-state solution
rhoPorousMRFSimple-	Steady-state solver for turbulent flow of compressible fluids
Foam	with RANS turbulence modelling, implicit or explicit porosity
	treatment and MRF for HVAC and similar applications
rhoPorousMRFPimple-	Transient solver for laminar or turbulent flow of compressible
Foam	fluids with support for porous media and MRF for HVAC and
vh o Cinon lo o Foom	similar applications Steedy state SIMPLEC solver for largingr or turbulant DANS
rhoSimplecFoam	Steady-state SIMPLEC solver for laminar or turbulent RANS flow of compressible fluids
rhoSimpleFoam	Steady-state SIMPLE solver for laminar or turbulent RANS
	flow of compressible fluids
sonicDyMFoam	Transient solver for trans-sonic/supersonic, laminar or turbu-
somed yinn barn	lent flow of a compressible gas with mesh motion
sonicFoam	Transient solver for trans-sonic/supersonic, laminar or turbu-
	lent flow of a compressible gas
sonicLiquidFoam	Transient solver for trans-sonic/supersonic, laminar flow of a
J	compressible liquid
	1 1
Multiphase flow	

Multiphase flow

Continued on next page

Continued from previous page

Continued from previous	Continued from previous page				
bubbleFoam	Solver for a system of 2 incompressible fluid phases with one				
	phase dispersed, e.g. gas bubbles in a liquid				
cavitatingFoam	Transient cavitation code based on the homogeneous equi-				
-	librium model from which the compressibility of the liq-				
	uid/vapour "mixture" is obtained				
compressibleInterFoam	Solver for 2 compressible, isothermal immiscible fluids using				
	a VOF (volume of fluid) phase-fraction based interface cap-				
	turing approach				
interFoam	Solver for 2 incompressible, isothermal immiscible fluids us-				
	ing a VOF (volume of fluid) phase-fraction based interface				
	capturing approach				
interDyMFoam	Solver for 2 incompressible, isothermal immiscible fluids using				
	a VOF (volume of fluid) phase-fraction based interface captur-				
	ing approach, with optional mesh motion and mesh topology				
·	changes including adaptive re-meshing.				
interMixingFoam	Solver for 3 incompressible fluids, two of which are miscible,				
	using a VOF method to capture the interface				
interPhaseChangeFoam	Solver for 2 incompressible, isothermal immiscible fluids with				
	phase-change (e.g. cavitation). Uses a VOF (volume of fluid)				
	phase-fraction based interface capturing approach				
LTSInterFoam	Local time stepping (LTS, steady-state) solver for 2 incom-				
	pressible, isothermal immiscible fluids using a VOF (volume				
	of fluid) phase-fraction based interface capturing approach				
MRFInterFoam	Multiple reference frame (MRF) solver for 2 incompressible,				
	isothermal immiscible fluids using a VOF (volume of fluid)				
	phase-fraction based interface capturing approach				
MRFMultiphaseInter-	Multiple reference frame (MRF) solver for $n$ incompressible				
Foam	fluids which captures the interfaces and includes surface-				
	tension and contact-angle effects for each phase				
multiphaseInterFoam	Solver for $n$ incompressible fluids which captures the interfaces				
	and includes surface-tension and contact-angle effects for each				
	phase				
porousInterFoam	Solver for 2 incompressible, isothermal immiscible fluids us-				
	ing a VOF (volume of fluid) phase-fraction based interface				
	capturing approach, with explicit handling of porous zones				
settlingFoam	Solver for 2 incompressible fluids for simulating the settling				
J	of the dispersed phase				
twoLiquidMixingFoam	Solver for mixing 2 incompressible fluids				
twoPhaseEulerFoam	Solver for a system of 2 incompressible fluid phases with one				
	phase dispersed, e.g. gas bubbles in a liquid				

# Direct numerical simulation (DNS)

dnsFoam	Direct numerical simulation solver for boxes of isotropic tur-
	bulence

## Combustion

Continued on next page

	I
chemFoam	Solver for chemistry problems - designed for use on single cell
	cases to provide comparison against other chemistry solvers -
	single cell mesh created on-the-fly - fields created on the fly
	from the initial conditions
coldEngineFoam	Solver for cold-flow in internal combustion engines
dieselEngineFoam	Solver for diesel engine spray and combustion
dieselFoam	Solver for diesel spray and combustion
engineFoam	Solver for internal combustion engines
fireFoam	Transient Solver for Fires and turbulent diffusion flames
PDRFoam	Solver for compressible premixed/partially-premixed combus-
	tion with turbulence modelling
reactingFoam	Solver for combustion with chemical reactions
rhoReactingFoam	Solver for combustion with chemical reactions using density
	based thermodynamics package
XiFoam	Solver for compressible premixed/partially-premixed combus-
	tion with turbulence modelling

# Continued from previous page

# Heat transfer and buoyancy-driven flows

fieur transfer and subjancy arrent news			
buoyantBaffleSimple-	Steady-state solver for buoyant, turbulent flow of compressible		
Foam	fluids using thermal baffles		
buoyantBoussinesqPim-	Transient solver for buoyant, turbulent flow of incompressible		
pleFoam	fluids		
buoyantBoussinesqSim-	Steady-state solver for buoyant, turbulent flow of incompress-		
pleFoam	ible fluids		
buoyantPimpleFoam	Transient solver for buoyant, turbulent flow of compressible		
	fluids for ventilation and heat-transfer		
buoyantSimpleFoam	Steady-state solver for buoyant, turbulent flow of compressible		
	fluids		
buoyantSimpleRadiation-	Steady-state solver for buoyant, turbulent flow of compressible		
Foam	fluids, including radiation, for ventilation and heat-transfer		
chtMultiRegionFoam	Combination of heatConductionFoam and buoyantFoam for		
	conjugate heat transfer between a solid region and fluid re-		
	gion		

# Particle-tracking flows

coalChemistryFoam	Transient solver for: - compressible, - turbulent flow, with -
	coal and limestone parcel injections, - energy source, and -
	combustion
icoUncoupledKinem-	Transient solver for the passive transport of a single kinematic
aticParcelDyMFoam	particle could
icoUncoupledKinem-	Transient solver for the passive transport of a single kinematic
aticParcelFoam	particle could
LTSReactingParcelFoam	Local time stepping (LTS) solver for steady, compressible,
	laminar or turbulent reacting and non-reacting flow with mul-
	tiphase Lagrangian parcels and porous media, including ex-
	plicit sources for mass, momentum and energy
	Continued on next page

Continued from previous	page
porousExplicitSource-	Transient PISO solver for compressible, laminar or turbulent
ReactingParcelFoam	flow with reacting multiphase Lagrangian parcels for porous
	media, including explicit sources for mass, momentum and
	energy
reactingParcelFilmFoam	Transient PISO solver for compressible, laminar or turbulent
	flow with reacting Lagrangian parcels, and surface film mod-
	elling
reactingParcelFoam	Transient PISO solver for compressible, laminar or turbulent
	flow with reacting Lagrangian parcels
uncoupledKinematic-	Transient solver for the passive transport of a single kinematic
ParcelFoam	particle could
Molecular dunamica	mathada
Molecular dynamics mdEquilibrationFoam	Equilibrates and/or preconditions molecular dynamics sys-
	tems
mdFoam	Molecular dynamics solver for fluid dynamics
indi odili	Norecular dynamics solver for huld dynamics
Direct simulation M	onte Carlo methods
dsmcFoam	Direct simulation Monte Carlo (DSMC) solver for 3D, tran-
	sient, multi- species flows
Floatnomognatics	
Electromagnetics electrostaticFoam	Solver for electrostatics
	Solver for the magnetic field generated by permanent magnets
magneticFoam mhdFoam	Solver for magnetohydrodynamics (MHD): incompressible,
mnuroam	laminar flow of a conducting fluid under the influence of a
	magnetic field
	magnetic neid
Stress analysis of sol	lids
solidDisplacement-	Transient segregated finite-volume solver of linear-elastic,
Foam	small-strain deformation of a solid body, with optional ther-
	mal diffusion and thermal stresses
solidEquilibriumDis-	Steady-state segregated finite-volume solver of linear-elastic,
placementFoam	small-strain deformation of a solid body, with optional ther-
	mal diffusion and thermal stresses
Finance	
financialFoam	Solves the Black-Scholes equation to price commodities
	solves the Diack-Scholes equation to price commodifies
	Table 3.5: Standard library solvers.

Table 3.5: Standard library solvers.

#### 3.6 Standard utilities

The utilities with the OpenFOAM distribution are in the **\$FOAM\_UTILITIES** directory, reached quickly by typing util at the command line. Again the names are reasonably descriptive, e.g.ideasToFoam converts mesh data from the format written by I-DEAS to the OpenFOAM format. The current list of utilities distributed with OpenFOAM is given in Table 3.6.

# Pre-processing

applyBoundaryLayer	Apply a simplified boundary-layer model to the velocity and
	turbulence fields based on the 1/7th power-law
applyWallFunction-	Updates OpenFOAM RAS cases to use the new (v1.6) wall
BoundaryConditions	function framework
boxTurb	Makes a box of turbulence which conforms to a given energy
	spectrum and is divergence free
changeDictionary	Utility to change dictionary entries, e.g. can be used to change
0 1	the patch type in the field and polyMesh/boundary files
dsmcInitialise	Initialise a case for dsmcFoam by reading the initialisation
	dictionary system/dsmclnitialise
engineSwirl	Generates a swirling flow for engine calulations
faceAgglomerate	(Currently no description)
foamUpgradeCyclics	Tool to upgrade mesh and fields for split cyclics
foamUpgradeFvSolution	Simple tool to upgrade the syntax of <i>system/fvSolution::solvers</i>
mapFields	Maps volume fields from one mesh to another, reading and
	interpolating all fields present in the time directory of both
	cases. Parallel and non-parallel cases are handled without the
	need to reconstruct them first
mdInitialise	Initialises fields for a molecular dynamics (MD) simulation
setFields	Set values on a selected set of cells/patchfaces through a dic-
	tionary
viewFactorGen	(Description not found)
wallFunctionTable	Generates a table suitable for use by tabulated wall functions
Mesh generation	
blockMesh	A multi-block mesh generator Extrude mesh from existing patch (by default outwards facing
	Extrude mesh from existing patch (by default outwards facing
blockMesh extrudeMesh	Extrude mesh from existing patch (by default outwards facing normals; optional flips faces) or from patch read from file.
blockMesh	Extrude mesh from existing patch (by default outwards facing normals; optional flips faces) or from patch read from file. Takes 2D mesh (all faces 2 points only, no front and back
blockMesh extrudeMesh	Extrude mesh from existing patch (by default outwards facing normals; optional flips faces) or from patch read from file. Takes 2D mesh (all faces 2 points only, no front and back faces) and creates a 3D mesh by extruding with specified
blockMesh extrudeMesh extrude2DMesh	Extrude mesh from existing patch (by default outwards facing normals; optional flips faces) or from patch read from file. Takes 2D mesh (all faces 2 points only, no front and back faces) and creates a 3D mesh by extruding with specified thickness
blockMesh extrudeMesh	Extrude mesh from existing patch (by default outwards facing normals; optional flips faces) or from patch read from file. Takes 2D mesh (all faces 2 points only, no front and back faces) and creates a 3D mesh by extruding with specified thickness Extrude faceZones into separate mesh (as a different region),
blockMesh extrudeMesh extrude2DMesh extrudeToRegionMesh	Extrude mesh from existing patch (by default outwards facing normals; optional flips faces) or from patch read from file. Takes 2D mesh (all faces 2 points only, no front and back faces) and creates a 3D mesh by extruding with specified thickness Extrude faceZones into separate mesh (as a different region), e.g. for creating liquid film regions
blockMesh extrudeMesh extrude2DMesh	Extrude mesh from existing patch (by default outwards facing normals; optional flips faces) or from patch read from file. Takes 2D mesh (all faces 2 points only, no front and back faces) and creates a 3D mesh by extruding with specified thickness Extrude faceZones into separate mesh (as a different region),
blockMesh extrudeMesh extrude2DMesh extrudeToRegionMesh	Extrude mesh from existing patch (by default outwards facing normals; optional flips faces) or from patch read from file. Takes 2D mesh (all faces 2 points only, no front and back faces) and creates a 3D mesh by extruding with specified thickness Extrude faceZones into separate mesh (as a different region), e.g. for creating liquid film regions
blockMesh extrudeMesh extrude2DMesh extrudeToRegionMesh snappyHexMesh	Extrude mesh from existing patch (by default outwards facing normals; optional flips faces) or from patch read from file. Takes 2D mesh (all faces 2 points only, no front and back faces) and creates a 3D mesh by extruding with specified thickness Extrude faceZones into separate mesh (as a different region), e.g. for creating liquid film regions
blockMesh extrudeMesh extrude2DMesh extrudeToRegionMesh snappyHexMesh Mesh conversion	Extrude mesh from existing patch (by default outwards facing normals; optional flips faces) or from patch read from file. Takes 2D mesh (all faces 2 points only, no front and back faces) and creates a 3D mesh by extruding with specified thickness Extrude faceZones into separate mesh (as a different region), e.g. for creating liquid film regions Automatic split hex mesher. Refines and snaps to surface
blockMesh extrudeMesh extrude2DMesh extrudeToRegionMesh snappyHexMesh Mesh conversion	Extrude mesh from existing patch (by default outwards facing normals; optional flips faces) or from patch read from file. Takes 2D mesh (all faces 2 points only, no front and back faces) and creates a 3D mesh by extruding with specified thickness Extrude faceZones into separate mesh (as a different region), e.g. for creating liquid film regions Automatic split hex mesher. Refines and snaps to surface Converts an ANSYS input mesh file, exported from I-DEAS,
blockMesh extrudeMesh extrude2DMesh extrudeToRegionMesh snappyHexMesh <u>Mesh conversion</u> ansysToFoam	Extrude mesh from existing patch (by default outwards facing normals; optional flips faces) or from patch read from file. Takes 2D mesh (all faces 2 points only, no front and back faces) and creates a 3D mesh by extruding with specified thickness Extrude faceZones into separate mesh (as a different region), e.g. for creating liquid film regions Automatic split hex mesher. Refines and snaps to surface Converts an ANSYS input mesh file, exported from I-DEAS, to OpenFOAM format
blockMesh extrudeMesh extrude2DMesh extrudeToRegionMesh snappyHexMesh <u>Mesh conversion</u> ansysToFoam cfx4ToFoam	Extrude mesh from existing patch (by default outwards facing normals; optional flips faces) or from patch read from file. Takes 2D mesh (all faces 2 points only, no front and back faces) and creates a 3D mesh by extruding with specified thickness Extrude faceZones into separate mesh (as a different region), e.g. for creating liquid film regions Automatic split hex mesher. Refines and snaps to surface Converts an ANSYS input mesh file, exported from I-DEAS, to OpenFOAM format Converts a CFX 4 mesh to OpenFOAM format
blockMesh extrudeMesh extrude2DMesh extrudeToRegionMesh snappyHexMesh <u>Mesh conversion</u> ansysToFoam cfx4ToFoam	Extrude mesh from existing patch (by default outwards facing normals; optional flips faces) or from patch read from file. Takes 2D mesh (all faces 2 points only, no front and back faces) and creates a 3D mesh by extruding with specified thickness Extrude faceZones into separate mesh (as a different region), e.g. for creating liquid film regions Automatic split hex mesher. Refines and snaps to surface Converts an ANSYS input mesh file, exported from I-DEAS, to OpenFOAM format Converts a CFX 4 mesh to OpenFOAM format Reads in a datToFoam mesh file and outputs a points file.
blockMesh extrudeMesh extrude2DMesh extrudeToRegionMesh snappyHexMesh <u>Mesh conversion</u> ansysToFoam cfx4ToFoam datToFoam	Extrude mesh from existing patch (by default outwards facing normals; optional flips faces) or from patch read from file. Takes 2D mesh (all faces 2 points only, no front and back faces) and creates a 3D mesh by extruding with specified thickness Extrude faceZones into separate mesh (as a different region), e.g. for creating liquid film regions Automatic split hex mesher. Refines and snaps to surface Converts an ANSYS input mesh file, exported from I-DEAS, to OpenFOAM format Converts a CFX 4 mesh to OpenFOAM format Reads in a datToFoam mesh file and outputs a points file. Used in conjunction with blockMesh
blockMesh extrudeMesh extrude2DMesh extrudeToRegionMesh snappyHexMesh <u>Mesh conversion</u> ansysToFoam cfx4ToFoam datToFoam fluent3DMeshToFoam fluentMeshToFoam	Extrude mesh from existing patch (by default outwards facing normals; optional flips faces) or from patch read from file. Takes 2D mesh (all faces 2 points only, no front and back faces) and creates a 3D mesh by extruding with specified thickness Extrude faceZones into separate mesh (as a different region), e.g. for creating liquid film regions Automatic split hex mesher. Refines and snaps to surface Converts an ANSYS input mesh file, exported from I-DEAS, to OpenFOAM format Converts a CFX 4 mesh to OpenFOAM format Reads in a datToFoam mesh file and outputs a points file. Used in conjunction with blockMesh Converts a Fluent mesh to OpenFOAM format
blockMesh extrudeMesh extrude2DMesh extrudeToRegionMesh snappyHexMesh <u>Mesh conversion</u> ansysToFoam cfx4ToFoam datToFoam fluent3DMeshToFoam fluentMeshToFoam foamMeshToFluent	Extrude mesh from existing patch (by default outwards facing normals; optional flips faces) or from patch read from file. Takes 2D mesh (all faces 2 points only, no front and back faces) and creates a 3D mesh by extruding with specified thickness Extrude faceZones into separate mesh (as a different region), e.g. for creating liquid film regions Automatic split hex mesher. Refines and snaps to surface Converts an ANSYS input mesh file, exported from I-DEAS, to OpenFOAM format Converts a CFX 4 mesh to OpenFOAM format Reads in a datToFoam mesh file and outputs a points file. Used in conjunction with blockMesh Converts a Fluent mesh to OpenFOAM format Converts a Fluent mesh to OpenFOAM format Writes out the OpenFOAM mesh in Fluent mesh format
blockMesh extrudeMesh extrude2DMesh extrudeToRegionMesh snappyHexMesh <u>Mesh conversion</u> ansysToFoam cfx4ToFoam datToFoam fluent3DMeshToFoam fluentMeshToFoam	Extrude mesh from existing patch (by default outwards facing normals; optional flips faces) or from patch read from file. Takes 2D mesh (all faces 2 points only, no front and back faces) and creates a 3D mesh by extruding with specified thickness Extrude faceZones into separate mesh (as a different region), e.g. for creating liquid film regions Automatic split hex mesher. Refines and snaps to surface Converts an ANSYS input mesh file, exported from I-DEAS, to OpenFOAM format Converts a CFX 4 mesh to OpenFOAM format Reads in a datToFoam mesh file and outputs a points file. Used in conjunction with blockMesh Converts a Fluent mesh to OpenFOAM format converts a Fluent mesh to OpenFOAM format itple region and region boundary handling Writes out the OpenFOAM mesh in Fluent mesh format Reads an OpenFOAM mesh and writes a PROSTAR (v4)
blockMesh extrudeMesh extrude2DMesh extrudeToRegionMesh snappyHexMesh <u>Mesh conversion</u> ansysToFoam cfx4ToFoam datToFoam fluent3DMeshToFoam fluentMeshToFoam foamMeshToFluent	Extrude mesh from existing patch (by default outwards facing normals; optional flips faces) or from patch read from file. Takes 2D mesh (all faces 2 points only, no front and back faces) and creates a 3D mesh by extruding with specified thickness Extrude faceZones into separate mesh (as a different region), e.g. for creating liquid film regions Automatic split hex mesher. Refines and snaps to surface Converts an ANSYS input mesh file, exported from I-DEAS, to OpenFOAM format Converts a CFX 4 mesh to OpenFOAM format Reads in a datToFoam mesh file and outputs a points file. Used in conjunction with blockMesh Converts a Fluent mesh to OpenFOAM format Converts a Fluent mesh to OpenFOAM format Writes out the OpenFOAM mesh in Fluent mesh format

Continued	from	previous	page

foamToSurface	Reads an OpenFOAM mesh and writes the boundaries in a
	surface format
gambitToFoam	Converts a <b>GAMBIT</b> mesh to OpenFOAM format
gmshToFoam	Reads .msh file as written by Gmsh
ideasUnvToFoam	I-Deas unv format mesh conversion
kivaToFoam	Converts a KIVA grid to OpenFOAM format
mshToFoam	Converts .msh file generated by the Adventure system
netgenNeutralToFoam	Converts neutral file format as written by Netgen v4.4
plot3dToFoam	Plot3d mesh (ascii/formatted format) converter
sammToFoam	Converts a STAR-CD $(v3)$ SAMM mesh to OpenFOAM format
star3ToFoam	Converts a STAR-CD (v3) PROSTAR mesh into OpenFOAM
	format
star4ToFoam	Converts a STAR-CD (v4) PROSTAR mesh into OpenFOAM
	format
tetgenToFoam	Converts .ele and .node and .face files, written by tetgen
writeMeshObj	For mesh debugging: writes mesh as three separate OBJ files
	which can be viewed with e.g. javaview

# Mesh manipulation

Mesn manipulation	
attachMesh	Attach topologically detached mesh using prescribed mesh
	modifiers
autoPatch	Divides external faces into patches based on (user supplied)
	feature angle
checkMesh	Checks validity of a mesh
createBaffles	Makes internal faces into boundary faces. Does not duplicate
	points, unlike mergeOrSplitBaffles
createPatch	Utility to create patches out of selected boundary faces. Faces
	come either from existing patches or from a faceSet
deformedGeom	Deforms a polyMesh using a displacement field U and a scaling
	factor supplied as an argument
flattenMesh	Flattens the front and back planes of a 2D cartesian mesh
insideCells	Picks up cells with cell centre 'inside' of surface. Requires
	surface to be closed and singly connected
mergeMeshes	Merge two meshes
mergeOrSplitBaffles	Detects faces that share points (baffles). Either merge them
	or duplicate the points
mirrorMesh	Mirrors a mesh around a given plane
moveDynamicMesh	Mesh motion and topological mesh changes utility
moveEngineMesh	Solver for moving meshes for engine calculations
moveMesh	Solver for moving meshes
objToVTK	Read obj line (not surface!) file and convert into vtk
polyDualMesh	Calculate the dual of a polyMesh. Adheres to all the feature
	and patch edges
refineMesh	Utility to refine cells in multiple directions
renumberMesh	Renumbers the cell list in order to reduce the bandwidth,
	reading and renumbering all fields from all the time directories
rotateMesh	Rotates the mesh and fields from the direction $n_1$ to the direction $n_1$ to the direction $n_1$ to the direction $n_1$ to the direction $n_2$ to the directi
	tion $n_2$
setSet	Manipulate a cell/face/point/ set or zone interactively
	Continued on next page

Continued from previou	ns page
setsToZones	Add $pointZones/faceZones/cellZones$ to the mesh from similar
	named pointSets/faceSets/cellSets
singleCellMesh	Removes all but one cells of the mesh. Used to generate mesh
	and fields that can be used for boundary-only data. Might
	easily result in illegal mesh though so only look at boundaries
	in paraview
splitMesh	Splits mesh by making internal faces external. Uses attachDe-
	tach
${\sf splitMeshRegions}$	Splits mesh into multiple regions
stitchMesh	'Stitches' a mesh
subsetMesh	Selects a section of mesh based on a <b>cellSet</b>
topoSet	Operates on cellSets/faceSets/pointSets through a dictionary
transformPoints	Transforms the mesh points in the polyMesh directory accord-
	ing to the translate, rotate and scale options
zipUpMesh	Reads in a mesh with hanging vertices and zips up the cells
	to guarantee that all polyhedral cells of valid shape are closed
Other mesh tools	
autoRefineMesh	Utility to refine cells near to a surface
collapseEdges	Collapse short edges and combines edges that are in line
combinePatchFaces	- Unecks for multiple parch faces on same cen and complines
combinePatchFaces	Checks for multiple patch faces on same cell and combines them. These result from e.g. refined neighbouring cells get-
combinePatchFaces	them. These result from e.g. refined neighbouring cells get-
	them. These result from e.g. refined neighbouring cells get- ting removed, leaving 4 exposed faces with same owner
combinePatchFaces modifyMesh PDRMesh	them. These result from e.g. refined neighbouring cells get- ting removed, leaving 4 exposed faces with same owner Manipulates mesh elements
modifyMesh	them. These result from e.g. refined neighbouring cells get- ting removed, leaving 4 exposed faces with same owner
modifyMesh PDRMesh	<ul><li>them. These result from e.g. refined neighbouring cells get-</li><li>ting removed, leaving 4 exposed faces with same owner</li><li>Manipulates mesh elements</li><li>Mesh and field preparation utility for PDR type simulations</li></ul>
modifyMesh PDRMesh refineHexMesh	<ul><li>them. These result from e.g. refined neighbouring cells getting removed, leaving 4 exposed faces with same owner</li><li>Manipulates mesh elements</li><li>Mesh and field preparation utility for PDR type simulations</li><li>Refines a hex mesh by 2x2x2 cell splitting</li></ul>
modifyMesh PDRMesh refineHexMesh	<ul> <li>them. These result from e.g. refined neighbouring cells getting removed, leaving 4 exposed faces with same owner Manipulates mesh elements</li> <li>Mesh and field preparation utility for PDR type simulations Refines a hex mesh by 2x2x2 cell splitting</li> <li>Tries to figure out what the refinement level is on refined</li> </ul>
modifyMesh PDRMesh refineHexMesh refinementLevel	<ul> <li>them. These result from e.g. refined neighbouring cells getting removed, leaving 4 exposed faces with same owner Manipulates mesh elements</li> <li>Mesh and field preparation utility for PDR type simulations Refines a hex mesh by 2x2x2 cell splitting</li> <li>Tries to figure out what the refinement level is on refined cartesian meshes. Run <i>before</i> snapping</li> </ul>
modifyMesh PDRMesh refineHexMesh refinementLevel refineWallLayer	<ul> <li>them. These result from e.g. refined neighbouring cells getting removed, leaving 4 exposed faces with same owner Manipulates mesh elements</li> <li>Mesh and field preparation utility for PDR type simulations Refines a hex mesh by 2x2x2 cell splitting</li> <li>Tries to figure out what the refinement level is on refined cartesian meshes. Run <i>before</i> snapping</li> <li>Utility to refine cells next to patches</li> </ul>
modifyMesh PDRMesh refineHexMesh refinementLevel refineWallLayer removeFaces	<ul> <li>them. These result from e.g. refined neighbouring cells getting removed, leaving 4 exposed faces with same owner Manipulates mesh elements</li> <li>Mesh and field preparation utility for PDR type simulations Refines a hex mesh by 2x2x2 cell splitting</li> <li>Tries to figure out what the refinement level is on refined cartesian meshes. Run <i>before</i> snapping</li> <li>Utility to refine cells next to patches</li> <li>Utility to remove faces (combines cells on both sides)</li> </ul>
modifyMesh PDRMesh refineHexMesh refinementLevel refineWallLayer removeFaces selectCells splitCells	them. These result from e.g. refined neighbouring cells get- ting removed, leaving 4 exposed faces with same owner Manipulates mesh elements Mesh and field preparation utility for PDR type simulations Refines a hex mesh by 2x2x2 cell splitting Tries to figure out what the refinement level is on refined cartesian meshes. Run <i>before</i> snapping Utility to refine cells next to patches Utility to remove faces (combines cells on both sides) Select cells in relation to surface Utility to split cells with flat faces
modifyMesh PDRMesh refineHexMesh refinementLevel refineWallLayer removeFaces selectCells splitCells <b>Post-processing gra</b>	them. These result from e.g. refined neighbouring cells get- ting removed, leaving 4 exposed faces with same owner Manipulates mesh elements Mesh and field preparation utility for PDR type simulations Refines a hex mesh by 2x2x2 cell splitting Tries to figure out what the refinement level is on refined cartesian meshes. Run <i>before</i> snapping Utility to refine cells next to patches Utility to remove faces (combines cells on both sides) Select cells in relation to surface Utility to split cells with flat faces
modifyMesh PDRMesh refineHexMesh refinementLevel refineWallLayer removeFaces selectCells splitCells	them. These result from e.g. refined neighbouring cells get- ting removed, leaving 4 exposed faces with same owner Manipulates mesh elements Mesh and field preparation utility for PDR type simulations Refines a hex mesh by 2x2x2 cell splitting Tries to figure out what the refinement level is on refined cartesian meshes. Run <i>before</i> snapping Utility to refine cells next to patches Utility to remove faces (combines cells on both sides) Select cells in relation to surface Utility to split cells with flat faces

Post-processing	data	converters
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data

i ost-processing data	Converters
foamDataToFluent	Translates OpenFOAM data to Fluent format
foamToEnsight	Translates OpenFOAM data to EnSight format
foamToEnsightParts	Translates OpenFOAM data to Ensight format. An Ensight
	part is created for each <b>cellZone</b> and patch
foamToFieldview9	Write out the OpenFOAM mesh in Version 3.0 Fieldview-UNS
	format (binary)
foamToGMV	Translates foam output to GMV readable files
foamToTecplot360	Tecplot binary file format writer
foamToVTK	Legacy VTK file format writer
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 smapToFoam
 Translates a STAR-CD SMAP data file into OpenFOAM field format

Post-processing	velocity fields
-----------------	-----------------

1 0	v
Со	Configurable graph drawing program
enstrophy	Calculates and writes the enstrophy of the velocity field $\tt U$
flowType	Calculates and writes the flowType of velocity field U
Lambda2	Calculates and writes the second largest eigenvalue of the sum
	of the square of the symmetrical and anti-symmetrical parts
	of the velocity gradient tensor
Mach	Calculates and optionally writes the local Mach number from
	the velocity field U at each time
Pe	Calculates and writes the Pe number as a surfaceScalar-
	Field obtained from field phi
Q	Calculates and writes the second invariant of the velocity gra-
	dient tensor
streamFunction	Calculates and writes the stream function of velocity field U
	at each time
uprime	Calculates and writes the scalar field of uprime $(\sqrt{2k/3})$
vorticity	Calculates and writes the vorticity of velocity field U

## Post-processing stress fields

stressComponents	Calculates and writes the scalar fields of the six components
	of the stress tensor <b>sigma</b> for each time

## Post-processing scalar fields

pPrime2	Calculates and writes the scalar field of pPrime2 $([p-\overline{p}]^2)$ at
	each time
Post-processing at	t walls
wallGradU	Calculates and writes the gradient of U at the wall
wallHeatFlux	Calculates and writes the heat flux for all patches as the
	boundary field of a volScalarField and also prints the inte-
	grated flux for all wall patches
wallShearStress	Calculates and writes the wall shear stress, for the specified
	times
yPlusLES	Calculates and reports yPlus for all wall patches, for the spec-
	ified times
yPlusRAS	Calculates and reports yPlus for all wall patches, for the spec-
	ified times when using RAS turbulence models

## Post-processing turbulence

createTurbulenceFields	Creates a full set of turbulence fields		
R	Calculates and writes the Reynolds stress R for the current		
	time step		
Post-processing patch data			
patchAverage	Calculates the average of the specified field over the specified		
	patch		

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patchIntegrate	Calculates the integral of the specified field over the specified		
	patch		

particleTracks	Generates a VTK file of particle tracks for cases that were
1	computed using a tracked-parcel-type cloud
steadyParticleTracks	Generates a VTK file of particle tracks for cases that were
, , , , , , , , , , , , , , , , , , ,	computed using a steady-state cloud NOTE: case must be
	re-constructed (if running in parallel) before use
Sampling post-proc	cessing
probeLocations	Probe locations
sample	Sample field data with a choice of interpolation schemes, same
	pling options and write formats
Miscellaneous post	-processing
dsmcFieldsCalc	Calculate intensive fields (U and T) from averaged extensive
	fields from a DSMC calculation
engineCompRatio	Calculate the geometric compression ratio. Note that if you have valves and/or extra volumes it will not work, since it
	calculates the volume at BDC and TCD
execFlowFunctionObjec	tts Execute the set of functionObjects specified in the selected
	dictionary (which defaults to <i>system/controlDict</i> ) for the se
	lected set of times. Alternative dictionaries should be placed in the <i>system</i> / folder
foamCalc	Generic utility for simple field calculations at specified times
foamListTimes	List times using timeSelector
pdfPlot	Generates a graph of a probability distribution function
postChannel	Post-processes data from channel flow calculations
ptot	For each time: calculate the total pressure
wdot	Calculates and writes wdot for each time
writeCellCentres	Write the three components of the cell centres as volScalar-
	Fields so they can be used in postprocessing in thresholding

# Surface mesh (e.g. STL) tools

	/			
surfaceAdd	Add two surfaces. Does geometric merge on points. Does not			
	check for overlapping/intersecting triangles			
surfaceAutoPatch	Patches surface according to feature angle. Like autoPatch			
surfaceCheck	(Currently no description)			
surfaceClean	- removes baffles - collapses small edges, removing triangles.			
	- converts sliver triangles into split edges by projecting point			
	onto base of triangle			
surfaceCoarsen	Surface coarsening using 'bunnylod':			
surfaceConvert	Converts from one surface mesh format to another			
surfaceFeatureConvert Convert between edgeMesh formats				
surfaceFeatureExtract	Extracts and writes surface features to file			
surfaceFind	Finds nearest face and vertex			
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surfaceInertia	Calculates the inertia tensor and principal axes and moments of a command line specified triSurface. Inertia can either be of the solid body or of a thin shell	
surfaceMeshConvert	Convert between surface formats with optional scaling or transformations (rotate/translate) on a coordinateSystem	
surfaceMeshConvert- Testing	Converts from one surface mesh format to another, but pri- marily used for testing functionality	
surfaceMeshExport	Export from surfMesh to various third-party surface formats with optional scaling or transformations (rotate/translate) on a coordinateSystem	
surfaceMeshImport	Import from various third-party surface formats into surfMesh with optional scaling or transformations (rotate/translate) on a coordinateSystem	
surfaceMeshInfo	Miscellaneous information about surface meshes	
surfaceMeshTriangulate	Extracts triSurface from a polyMesh. Triangulates all bound- ary faces. Region numbers on triangles are the patch numbers of the polyMesh. Optionally only triangulates named patches	
surfaceOrient	Set normal consistent with respect to a user provided 'outside' point. If -inside the point is considered inside	
surfacePointMerge	Merges points on surface if they are within absolute distance. Since absolute distance use with care!	
surfaceRedistributePar	(Re)distribution of triSurface. Either takes an undecomposed surface or an already decomposed surface and redistribute it so each processor has all triangles that overlap its mesh	
surfaceRefineRedGreen	Refine by splitting all three edges of triangle ('red' refine- ment). Neighbouring triangles (which are not marked for re- finement get split in half ('green') refinement. (R. Verfuerth, "A review of a posteriori error estimation and adaptive mesh refinement techniques", Wiley-Teubner, 1996)	
surfaceSmooth	Example of a simple laplacian smoother	
surfaceSplitByPatch	Writes regions of triSurface to separate files	
surfaceSplitNonMani-	Takes multiply connected surface and tries to split surface at	
folds	multiply connected edges by duplicating points. Introduces concept of - borderEdge. Edge with 4 faces connected to it. - borderPoint. Point connected to exactly 2 borderEdges borderLine. Connected list of borderEdges	
surfaceSubset	A surface analysis tool which sub-sets the triSurface to choose only a part of interest. Based on subsetMesh	
surfaceToPatch	Reads surface and applies surface regioning to a mesh. Uses boundaryMesh to do the hard work	
surfaceTransformPoints	Transform (scale/rotate) a surface. Like transformPoints but for surfaces	

# Parallel processing

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page
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redistributeMeshPar	Redistributes existing decomposed mesh and fields according		
	to the current settings in the decomposeParDict file		

#### Thermophysical-related utilities

adiabaticFlameT	Calculates the adiabatic flame temperature for a given fuel		
	over a range of unburnt temperatures and equivalence ratios		
chemkinToFoam	Converts CHEMKIN 3 thermodynamics and reaction data files		
	into OpenFOAM format		
equilibriumCO	Calculates the equilibrium level of carbon monoxide		
equilibriumFlameT	Calculates the equilibrium flame temperature for a given f		
	and pressure for a range of unburnt gas temperatures and		
	equivalence ratios; the effects of dissociation on $O_2$ , $H_2O$ and		
	$CO_2$ are included		
mixtureAdiabaticFlameT	Calculates the adiabatic flame temperature for a given mix-		
	ture at a given temperature		

#### Miscellaneous utilities

expandDictionary	Read the dictionary provided as an argument, expand the macros etc. and write the resulting dictionary to standard output
foamDebugSwitches	Write out all library debug switches
foamFormatConvert	Converts all IOobjects associated with a case into the format specified in the <i>controlDict</i>
foamInfoExec	Interrogates a case and prints information to stdout
patchSummary	Writes fields and boundary condition info for each patch at each requested time instance

Table 3.6: Standard library utilities.

# 3.7 Standard libraries

The libraries with the OpenFOAM distribution are in the *\$FOAM\_LIB/\$WM\_OPTIONS* directory, reached quickly by typing lib at the command line. Again, the names are prefixed by lib and reasonably descriptive, *e.g.* incompressibleTransportModels contains the library of incompressible transport models. For ease of presentation, the libraries are separated into two types:

- General libraries those that provide general classes and associated functions listed in Table 3.7;
- Model libraries those that specify models used in computational continuum mechanics, listed in Table 3.8, Table 3.9 and Table 3.10.

Library of saste	openii orinii toono	openn er nn	
algorithms	Algorithms		
containers	Container classes		
db	Database classes		
		(	Y Y

## Library of basic OpenFOAM tools — OpenFOAM

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10
dimensioned <type> class and derivatives</type>
dimensionSet class
Field classes
Global settings
graph class
Interpolation schemes
Matrix classes
Memory management tools
Mesh classes
Primitive classes

# Finite volume method library — finiteVolume

	<sup>v</sup>
cfdTools	CFD tools
fields	Volume, surface and patch field classes; includes boundary
	conditions
finiteVolume	Finite volume discretisation
fvMatrices	Matrices for finite volume solution
fvMesh	Meshes for finite volume discretisation
interpolation	Field interpolation and mapping
surfaceMesh	Mesh surface data for finite volume discretisation
volMesh	Mesh volume (cell) data for finite volume discretisation

# Post-processing libraries

I I I I I I I I I I I I I I I I I I I	
fieldFunctionObjects	Field function objects including field averaging, min/max, etc.
foamCalcFunctions	Functions for the foamCalc utility
forces	Tools for post-processing force/lift/drag data with function
	objects
jobControl	Tools for controlling job running with a function object
postCalc	For using functionality of a function object as a post-
	processing activity
sampling	Tools for sampling field data at prescribed locations in a do-
	main
systemCall	General function object for making system calls while running
	a case
utilityFunctionObjects	Utility function objects

# Solution and mesh manipulation libraries

autoMesh	Library of functionality for the snappyHexMesh utility
blockMesh	Library of functionality for the <b>blockMesh</b> utility
dynamicMesh	For solving systems with moving meshes
dynamicFvMesh	Library for a finite volume mesh that can move and undergo
	topological changes
edgeMesh	For handling edge-based mesh descriptions
fvMotionSolvers	Finite volume mesh motion solvers
ODE	Solvers for ordinary differential equations
meshTools	Tools for handling a OpenFOAM mesh
surfMesh	Library for handling surface meshes of different formats
triSurface	For handling standard triangulated surface-based mesh de-
	scriptions

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topoChangerFvMesh Topological changes functionality (largely redundant)

Lagrangian particle traching instances						
coalCombustion	Coal dust combustion modelling					
dieselSpray	Diesel spray and injection modelling					
distributionModels	Particle distribution function modelling					
dsmc	Direct simulation Monte Carlo method modelling					
lagrangian	Basic Lagrangian, or particle-tracking, solution scheme					
lagrangianIntermediate	Particle-tracking kinematics, thermodynamics, multispecies					
	reactions, particle forces, <i>etc.</i>					
potential	Intermolecular potentials for molecular dynamics					
molecule	Molecule classes for molecular dynamics					
molecularMeasurements	For making measurements in molecular dynamics					
solidParticle	Solid particle implementation					

## Lagrangian particle tracking libraries

# Miscellaneous libraries

minore in the second se	
conversion	Tools for mesh and data conversions
decompositionMethods	Tools for domain decomposition
engine	Tools for engine calculations
fileFormats	Core routines for reading/writing data in some third-party
	formats
genericFvPatchField	A generic patch field
MGridGenGAMG-	Library for cell agglomeration using the MGridGen algorithm
Agglomeration	
pairPatchAgglom-	Primitive pair patch agglomeration method
eration	1 1 000
OSspecific	Operating system specific functions
randomProcesses	Tools for analysing and generating random processes

## Parallel libraries

distributed	Tools for searching and IO on distributed surfaces
reconstruct	Mesh/field reconstruction library
scotchDecomp	Scotch domain decomposition library
ptsotchDecomp	PTScotch domain decomposition library

Table 3.7: Shared object libraries for general use.

## Basic thermophysical models — basicThermophysicalModels

hPsiThermo	General thermophysical model calculation based on en-
	thalpy $h$ and compressibility $\psi$
hsPsiThermo	General thermophysical model calculation based on sensi-
	ble enthalpy $h_s$ and compressibility $\psi$
ePsiThermo	General thermophysical model calculation based on inter-
	nal energy $e$ and compressibility $\psi$
hRhoThermo	General thermophysical model calculation based on en-
	thalpy $h$
hsRhoThermo	General thermophysical model calculation based on sensi-
	ble enthalpy $h_s$
	Continued on next page

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pureMixture	General thermophysical model calculation for passive gas
	mixtures

Reaction models — reac	tionThermophysicalModels
hPsiMixtureThermo	Calculates enthalpy for combustion mixture based on en-
	thalpy $h$ and $\psi$
hsPsiMixtureThermo	Calculates enthalpy for combustion mixture based on sen-
	sible enthalpy $h_s$ and $\psi$
hRhoMixtureThermo	Calculates enthalpy for combustion mixture based on en-
	thalpy $h$ and $\rho$
hsRhoMixtureThermo	Calculates enthalpy for combustion mixture based on sen-
	sible enthalpy $h_s$ and $\rho$
hhuMixtureThermo	Calculates enthalpy for unburnt gas and combustion mix-
	ture
homogeneousMixture	Combustion mixture based on normalised fuel mass frac-
the NALL	tion b
inhomogeneousMixture	Combustion mixture based on $b$ and total fuel mass fraction
	$f_t$
veryInhomogeneousMixture	Combustion mixture based on $b$ , $f_t$ and unburnt fuel mass
dia a d Minterna	fraction $f_u$
dieselMixture	Combustion mixture based on $f_t$ and $f_u$
basicMultiComponent-	Basic mixture based on multiple components
Mixture	Device during the set of an analytical and an analytical
multiComponentMixture	Derived mixture based on multiple components
reactingMixture	Combustion mixture using thermodynamics and reaction
o grav Alivetica a	schemes
egrMixture	Exhaust gas recirculation mixture

Radiation models — radiation mo	diationModels
P1	P1 model
fvDOM	Finite volume discrete ordinate method
viewFactor	View factor radiation model
Laminar flame speed n	$\mathrm{nodels}$ — laminarFlameSpeedModels
<b>_</b>	Constant laminar flame speed
	Gulder's laminar flame speed model
GuldersEGRLaminar-	Gulder's laminar flame speed model with exhaust gas re-
FlameSpeed	circulation modelling
Barotropic compressibility	${ m lity} \; { m models} - { m barotropicCompressibilityModels}$
linear	Linear compressibility model
Chung	Chung compressibility model
Wallis	Wallis compressibility model

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Thermophysical properties of gaseous species — specie		
icoPolynomial	Incompressible polynomial equation of state, <i>e.g.</i> for liquids	
perfectGas	Perfect gas equation of state	
eConstThermo	Constant specific heat $c_p$ model with evaluation of internal energy $e$ and entropy $s$	
hConstThermo	Constant specific heat $c_p$ model with evaluation of enthalpy $h$ and entropy $s$	
hPolynomialThermo	$c_p$ evaluated by a function with coefficients from polynomials, from which $h, s$ are evaluated	
janafThermo	$c_p$ evaluated by a function with coefficients from JANAF thermodynamic tables, from which $h, s$ are evaluated	
specieThermo	Thermophysical properties of species, derived from $c_p$ , $h$ and/or $s$	
constTransport	Constant transport properties	
polynomialTransport	Polynomial based temperature-dependent transport properties	
sutherlandTransport	Sutherland's formula for temperature-dependent transport properties	

,	
NSRDSfunctions	National Standard Reference Data System (NSRDS) -
	American Institute of Chemical Engineers (AICHE) data
	compilation tables
APIfunctions	American Petroleum Institute (API) function for vapour
	mass diffusivity

# $\underline{\quad Chemistry \ model} - {\sf chemistryModel}$

J		
chemistryModel	Chemical reaction model	
chemistrySolver	Chemical reaction solver	

# Other libraries

liquidProperties	Thermophysical properties of liquids
liquidMixtureProperties	Thermophysical properties of liquid mixtures
basicSolidThermo	Thermophysical models of solids
solid	Thermodynamics of solid species
SLGThermo	Thermodynamic package for solids, liquids and gases
solidProperties	Thermophysical properties of solids
solidMixtureProperties	Thermophysical properties of solid mixtures
thermalPorousZone	Porous zone definition based on cell zones that includes
	terms for energy equations
solidMixtureProperties	Thermophysical properties of solid mixtures Porous zone definition based on cell zones that includes

Table 3.8: Libraries of thermophysical models.

RAS turbulence models for incompressible fluids — incompressibleRASModels	RAS turbulence	models for	incompressible	fluids - inco	ompressibleRASModels
---	----------------	------------	----------------	---------------	----------------------

laminar	Dummy turbulence model for laminar flow
kEpsilon	Standard high- $Re \ k - \varepsilon$ model
kOmega	Standard high- $Re \ k - \omega$ model
kOmegaSST	$k - \omega$ -SST model
-	Continued on next page

0 1 1	C		
Continued	from	previous	page

Commuted from previous pa	ige
RNGkEpsilon	RNG $k - \varepsilon$ model
NonlinearKEShih	Non-linear Shih $k - \varepsilon$ model
LienCubicKE	Lien cubic $k - \varepsilon$ model
qZeta	$q-\zeta$ model
LaunderSharmaKE	Launder-Sharma low- $Re \ k - \varepsilon$ model
LamBremhorstKE	Lam-Bremhorst low- $Re \ k - \varepsilon$ model
LienCubicKELowRe	Lien cubic low- $Re \ k - \varepsilon$ model
LienLeschzinerLowRe	Lien-Leschziner low- $Re \ k - \varepsilon$ model
LRR	Launder-Reece-Rodi RSTM
LaunderGibsonRSTM	Launder-Gibson RSTM with wall-reflection terms
realizableKE	Realizable $k - \varepsilon$ model
SpalartAllmaras	Spalart-Allmaras 1-eqn mixing-length model

#### RAS turbulence models for compressible fluids - compressible RASModels

laminar	Dummy turbulence model for laminar flow
kEpsilon	Standard $k - \varepsilon$ model
kOmegaSST	$k - \omega - SST \mod{1}$
RNGkEpsilon	RNG $k - \varepsilon$ model
LaunderSharmaKE	Launder-Sharma low- $Re \ k - \varepsilon$ model
LRR	Launder-Reece-Rodi RSTM
LaunderGibsonRSTM	Launder-Gibson RSTM
realizableKE	Realizable $k - \varepsilon$ model
SpalartAllmaras	Spalart-Allmaras 1-eqn mixing-length model

# Large-eddy simulation (LES) filters — LESfilters

laplaceFilter	Laplace filters
simpleFilter	Simple filter
anisotropicFilter	Anisotropic filter

#### Large-eddy simulation deltas — LESdeltas

PrandtlDelta	Prandtl delta
cubeRootVoIDeIta	Cube root of cell volume delta
maxDeltaxyz	Maximum of x, y and z; for structured hex cells only
smoothDelta	Smoothing of delta

## Incompressible LES turbulence models - incompressible LES Models

Smagorinsky	Smagorinsky model	
0		
Smagorinsky2	Smagorinsky model with 3-D filter	
dynSmagorinsky	Dynamic Smagorinsky	
homogenousDynSmag-	Homogeneous dynamic Smagorinsky model	
orinsky		
dynLagrangian	Lagrangian two equation eddy-viscosity model	
scaleSimilarity	Scale similarity model	
mixedSmagorinsky	Mixed Smagorinsky/scale similarity model	
dynMixedSmagorinsky	Dynamic mixed Smagorinsky/scale similarity model	
kOmegaSSTSAS	$k - \omega$ -SST scale adaptive simulation (SAS) model	
oneEqEddy	k-equation eddy-viscosity model	
dynOneEqEddy	Dynamic $k$ -equation eddy-viscosity model	
locDynOneEqEddy	Localised dynamic $k$ -equation eddy-viscosity model	
	Continued on next page	

#### Continued from previous page

spectEddyVisc	Spectral eddy viscosity model		
LRDDiffStress	LRR differential stress model		
DeardorffDiffStress	Deardorff differential stress model		
SpalartAllmaras	Spalart-Allmaras model		
SpalartAllmarasDDES	Spalart-Allmaras delayed detached eddy simulation		
	(DDES) model		
SpalartAllmarasIDDES	Spalart-Allmaras improved DDES (IDDES) model		

# 

	•
Smagorinsky	Smagorinsky model
oneEqEddy	k-equation eddy-viscosity model
dynOneEqEddy	Dynamic k-equation eddy-viscosity model
lowReOneEqEddy	Low- $Re\ k$ -equation eddy-viscosity model
DeardorffDiffStress	Deardorff differential stress model
SpalartAllmaras	Spalart-Allmaras 1-eqn mixing-length model

Table 3.9: Libraries of RAS and LES turbulence models.

Transport models for	$incompressible \ fluids incompressible Transport {\sf Models}$
Newtonian	Linear viscous fluid model
CrossPowerLaw	Cross Power law nonlinear viscous model
BirdCarreau	Bird-Carreau nonlinear viscous model
HerschelBulkley	Herschel-Bulkley nonlinear viscous model
powerLaw	Power-law nonlinear viscous model
interfaceProperties	Models for the interface, <i>e.g.</i> contact angle, in multiphase
	simulations

# Miscellaneous transport modelling libraries

interfaceProperties	Calculation of interface properties	
twoPhaseInterfaceProperties Two phase interface properties models, including boundary		
	conditions	
surfaceFilmModels	Surface film models	

Table 3.10: Shared object libraries of transport models.

# Chapter 4 OpenFOAM cases

This chapter deals with the file structure and organisation of OpenFOAM cases. Normally, a user would assign a name to a case, *e.g.* the tutorial case of flow in a cavity is simply named cavity. This name becomes the name of a directory in which all the case files and subdirectories are stored. The case directories themselves can be located anywhere but we recommend they are within a *run* subdirectory of the user's project directory, *i.e.\$HOME/OpenFOAM/\${USER}-2.1.0* as described at the beginning of chapter 2. One advantage of this is that the *\$FOAM\_RUN* environment variable is set to *\$HOME/OpenFOAM/\${USER}-2.1.0/run* by default; the user can quickly move to that directory by executing a preset alias, *run*, at the command line.

The tutorial cases that accompany the OpenFOAM distribution provide useful examples of the case directory structures. The tutorials are located in the *\$FOAM\_TUTORIALS* directory, reached quickly by executing the tut alias at the command line. Users can view tutorial examples at their leisure while reading this chapter.

# 4.1 File structure of OpenFOAM cases

The basic directory structure for a OpenFOAM case, that contains the minimum set of files required to run an application, is shown in Figure 4.1 and described as follows:



Figure 4.1: Case directory structure

- A constant directory that contains a full description of the case mesh in a subdirectory *polyMesh* and files specifying physical properties for the application concerned, *e.g.transportProperties*.
- A system directory for setting parameters associated with the solution procedure itself. It contains *at least* the following 3 files: *controlDict* where run control parameters are set including start/end time, time step and parameters for data output; *fvSchemes* where discretisation schemes used in the solution may be selected at run-time; and, *fvSolution* where the equation solvers, tolerances and other algorithm controls are set for the run.
- The 'time' directories containing individual files of data for particular fields. The data can be: either, initial values and boundary conditions that the user must specify to define the problem; or, results written to file by OpenFOAM. Note that the OpenFOAM fields must always be initialised, even when the solution does not strictly require it, as in steady-state problems. The name of each time directory is based on the simulated time at which the data is written and is described fully in section 4.3. It is sufficient to say now that since we usually start our simulations at time t = 0, the initial conditions are usually stored in a directory named 0 or 0.000000e+00, depending on the name format specified. For example, in the cavity tutorial, the velocity field U and pressure field p are initialised from files 0/U and 0/p respectively.

# 4.2 Basic input/output file format

OpenFOAM needs to read a range of data structures such as strings, scalars, vectors, tensors, lists and fields. The input/output (I/O) format of files is designed to be extremely flexible to enable the user to modify the I/O in OpenFOAM applications as easily as possible. The I/O follows a simple set of rules that make the files extremely easy to understand, in contrast to many software packages whose file format may not only be difficult to understand intuitively but also not be published anywhere. The OpenFOAM file format is described in the following sections.

# 4.2.1 General syntax rules

The format follows some general principles of C++ source code.

- Files have free form, with no particular meaning assigned to any column and no need to indicate continuation across lines.
- Lines have no particular meaning except to a // comment delimiter which makes OpenFOAM ignore any text that follows it until the end of line.
- $\bullet$  A comment over multiple lines is done by enclosing the text between /\* and \*/ delimiters.

# 4.2.2 Dictionaries

OpenFOAM uses *dictionaries* as the most common means of specifying data. A dictionary is an entity that contains data entries that can be retrieved by the I/O by means of *keywords*. The keyword entries follow the general format

Most entries are single data entries of the form:

```
<keyword> <dataEntry>;
```

Most OpenFOAM data files are themselves dictionaries containing a set of keyword entries. Dictionaries provide the means for organising entries into logical categories and can be specified hierarchically so that any dictionary can itself contain one or more dictionary entries. The format for a dictionary is to specify the dictionary name followed by keyword entries enclosed in curly braces {} as follows

```
<dictionaryName>
{
    ... keyword entries ...
}
```

# 4.2.3 The data file header

All data files that are read and written by OpenFOAM begin with a dictionary named FoamFile containing a standard set of keyword entries, listed in Table 4.1. The table

Keyword	Description	Entry
version	I/O format version	2.0
format	Data format	ascii / binary
location	Path to the file, in ""	(optional)
class	OpenFOAM class constructed from the	typically dictionary or a
	data file concerned	field, <i>e.g.</i> volVectorField
object	Filename	e.g.controlDict

Table 4.1: Header keywords entries for data files.

provides brief descriptions of each entry, which is probably sufficient for most entries with the notable exception of class. The class entry is the name of the C++ class in the OpenFOAM library that will be constructed from the data in the file. Without knowledge of the underlying code which calls the file to be read, and knowledge of the OpenFOAM classes, the user will probably be unable to surmise the class entry correctly. However, most data files with simple keyword entries are read into an internal dictionary class and therefore the class entry is dictionary in those cases.

The following example shows the use of keywords to provide data for a case using the types of entry described so far. The extract, from an *fvSolution* dictionary file, contains 2 dictionaries, *solvers* and *PISO*. The *solvers* dictionary contains multiple data entries for solver and tolerances for each of the pressure and velocity equations, represented by the p and U keywords respectively; the *PISO* dictionary contains algorithm controls.

 17

 18
 solvers

 19
 {

 20
 p

 21
 {

 22
 solver
 PCG;

 23
 preconditioner
 DIC;

 24
 tolerance
 1e-06;

```
relTol
                        0;
25
       }
26
27
       U
28
       Ĩ
29
                        PBiCG;
          solver
30
          preconditioner
                        DILU:
31
                        1e-05;
32
          tolerance
                        Ō;
33
          relTol
       }
34
   }
35
36
   PISO
37
38
   ł
       nCorrectors
                    2:
39
       nNonOrthogonalCorrectors 0;
40
       pRefCell
                    0;
0:
41
       pRefValue
42
   }
43
44
45
46
   11
```

## 4.2.4 Lists

OpenFOAM applications contain lists, e.g. a list of vertex coordinates for a mesh description. Lists are commonly found in I/O and have a format of their own in which the entries are contained within round braces (). There is also a choice of format preceeding the round braces:

simple the keyword is followed immediately by round braces

```
<listName>
(
... entries ...
);
```

**numbered** the keyword is followed by the number of elements  $\langle n \rangle$  in the list

```
<listName>
<n>
(
... entries ...
);
```

token identifier the keyword is followed by a class name identifier Label<Type> where <Type> states what the list contains, *e.g.* for a list of scalar elements is

```
<listName>
List<scalar>
<n> // optional
(
... entries ...
);
```

Note that *<scalar>* in *List<scalar>* is not a generic name but the actual text that should be entered.

The simple format is a convenient way of writing a list. The other formats allow the code to read the data faster since the size of the list can be allocated to memory in advance of reading the data. The simple format is therefore preferred for short lists, where read time is minimal, and the other formats are preferred for long lists.

# 4.2.5 Scalars, vectors and tensors

A scalar is a single number represented as such in a data file. A vector is a VectorSpace of rank 1 and dimension 3, and since the number of elements is always fixed to 3, the simple List format is used. Therefore a vector (1.0, 1.1, 1.2) is written:

(1.0 1.1 1.2)

In OpenFOAM, a tensor is a VectorSpace of rank 2 and dimension 3 and therefore the data entries are always fixed to 9 real numbers. Therefore the identity tensor can be written:

This example demonstrates the way in which OpenFOAM ignores the line return is so that the entry can be written over multiple lines. It is treated no differently to listing the numbers on a single line:

(10001001)

# 4.2.6 Dimensional units

In continuum mechanics, properties are represented in some chosen units, *e.g.* mass in kilograms (kg), volume in cubic metres (m<sup>3</sup>), pressure in Pascals (kg m<sup>-1</sup> s<sup>-2</sup>). Algebraic operations must be performed on these properties using consistent units of measurement; in particular, addition, subtraction and equality are only physically meaningful for properties of the same dimensional units. As a safeguard against implementing a meaningless operation, OpenFOAM attaches dimensions to field data and physical properties and performs dimension checking on any tensor operation.

The I/O format for a dimensionSet is 7 scalars delimited by square brackets, e.g.

[0 2 -1 0 0 0 0]

No.	Property	SI unit	USCS unit
1	Mass	kilogram (kg)	pound-mass (lbm)
2	Length	metre $(m)$	foot (ft)
3	Time	———— second	(s) — — — —
4	Temperature	Kelvin (K)	degree Rankine (°R)
5	Quantity	kilogram-mole (kgmol)	pound-mole (lbmol)
6	Current	ampere	e (A) — — — —
7	Luminous intensity	candela	(cd) ————

Table 4.2: Base units for SI and USCS

where each of the values corresponds to the power of each of the base units of measurement listed in Table 4.2. The table gives the base units for the Système International (SI) and the United States Customary System (USCS) but OpenFOAM can be used

with any system of units. All that is required is that the *input data is correct for the chosen set of units*. It is particularly important to recognise that OpenFOAM requires some dimensioned physical constants, *e.g.* the Universal Gas Constant *R*, for certain calculations, *e.g.* thermophysical modelling. These dimensioned constants are specified in a *DimensionedConstant* sub-dictionary of main *controlDict* file of the OpenFOAM installation (*\$WM\_PROJECT\_DIR/etc/controlDict*). By default these constants are set in SI units. Those wishing to use the USCS or any other system of units should modify these constants to their chosen set of units accordingly.

# 4.2.7 Dimensioned types

Physical properties are typically specified with their associated dimensions. These entries have the format that the following example of a dimensionedScalar demonstrates:

nu nu [0 2 -1 0 0 0 0] 1;

The first nu is the keyword; the second nu is the word name stored in class word, usually chosen to be the same as the keyword; the next entry is the dimensionSet and the final entry is the scalar value.

# 4.2.8 Fields

Much of the I/O data in OpenFOAM are tensor fields, e.g. velocity, pressure data, that are read from and written into the time directories. OpenFOAM writes field data using keyword entries as described in Table 4.3.

Keyword	Description	Example
dimensions	Dimensions of field	[1 1 -2 0 0 0 0]
internalField	Value of internal field	uniform (1 0 0)
boundaryField	Boundary field	see file listing in section $4.2.8$

Table 4.3: Main keywords used in field dictionaries.

The data begins with an entry for its dimensions. Following that, is the internalField, described in one of the following ways.

**Uniform field** a single value is assigned to all elements within the field, taking the form:

internalField uniform <entry>;

**Nonuniform field** each field element is assigned a unique value from a list, taking the following form where the token identifier form of list is recommended:

internalField nonuniform <List>;

The boundaryField is a dictionary containing a set of entries whose names correspond to each of the names of the boundary patches listed in the *boundary* file in the *polyMesh* directory. Each patch entry is itself a dictionary containing a list of keyword entries. The compulsory entry, type, describes the patch field condition specified for the field. The remaining entries correspond to the type of patch field condition selected and can
typically include field data specifying initial conditions on patch faces. A selection of patch field conditions available in OpenFOAM are listed in Table 5.3 and Table 5.4 with a description and the data that must be specified with it. Example field dictionary entries for velocity  $\tt U$  are shown below:

```
[0 \ 1 \ -1 \ 0 \ 0 \ 0];
    dimensions
17
18
                     uniform (0 0 0);
    internalField
19
20
    boundaryField
21
22
    ſ
        movingWall
23
24
         {
25
             type
                              fixedValue;
                              uniform (1 \ 0 \ 0);
             value
26
        }
27
28
        fixedWalls
29
30
                              fixedValue;
             type
31
                              uniform (0 \ 0 \ 0);
             value
32
        }
33
34
        frontAndBack
35
36
         Ł
             type
                              empty;
37
        }
38
    }
39
40
                                     41
```

## 4.2.9 Directives and macro substitutions

There is additional file syntax that offers great flexibility for the setting up of OpenFOAM case files, namely directives and macro substitutions. Directives are commands that can be contained within case files that begin with the hash (#) symbol. Macro substitutions begin with the dollar (\$) symbol.

At present there are 4 directive commands available in OpenFOAM:

- #inputMode has two options: merge, which merges keyword entries in successive dictionaries, so that a keyword entry specified in one place will be overridden by a later specification of the same keyword entry; overwrite, which overwrites the contents of an entire dictionary; generally, use merge;
- #remove <keywordEntry> removes any included keyword entry; can take a word or regular expression;

## 4.2.10 The #include and #inputMode directives

For example, let us say a user wishes to set an initial value of pressure once to be used as the internal field and initial value at a boundary. We could create a file, *e.g.* named *initialConditions*, which contains the following entries:

pressure 1e+05;
#inputMode merge

In order to use this pressure for both the internal and initial boundary fields, the user would simply include the following macro substitutions in the pressure field file p:

```
#include "initialConditions"
internalField uniform $pressure;
boundaryField
{
    patch1
    {
        type fixedValue;
        value $internalField;
    }
}
```

This is a fairly trivial example that simply demonstrates how this functionality works. However, the functionality can be used in many, more powerful ways particularly as a means of generalising case data to suit the user's needs. For example, if a user has a set of cases that require the same RAS turbulence model settings, a single file can be created with those settings which is simply included in the *RASProperties* file of each case. Macro substitutions can extend well beyond a single value so that, for example, sets of boundary conditions can be predefined and called by a single macro. The extent to which such functionality can be used is almost endless.

### 4.2.11 The #codeStream directive

The #codeStream directive takes C++ code which is compiled and executed to deliver the dictionary entry. The code and compilation instructions are specified through the following keywords.

- code: specifies the code, called with arguments OStream& os and const dictionary& dict which the user can use in the code, *e.g.* to lookup keyword entries from within the current case dictionary (file).
- codeInclude (optional): specifies additional C++ #include statements to include OpenFOAM files.
- codeOptions (optional): specifies any extra compilation flags to be added to EXE\_INC in *Make/options*.
- codeLibs (optional): specifies any extra compilation flags to be added to LIB\_LIBS in *Make/options*.

Code, like any string, can be written across multiple lines by enclosing it within hashbracket delimiters, *i.e.*  $\#\{\ldots,\#\}$ . Anything in between these two delimiters becomes a string with all newlines, quotes, *etc.* preserved.

An example of **#codeStream** is given below. The code in the *controlDict* file looks up dictionary entries and does a simple calculation for the write interval:

```
startTime 0;
endTime 100;
...
writeInterval #codeStream
{
    code
    #{
```

Open √FOAM-2.1.0

```
scalar start = readScalar(dict.lookup("startTime"));
scalar end = readScalar(dict.lookup("endTime"));
label nDumps = 5;
os << ((end - start)/nDumps);
#};
```

# 4.3 Time and data input/output control

The OpenFOAM solvers begin all runs by setting up a database. The database controls I/O and, since output of data is usually requested at intervals of time during the run, time is an inextricable part of the database. The *controlDict* dictionary sets input parameters *essential* for the creation of the database. The keyword entries in *controlDict* are listed in Table 4.4. Only the time control and writeInterval entries are truly compulsory, with the database taking default values indicated by † in Table 4.4 for any of the optional entries that are omitted.

Time control	
startFrom	Controls the start time of the simulation.
- firstTime	Earliest time step from the set of time directories.
- startTime	Time specified by the startTime keyword entry.
- latestTime	Most recent time step from the set of time directories.
startTime	Start time for the simulation with startFrom startTime;
stopAt	Controls the end time of the simulation.
- endTime	Time specified by the endTime keyword entry.
- writeNow	Stops simulation on completion of current time step and writes data.
- noWriteNow	Stops simulation on completion of current time step and does not write out data.
- nextWrite	Stops simulation on completion of next scheduled write time, spec- ified by writeControl.
endTime	End time for the simulation when stopAt endTime; is specified.
deltaT	Time step of the simulation.

### Data writing

writeControl	Controls the timing of write output to file.
- timeStep†	Writes data every writeInterval time steps.
- runTime	Writes data every writeInterval seconds of simulated time.
- adjustableRu	nTime Writes data every writeInterval seconds of simulated time,
	adjusting the time steps to coincide with the writeInterval if
	necessary — used in cases with automatic time step adjustment.
- cpuTime	Writes data every writeInterval seconds of CPU time.
- clockTime	Writes data out every writeInterval seconds of real time.
writeInterval	Scalar used in conjunction with writeControl described above.

Continued on next page

Continued from prepurgeWrite	Integer representing a limit on the number of time directories that are stored by overwriting time directories on a cyclic basis. Exam- ple of $t_0 = 5$ s, $\Delta t = 1$ s and purgeWrite 2;: data written into 2 directories, 6 and 7, before returning to write the data at 8 s in 6, data at 9 s into 7, etc. To disable the time directory limit, specify purgeWrite 0;† For steady-state solutions, results from previous iterations can be continuously overwritten by specifying purgeWrite 1;
writeFormat – ascii† – binary	Specifies the format of the data files. ASCII format, written to writePrecision significant figures. Binary format.
writePrecision	Integer used in conjunction with $\verb writeFormat $ described above, 6† by default
	n Specifies the compression of the data files. No compression.† gzip compression.
timeFormat - fixed - scientific - general†	Choice of format of the naming of the time directories. $\pm m.dddddd$ where the number of $ds$ is set by timePrecision. $\pm m.dddddde \pm xx$ where the number of $ds$ is set by timePrecision. Specifies scientific format if the exponent is less than -4 or greater than or equal to that specified by timePrecision.
timePrecision	Integer used in conjunction with $\verb"timeFormat"$ described above, 6 <sup>†</sup> by default
graphFormat - raw† - gnuplot - xmgr - jplot	Format for graph data written by an application. Raw ASCII format in columns. Data in gnuplot format. Data in Grace/xmgr format. Data in jPlot format.

### Data reading

runTimeModifiable yes†/no switch for whether dictionaries, *e.g.controlDict*, are reread by OpenFOAM at the beginning of each time step.

### Run-time loadable functionality

libs	List of additional libraries (on \$LD_LIBRARY_PATH) to be loaded
	at run-time, $e.g.$ ( "libUser1.so" "libUser2.so" )
functions	List of functions, <i>e.g.</i> <b>probes</b> to be loaded at run-time; see examples
	in <i>\$FOAM_TUTORIALS</i>

† denotes default entry if associated keyword is omitted.

Continued on next page

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Table 4.4: Keyword entries in the *controlDict* dictionary.

Example entries from a *controlDict* dictionary are given below:

```
17
    application
                    icoFoam;
18
19
    startFrom
                    startTime;
20
21
    startTime
                    0:
22
23
    stopAt
                    endTime;
24
25
26
    endTime
                    0.5;
27
    deltaT
                    0.005;
28
29
    writeControl
                    timeStep;
30
31
    writeInterval
                    20;
32
33
    purgeWrite
                    0;
34
35
    writeFormat
                    ascii;
36
37
    writePrecision
38
                    6:
39
    writeCompression off;
40
41
    timeFormat
                    general;
42
43
    timePrecision
                    6;
44
45
    runTimeModifiable true;
46
47
48
                            49
```

## 4.4 Numerical schemes

The fvSchemes dictionary in the *system* directory sets the numerical schemes for terms, such as derivatives in equations, that appear in applications being run. This section describes how to specify the schemes in the fvSchemes dictionary.

The terms that must typically be assigned a numerical scheme in *fvSchemes* range from derivatives, *e.g.* gradient  $\nabla$ , and interpolations of values from one set of points to another. The aim in OpenFOAM is to offer an unrestricted choice to the user. For example, while linear interpolation is effective in many cases, OpenFOAM offers complete freedom to choose from a wide selection of interpolation schemes for all interpolation terms.

The derivative terms further exemplify this freedom of choice. The user first has a choice of discretisation practice where standard Gaussian finite volume integration is the common choice. Gaussian integration is based on summing values on cell faces, which must be interpolated from cell centres. The user again has a completely free choice of interpolation scheme, with certain schemes being specifically designed for particular derivative terms, especially the convection divergence  $\nabla \cdot$  terms.

The set of terms, for which numerical schemes must be specified, are subdivided within the *fvSchemes* dictionary into the categories listed in Table 4.5. Each keyword in Table 4.5 is the name of a sub-dictionary which contains terms of a particular type, *e.g.gradSchemes* contains all the gradient derivative terms such as grad(p) (which represents  $\nabla p$ ). Further examples can be seen in the extract from an *fvSchemes* dictionary below:

```
17
18 ddtSchemes
19 {
20 default Euler;
21 }
```

Keyword	Category of mathematical terms
interpolationSchemes	Point-to-point interpolations of values
snGradSchemes	Component of gradient normal to a cell face
gradSchemes	Gradient $\nabla$
divSchemes	Divergence $\nabla \cdot$
laplacianSchemes	Laplacian $\nabla^2$
timeScheme	First and second time derivatives $\partial/\partial t$ , $\partial^2/\partial^2 t$
fluxRequired	Fields which require the generation of a flux

Table 4.5: Main keywords used in *fvSchemes*.

```
22
    gradSchemes
23
24
     {
         default
                            Gauss linear;
25
         grad(p)
                            Gauss linear;
26
    }
27
28
    divSchemes
29
30
         default
31
                            none:
         div(phi,U)
                            Gauss linear;
32
33
    }
34
    laplacianSchemes
35
36
    {
         default none;
laplacian(nu,U) Gauss linear corrected;
37
38
         laplacian((1|A(U)),p) Gauss linear corrected;
39
    }
40
41
    interpolationSchemes
42
43
    {
         default
                            linear;
44
         interpolate(HbyA) linear;
45
    }
46
47
    snGradSchemes
48
49
    ſ
         default
                            corrected;
50
    }
51
52
    fluxRequired
53
54
    {
         default
55
                            no;
56
         р
    }
57
58
59
                                                                     **********************//
60
```

The example shows that the fvSchemes dictionary contains the following:

- 6 ... Schemes subdictionaries containing keyword entries for each term specified within including: a default entry; other entries whose names correspond to a word identifier for the particular term specified, e.g.grad(p) for ∇p
- a *fluxRequired* sub-dictionary containing fields for which the flux is generated in the application, *e.g.*p in the example.

If a default scheme is specified in a particular ... Schemes sub-dictionary, it is assigned to all of the terms to which the sub-dictionary refers, *e.g.* specifying a default in grad-Schemes sets the scheme for all gradient terms in the application, *e.g.*  $\nabla p$ ,  $\nabla U$ . When a default is specified, it is not necessary to specify each specific term itself in that subdictionary, *i.e.* the entries for grad(p), grad(U) in this example. However, if any of these terms are included, the specified scheme overrides the default scheme for that term. Alternatively the user may insist on no default scheme by the none entry. In this instance the user is obliged to specify all terms in that sub-dictionary individually. Setting default to none may appear superfluous since default can be overridden. However, specifying none forces the user to specify all terms individually which can be useful to remind the user which terms are actually present in the application.

The following sections describe the choice of schemes for each of the categories of terms in Table 4.5.

## 4.4.1 Interpolation schemes

The *interpolationSchemes* sub-dictionary contains terms that are interpolations of values typically from cell centres to face centres. A *selection* of interpolation schemes in OpenFOAM are listed in Table 4.6, being divided into 4 categories: 1 category of general schemes; and, 3 categories of schemes used primarily in conjunction with Gaussian discretisation of convection (divergence) terms in fluid flow, described in section 4.4.5. It is *highly unlikely* that the user would adopt any of the convection-specific schemes for general field interpolations in the *interpolationSchemes* sub-dictionary, but, as valid interpolation schemes, they are described here rather than in section 4.4.5. Note that additional schemes such as UMIST are available in OpenFOAM but only those schemes that are generally recommended are listed in Table 4.6.

A general scheme is simply specified by quoting the keyword and entry, *e.g.* a linear scheme is specified as default by:

```
default linear;
```

The convection-specific schemes calculate the interpolation based on the flux of the flow velocity. The specification of these schemes requires the name of the flux field on which the interpolation is based; in most OpenFOAM applications this is **phi**, the name commonly adopted for the **surfaceScalarField** velocity flux  $\phi$ . The 3 categories of convection-specific schemes are referred to in this text as: general convection; normalised variable (NV); and, total variation diminishing (TVD). With the exception of the **blended** scheme, the general convection and TVD schemes are specified by the scheme and flux, *e.g.* an **upwind** scheme based on a flux **phi** is specified as **default** by:

```
default upwind phi;
```

Some TVD/NVD schemes require a coefficient  $\psi, 0 \leq \psi \leq 1$  where  $\psi = 1$  corresponds to TVD conformance, usually giving best convergence and  $\psi = 0$  corresponds to best accuracy. Running with  $\psi = 1$  is generally recommended. A limitedLinear scheme based on a flux phi with  $\psi = 1.0$  is specified as default by:

default limitedLinear 1.0 phi;

### 4.4.1.1 Schemes for strictly bounded scalar fields

There are enhanced versions of some of the limited schemes for scalars that need to be strictly bounded. To bound between user-specified limits, the scheme name should be preceded by the word limited and followed by the lower and upper limits respectively. For example, to bound the vanLeer scheme strictly between -2 and 3, the user would specify:

### default limitedVanLeer -2.0 3.0;

There are specialised versions of these schemes for scalar fields that are commonly bounded between 0 and 1. These are selected by adding 01 to the name of the scheme. For example, to bound the vanLeer scheme strictly between 0 and 1, the user would specify:

default vanLeer01;

Strictly bounded versions are available for the following schemes: limitedLinear, vanLeer, Gamma, limitedCubic, MUSCL and SuperBee.

### 4.4.1.2 Schemes for vector fields

There are improved versions of some of the limited schemes for vector fields in which the limiter is formulated to take into account the direction of the field. These schemes are selected by adding V to the name of the general scheme, *e.g.limitedLinearV* for limitedLinear. 'V' versions are available for the following schemes: limitedLinearV, vanLeerV, GammaV, limitedCubicV and SFCDV.

Centred schemes	
linear	Linear interpolation (central differencing)
cubicCorrection	Cubic scheme
midPoint	Linear interpolation with symmetric weighting
Upwinded convecti	on schemes
upwind	Upwind differencing
linearUpwind	Linear upwind differencing
skewLinear	Linear with skewness correction
filteredLinear2	Linear with filtering for high-frequency ringing
TVD schemes	
limitedLinear	limited linear differencing
vanLeer	van Leer limiter
MUSCL	MUSCL limiter
limitedCubic	Cubic limiter
NVD schemes	
SFCD	Self-filtered central differencing
$\texttt{Gamma}\ \psi$	Gamma differencing

Table 4.6: Interpolation schemes.

## 4.4.2 Surface normal gradient schemes

The *snGradSchemes* sub-dictionary contains surface normal gradient terms. A surface normal gradient is evaluated at a cell face; it is the component, normal to the face, of the gradient of values at the centres of the 2 cells that the face connects. A surface normal

gradient may be specified in its own right and is also required to evaluate a Laplacian term using Gaussian integration.

The available schemes are listed in Table 4.7 and are specified by simply quoting the keyword and entry, with the exception of limited which requires a coefficient  $\psi, 0 \le \psi \le 1$  where

		corresponds to uncorrected,	
a/ı —	0.333	non-orthogonal correction $\leq 0.5 \times$ orthogonal part,	$(1 \ 1)$
$\psi = \langle$	0.5	non-orthogonal correction $\leq 0.5 \times$ orthogonal part, non-orthogonal correction $\leq$ orthogonal part,	(4.1)
	1	corresponds to corrected.	

A limited scheme with  $\psi = 0.5$  is therefore specified as default by:

default limited 0.5;

Scheme	Description
corrected	Explicit non-orthogonal correction
uncorrected	No non-orthogonal correction
limited $\psi$	Limited non-orthogonal correction
bounded	Bounded correction for positive scalars
fourth	Fourth order

Table 4.7: Surface normal gradient schemes.

## 4.4.3 Gradient schemes

The *gradSchemes* sub-dictionary contains gradient terms. The discretisation scheme for each term can be selected from those listed in Table 4.8.

Discretisation scheme	Description
Gauss <interpolationscheme></interpolationscheme>	Second order, Gaussian integration
leastSquares	Second order, least squares
fourth	Fourth order, least squares
${\tt cellLimited} < {\tt gradScheme} >$	Cell limited version of one of the above schemes
faceLimited <gradscheme></gradscheme>	Face limited version of one of the above schemes

Table 4.8: Discretisation schemes available in *gradSchemes*.

The discretisation scheme is sufficient to specify the scheme completely in the cases of leastSquares and fourth, e.g.

grad(p) leastSquares;

The **Gauss** keyword specifies the standard finite volume discretisation of Gaussian integration which requires the interpolation of values from cell centres to face centres. Therefore, the **Gauss** entry must be followed by the choice of interpolation scheme from Table 4.6. It would be extremely unusual to select anything other than general interpolation schemes and in most cases the linear scheme is an effective choice, *e.g.* 

grad(p) Gauss linear;

Limited versions of any of the 3 base gradient schemes — Gauss, leastSquares and fourth — can be selected by preceding the discretisation scheme by cellLimited (or faceLimited), *e.g.* a cell limited Gauss scheme

grad(p) cellLimited Gauss linear 1;

### 4.4.4 Laplacian schemes

The *laplacianSchemes* sub-dictionary contains Laplacian terms. Let us discuss the syntax of the entry in reference to a typical Laplacian term found in fluid dynamics,  $\nabla \cdot (\nu \nabla \mathbf{U})$ , given the word identifier laplacian(nu,U). The Gauss scheme is the only choice of discretisation and requires a selection of both an interpolation scheme for the diffusion coefficient, *i.e.*  $\nu$  in our example, and a surface normal gradient scheme, *i.e.*  $\nabla \mathbf{U}$ . To summarise, the entries required are:

```
Gauss <interpolationScheme> <snGradScheme>
```

The interpolation scheme is selected from Table 4.6, the typical choices being from the general schemes and, in most cases, linear. The surface normal gradient scheme is selected from Table 4.7; the choice of scheme determines numerical behaviour as described in Table 4.9. A typical entry for our example Laplacian term would be:

laplacian(nu,U) Gauss linear corrected;

Scheme	Numerical behaviour
corrected	Unbounded, second order, conservative
uncorrected	Bounded, first order, non-conservative
limited $\psi$	Blend of corrected and uncorrected
bounded	First order for bounded scalars
fourth	Unbounded, fourth order, conservative

Table 4.9: Behaviour of surface normal schemes used in laplacianSchemes.

## 4.4.5 Divergence schemes

The *divSchemes* sub-dictionary contains divergence terms. Let us discuss the syntax of the entry in reference to a typical convection term found in fluid dynamics  $\nabla \cdot (\rho UU)$ , which in OpenFOAM applications is commonly given the identifier div(phi,U), where phi refers to the flux  $\phi = \rho U$ .

The **Gauss** scheme is the only choice of discretisation and requires a selection of the interpolation scheme for the dependent field, *i.e.*  $\mathbf{U}$  in our example. To summarise, the entries required are:

Gauss <interpolationScheme>

The interpolation scheme is selected from the full range of schemes in Table 4.6, both general and convection-specific. The choice critically determines numerical behaviour as described in Table 4.10. The syntax here for specifying convection-specific interpolation schemes *does not include the flux* as it is already known for the particular term, *i.e.* for div(phi,U), we know the flux is phi so specifying it in the interpolation scheme would only invite an inconsistency. Specification of upwind interpolation in our example would therefore be:

div(phi,U) Gauss upwind;

Scheme	Numerical behaviour
linear	Second order, unbounded
skewLinear	Second order, (more) unbounded, skewness correction
cubicCorrected	Fourth order, unbounded
upwind	First order, bounded
linearUpwind	First/second order, bounded
QUICK	First/second order, bounded
TVD schemes	First/second order, bounded
SFCD	Second order, bounded
NVD schemes	First/second order, bounded

Table 4.10: Behaviour of interpolation schemes used in *divSchemes*.

## 4.4.6 Time schemes

The first time derivative  $(\partial/\partial t)$  terms are specified in the *ddtSchemes* sub-dictionary. The discretisation scheme for each term can be selected from those listed in Table 4.11.

There is an off-centering coefficient  $\psi$  with the CrankNicholson scheme that blends it with the Euler scheme. A coefficient of  $\psi = 1$  corresponds to pure CrankNicholson and and  $\psi = 0$  corresponds to pure Euler. The blending coefficient can help to improve stability in cases where pure CrankNicholson are unstable.

Scheme	Description
Euler	First order, bounded, implicit
localEuler	Local-time step, first order, bounded, implicit
${\tt CrankNicholson}\;\psi$	Second order, bounded, implicit
backward	Second order, implicit
steadyState	Does not solve for time derivatives

Table 4.11: Discretisation schemes available in *ddtSchemes*.

When specifying a time scheme it must be noted that an application designed for transient problems will not necessarily run as steady-state and visa versa. For example the solution will not converge if steadyState is specified when running icoFoam, the transient, laminar incompressible flow code; rather, simpleFoam should be used for steady-state, incompressible flow.

Any second time derivative  $(\partial^2/\partial t^2)$  terms are specified in the *d2dt2Schemes* subdictionary. Only the Euler scheme is available for *d2dt2Schemes*.

## 4.4.7 Flux calculation

The *fluxRequired* sub-dictionary lists the fields for which the flux is generated in the application. For example, in many fluid dynamics applications the flux is generated after solving a pressure equation, in which case the *fluxRequired* sub-dictionary would simply be entered as follows, **p** being the **word** identifier for pressure:

```
fluxRequired
{
    p;
}
```

# 4.5 Solution and algorithm control

The equation solvers, tolerances and algorithms are controlled from the *fvSolution* dictionary in the *system* directory. Below is an example set of entries from the *fvSolution* dictionary required for the icoFoam solver.

```
17
   solvers
18
19
       р
{
20
21
           solver
                           PCG;
22
           preconditioner
                          DIC
23
                          1e-Ó6;
           tolerance
24
                          0;
25
           relTol
       }
26
27
       U
28
29
30
           solver
                          PBiCG;
           preconditioner
                          DILU;
31
           tolerance
                          1e-05;
32
                          0;
33
           relTol
       }
34
   }
35
36
   PISO
37
38
       nCorrectors
                       2;
39
       nNonOrthogonalCorrectors 0;
40
       pRefCell
                       0;
41
42
       pRefValue
                       0:
   }
43
44
45
   46
```

fvSolution contains a set of subdictionaries that are specific to the solver being run. However, there is a small set of standard subdictionaries that cover most of those used by the standard solvers. These subdictionaries include *solvers*, *relaxationFactors*, *PISO* and *SIMPLE* which are described in the remainder of this section.

## 4.5.1 Linear solver control

The first sub-dictionary in our example, and one that appears in all solver applications, is **solvers**. It specifies each linear-solver that is used for each discretised equation; it is emphasised that the term *linear*-solver refers to the method of number-crunching to solve the set of linear equations, as opposed to *application* solver which describes the set of equations and algorithms to solve a particular problem. The term 'linear-solver' is abbreviated to 'solver' in much of the following discussion; we hope the context of the term avoids any ambiguity.

The syntax for each entry within *solvers* uses a keyword that is the word relating to the variable being solved in the particular equation. For example, icoFoam solves equations for velocity  $\mathbf{U}$  and pressure p, hence the entries for  $\mathbf{U}$  and  $\mathbf{p}$ . The keyword is followed by a dictionary containing the type of solver and the parameters that the solver uses. The solver is selected through the solver keyword from the choice in OpenFOAM, listed in Table 4.12. The parameters, including tolerance, relTol, preconditioner, *etc.* are described in following sections.

Solver	Keyword
Preconditioned (bi-)conjugate gradient	PCG/PBiCG <sup>†</sup>
Solver using a smoother	${\tt smoothSolver}$
Generalised geometric-algebraic multi-grid	GAMG
Diagonal solver for explicit systems	diagonal
†PCG for symmetric matrices, PBiCG for asymmetric	

Table 4.12: Linear solvers.

The solvers distinguish between symmetric matrices and asymmetric matrices. The symmetry of the matrix depends on the structure of the equation being solved and, while the user may be able to determine this, it is not essential since OpenFOAM will produce an error message to advise the user if an inappropriate solver has been selected, e.g.

```
--> FOAM FATAL IO ERROR : Unknown asymmetric matrix solver PCG
Valid asymmetric matrix solvers are :
3
(
PBiCG
smoothSolver
GAMG
)
```

### 4.5.1.1 Solution tolerances

The sparse matrix solvers are iterative, *i.e.* they are based on reducing the equation residual over a succession of solutions. The residual is ostensibly a measure of the error in the solution so that the smaller it is, the more accurate the solution. More precisely, the residual is evaluated by substituting the current solution into the equation and taking the magnitude of the difference between the left and right hand sides; it is also normalised to make it independent of the scale of the problem being analysed.

Before solving an equation for a particular field, the initial residual is evaluated based on the current values of the field. After each solver iteration the residual is re-evaluated. The solver stops if *either* of the following conditions are reached:

- the residual falls below the *solver tolerance*, tolerance;
- the ratio of current to initial residuals falls below the *solver relative tolerance*, relTol;
- the number of iterations exceeds a *maximum number of iterations*, maxIter;

The solver tolerance should represent the level at which the residual is small enough that the solution can be deemed sufficiently accurate. The solver relative tolerance limits the relative improvement from initial to final solution. In transient simulations, it is usual to set the solver relative tolerance to 0 to force the solution to converge to the solver tolerance in each time step. The tolerances, tolerance and relTol must be specified in the dictionaries for all solvers; maxIter is optional.

### 4.5.1.2 Preconditioned conjugate gradient solvers

There are a range of options for preconditioning of matrices in the conjugate gradient solvers, represented by the **preconditioner** keyword in the solver dictionary. The preconditioners are listed in Table 4.13.

Preconditioner	Keyword
Diagonal incomplete-Cholesky (symmetric)	DIC
Faster diagonal incomplete-Cholesky (DIC with caching)	FDIC
Diagonal incomplete-LU (asymmetric)	DILU
Diagonal	diagonal
Geometric-algebraic multi-grid	GAMG
No preconditioning	none

Table 4.13: Preconditioner options.

### 4.5.1.3 Smooth solvers

The solvers that use a smoother require the smoother to be specified. The smoother options are listed in Table 4.14. Generally GaussSeidel is the most reliable option, but for bad matrices DIC can offer better convergence. In some cases, additional post-smoothing using GaussSeidel is further beneficial, *i.e.* the method denoted as DICGaussSeidel

Smoother	Keyword
Gauss-Seidel	GaussSeidel
Diagonal incomplete-Cholesky (symmetric)	DIC
Diagonal incomplete-Cholesky with Gauss-Seidel (symmetric)	DICGaussSeidel

Table 4.14: Smoother options.

The user must also pecify the number of sweeps, by the **nSweeps** keyword, before the residual is recalculated, following the tolerance parameters.

### 4.5.1.4 Geometric-algebraic multi-grid solvers

The generalised method of geometric-algebraic multi-grid (GAMG) uses the principle of: generating a quick solution on a mesh with a small number of cells; mapping this solution onto a finer mesh; using it as an initial guess to obtain an accurate solution on the fine mesh. GAMG is faster than standard methods when the increase in speed by solving first on coarser meshes outweighs the additional costs of mesh refinement and mapping of field data. In practice, GAMG starts with the mesh specified by the user and coarsens/refines the mesh in stages. The user is only required to specify an approximate mesh size at the most coarse level in terms of the number of cells nCoarsestCells.

The agglomeration of cells is performed by the algorithm specified by the agglomerator keyword. Presently we recommend the faceAreaPair method. It is worth noting there is

an MGridGen option that requires an additional entry specifying the shared object library for MGridGen:

```
geometricGamgAgglomerationLibs ("libMGridGenGamgAgglomeration.so");
```

In the experience of OpenCFD, the MGridGen method offers no obvious benefit over the faceAreaPair method. For all methods, agglomeration can be optionally cached by the cacheAgglomeration switch.

Smoothing is specified by the smoother as described in section 4.5.1.3. The number of sweeps used by the smoother at different levels of mesh density are specified by the nPreSweeps, nPostSweeps and nFinestSweeps keywords. The nPreSweeps entry is used as the algorithm is coarsening the mesh, nPostSweeps is used as the algorithm is refining, and nFinestSweeps is used when the solution is at its finest level.

The mergeLevels keyword controls the speed at which coarsening or refinement levels is performed. It is often best to do so only at one level at a time, *i.e.* set mergeLevels 1. In some cases, particularly for simple meshes, the solution can be safely speeded up by coarsening/refining two levels at a time, *i.e.* setting mergeLevels 2.

### 4.5.2 Solution under-relaxation

A second sub-dictionary of *fvSolution* that is often used in OpenFOAM is *relaxationFactors* which controls under-relaxation, a technique used for improving stability of a computation, particularly in solving steady-state problems. Under-relaxation works by limiting the amount which a variable changes from one iteration to the next, either by modifying the solution matrix and source prior to solving for a field or by modifying the field directly. An under-relaxation factor  $\alpha, 0 < \alpha \leq 1$  specifies the amount of under-relaxation, ranging from none at all for  $\alpha = 1$  and increasing in strength as  $\alpha \to 0$ . The limiting case where  $\alpha = 0$  represents a solution which does not change at all with successive iterations. An optimum choice of  $\alpha$  is one that is small enough to ensure stable computation but large enough to move the iterative process forward quickly; values of  $\alpha$  as high as 0.9 can ensure stability in some cases and anything much below, say, 0.2 are prohibitively restrictive in slowing the iterative process.

The user can specify the relaxation factor for a particular field by specifying first the word associated with the field, then the factor. The user can view the relaxation factors used in a tutorial example of simpleFoam for incompressible, laminar, steady-state flows.

```
17
18
     solvers
     ſ
19
          р
{
20
21
                                      PCG;
DIC;
1e-06;
                solver
22
                preconditioner
tolerance
23
24
                                      0.01;
25
                relTol
           }
26
27
           ñ
28
           ſ
29
                                      PBiCG;
                solver
30
                                      DILU;
                preconditioner
31
                                      1e-05;
32
                tolerance
                                      0.1;
33
                relTol
           }
34
35
          k
{
36
37
                                      PBiCG;
                solver
38
                preconditioner
                                      DILU;
39
                                      1e-05;
                tolerance
40
                                      0.1;
                relTol
41
           }
42
```

```
43
         epsilon
44
          ł
45
              solver
                                 PBiCG;
46
                                 DILU;
              preconditioner
47
                                 1e-05;
              tolerance
48
              relTol
                                 0.1:
49
         }
50
51
         R
{
52
53
                                 PBiCG;
              solver
54
              preconditioner
                                 DILU;
55
                                 1e-05;
              tolerance
56
              relTol
                                 0.1:
57
         }
58
59
         nuTilda
60
61
                                 PBiCG;
              solver
62
                                 DILU;
1e-05;
63
              preconditioner
64
              tolerance
              relTol
                                 0.1;
65
         }
66
    }
67
68
    SIMPLE
69
70
         nNonOrthogonalCorrectors 0;
71
72
         residualControl
73
74
                                 1e-2;
75
              P
               1e-3;
(k|epsilon|omega)" 1e-3;
76
77
         }
78
    }
79
80
    relaxationFactors
81
82
         fields
83
         {
84
                                 0.3;
              р
85
         7
86
         equations
87
         ł
88
                                 0.7;
0.7;
0.7;
0.7;
0.7;
89
              k
90
              epsilon
91
              R
92
              nuTilda
93
         }
94
    }
95
96
97
```

#### 4.5.3**PISO and SIMPLE algorithms**

Most fluid dynamics solver applications in OpenFOAM use the pressure-implicit splitoperator (PISO) or semi-implicit method for pressure-linked equations (SIMPLE) algorithms. These algorithms are iterative procedures for solving equations for velocity and pressure, PISO being used for transient problems and SIMPLE for steady-state.

Both algorithms are based on evaluating some initial solutions and then correcting them. SIMPLE only makes 1 correction whereas PISO requires more than 1, but typically not more than 4. The user must therefore specify the number of correctors in the PISO dictionary by the nCorrectors keyword as shown in the example on page U-120.

An additional correction to account for mesh non-orthogonality is available in both SIMPLE and PISO in the standard OpenFOAM solver applications. A mesh is orthogonal if, for each face within it, the face normal is parallel to the vector between the centres of the cells that the face connects, e.q. a mesh of hexahedral cells whose faces are aligned with a Cartesian coordinate system. The number of non-orthogonal correctors is specified by the nNonOrthogonalCorrectors keyword as shown in the examples above and on

98

page U-120. The number of non-orthogonal correctors should correspond to the mesh for the case being solved, *i.e.* 0 for an orthogonal mesh and increasing with the degree of non-orthogonality up to, say, 20 for the most non-orthogonal meshes.

### 4.5.3.1 Pressure referencing

In a closed incompressible system, pressure is relative: it is the pressure range that matters not the absolute values. In these cases, the solver sets a reference level of pRefValue in cell pRefCell where p is the name of the pressure solution variable. Where the pressure is  $p\_rgh$ , the names are  $p\_rhgRefValue$  and  $p\_rhgRefCell$  respectively. These entries are generally stored in the *PISO/SIMPLE* sub-dictionary and are used by those solvers that require them when the case demands it. If ommitted, the solver will not run, but give a message to alert the user to the problem.

## 4.5.4 Other parameters

The *fvSolutions* dictionaries in the majority of standard OpenFOAM solver applications contain no other entries than those described so far in this section. However, in general the *fvSolution* dictionary may contain any parameters to control the solvers, algorithms, or in fact anything. For a given solver, the user can look at the source code to find the parameters required. Ultimately, if any parameter or sub-dictionary is missing when an solver is run, it will terminate, printing a detailed error message. The user can then add missing parameters accordingly.

# Chapter 5

# Mesh generation and conversion

This chapter describes all topics relating to the creation of meshes in OpenFOAM: section 5.1 gives an overview of the ways a mesh may be described in OpenFOAM; section 5.3 covers the blockMesh utility for generating simple meshes of blocks of hexahedral cells; section 5.4 covers the snappyHexMesh utility for generating complex meshes of hexahedral and split-hexahedral cells automatically from triangulated surface geometries; section 5.5 describes the options available for conversion of a mesh that has been generated by a third-party product into a format that OpenFOAM can read.

### 5.1 Mesh description

This section provides a specification of the way the OpenFOAM C++ classes handle a mesh. The mesh is an integral part of the numerical solution and must satisfy certain criteria to ensure a valid, and hence accurate, solution. During any run, OpenFOAM checks that the mesh satisfies a fairly stringent set of validity constraints and will cease running if the constraints are not satisfied. The consequence is that a user may experience some frustration in 'correcting' a large mesh generated by third-party mesh generators before OpenFOAM will run using it. This is unfortunate but we make no apology for OpenFOAM simply adopting good practice to ensure the mesh is valid; otherwise, the solution is flawed before the run has even begun.

By default OpenFOAM defines a mesh of arbitrary polyhedral cells in 3-D, bounded by arbitrary polygonal faces, *i.e.* the cells can have an unlimited number of faces where, for each face, there is no limit on the number of edges nor any restriction on its alignment. A mesh with this general structure is known in OpenFOAM as a **polyMesh**. This type of mesh offers great freedom in mesh generation and manipulation in particular when the geometry of the domain is complex or changes over time. The price of absolute mesh generality is, however, that it can be difficult to convert meshes generated using conventional tools. The OpenFOAM library therefore provides **cellShape** tools to manage conventional mesh formats based on sets of pre-defined cell shapes.

### 5.1.1 Mesh specification and validity constraints

Before describing the OpenFOAM mesh format, polyMesh, and the cellShape tools, we will first set out the validity constraints used in OpenFOAM. The conditions that a mesh must satisfy are:

### 5.1.1.1 Points

A point is a location in 3-D space, defined by a vector in units of metres (m). The points are compiled into a list and each point is referred to by a label, which represents its position in the list, starting from zero. The point list cannot contain two different points at an exactly identical position nor any point that is not part at least one face.

### 5.1.1.2 Faces

A face is an ordered list of points, where a point is referred to by its label. The ordering of point labels in a face is such that each two neighbouring points are connected by an edge, *i.e.* you follow points as you travel around the circumference of the face. Faces are compiled into a list and each face is referred to by its label, representing its position in the list. The direction of the face normal vector is defined by the right-hand rule, *i.e.* looking towards a face, if the numbering of the points follows an anti-clockwise path, the normal vector points towards you, as shown in Figure 5.1.



Figure 5.1: Face area vector from point numbering on the face

There are two types of face:

- **Internal faces** Those faces that connect two cells (and it can never be more than two). For each internal face, the ordering of the point labels is such that the face normal points into the cell with the larger label, *i.e.* for cells 2 and 5, the normal points into 5;
- **Boundary faces** Those belonging to one cell since they coincide with the boundary of the domain. A boundary face is therefore addressed by one cell(only) and a boundary patch. The ordering of the point labels is such that the face normal points outside of the computational domain.

Faces are generally expected to be convex; at the very least the face centre needs to be inside the face. Faces are allowed to be warped, i.e. not all points of the face need to be coplanar.

### 5.1.1.3 Cells

A cell is a list of faces in arbitrary order. Cells must have the properties listed below.

**Contiguous** The cells must completely cover the computational domain and must not overlap one another.

Convex Every cell must be convex and its cell centre inside the cell.

**Closed** Every cell must be *closed*, both geometrically and topologically where:

- geometrical closedness requires that when all face area vectors are oriented to point outwards of the cell, their sum should equal the zero vector to machine accuracy;
- topological closedness requires that all the edges in a cell are used by exactly two faces of the cell in question.
- **Orthogonality** For all internal faces of the mesh, we define the centre-to-centre vector as that connecting the centres of the 2 cells that it adjoins oriented from the centre of the cell with smaller label to the centre of the cell with larger label. The orthogonality constraint requires that for each internal face, the angle between the face area vector, oriented as described above, and the centre-to-centre vector must always be less than 90°.

### 5.1.1.4 Boundary

A boundary is a list of patches, each of which is associated with a boundary condition. A patch is a list of face labels which clearly must contain only boundary faces and no internal faces. The boundary is required to be closed, *i.e.* the sum all boundary face area vectors equates to zero to machine tolerance.

## 5.1.2 The polyMesh description

The *constant* directory contains a full description of the case polyMesh in a subdirectory *polyMesh*. The polyMesh description is based around faces and, as already discussed, internal cells connect 2 cells and boundary faces address a cell and a boundary patch. Each face is therefore assigned an 'owner' cell and 'neighbour' cell so that the connectivity across a given face can simply be described by the owner and neighbour cell labels. In the case of boundaries, the connected cell is the owner and the neighbour is assigned the label '-1'. With this in mind, the I/O specification consists of the following files:

- *points* a list of vectors describing the cell vertices, where the first vector in the list represents vertex 0, the second vector represents vertex 1, *etc.*;
- *faces* a list of faces, each face being a list of indices to vertices in the points list, where again, the first entry in the list represents face 0, *etc.*;
- *owner* a list of owner cell labels, the index of entry relating directly to the index of the face, so that the first entry in the list is the owner label for face 0, the second entry is the owner label for face 1, etc;

neighbour a list of neighbour cell labels;

**boundary** a list of patches, containing a dictionary entry for each patch, declared using the patch name, *e.g.* 

```
movingWall
{
    type patch;
    nFaces 20;
    startFace 760;
```

}

The startFace is the index into the face list of the first face in the patch, and nFaces is the number of faces in the patch.

Note that if the user wishes to know how many cells are in their domain, there is a note in the FoamFile header of the owner file that contains an entry for nCells.

## 5.1.3 The cellShape tools

We shall describe the alternative **cellShape** tools that may be used particularly when converting some standard (simpler) mesh formats for the use with OpenFOAM library.

The vast majority of mesh generators and post-processing systems support only a fraction of the possible polyhedral cell shapes in existence. They define a mesh in terms of a limited set of 3D cell geometries, referred to as *cell shapes*. The OpenFOAM library contains definitions of these standard shapes, to enable a conversion of such a mesh into the **polyMesh** format described in the previous section.

The cellShape models supported by OpenFOAM are shown in Table 5.1. The shape is defined by the ordering of point labels in accordance with the numbering scheme contained in the shape model. The ordering schemes for points, faces and edges are shown in Table 5.1. The numbering of the points must not be such that the shape becomes twisted or degenerate into other geometries, *i.e.* the same point label cannot be used more that once is a single shape. Moreover it is unnecessary to use duplicate points in OpenFOAM since the available shapes in OpenFOAM cover the full set of degenerate hexahedra.

The cell description consists of two parts: the name of a cell model and the ordered list of labels. Thus, using the following list of points

```
8
(
(0 0 0)
(1 0 0)
(1 1 0)
(0 1 0)
(0 0 0.5)
(1 0 0.5)
(1 1 0.5)
(0 1 0.5)
)
```

A hexahedral cell would be written as:

(hex 8(0 1 2 3 4 5 6 7))

Here the hexahedral cell shape is declared using the keyword **hex**. Other shapes are described by the keywords listed in Table 5.1.

## 5.1.4 1- and 2-dimensional and axi-symmetric problems

OpenFOAM is designed as a code for 3-dimensional space and defines all meshes as such. However, 1- and 2- dimensional and axi-symmetric problems can be simulated in OpenFOAM by generating a mesh in 3 dimensions and applying special boundary conditions on any patch in the plane(s) normal to the direction(s) of interest. More specifically, 1- and 2- dimensional problems use the **empty** patch type and axi-symmetric problems use the **wedge** type. The use of both are described in section 5.2.2 and the generation of wedge geometries for axi-symmetric problems is discussed in section 5.3.3.



Table 5.1: Vertex, face and edge numbering for cellShapes.

## 5.2 Boundaries

In this section we discuss the way in which boundaries are treated in OpenFOAM. The subject of boundaries is a little involved because their role in modelling is not simply that of a geometric entity but an integral part of the solution and numerics through boundary conditions or inter-boundary 'connections'. A discussion of boundaries sits uncomfortably between a discussion on meshes, fields, discretisation, computational processing *etc.* Its placement in this Chapter on meshes is a choice of convenience.

We first need to consider that, for the purpose of applying boundary conditions, a boundary is generally broken up into a set of *patches*. One patch may include one or more enclosed areas of the boundary surface which do not necessarily need to be physically connected.

There are three attributes associated with a patch that are described below in their natural hierarchy and Figure 5.2 shows the names of different patch types introduced at each level of the hierarchy. The hierarchy described below is very similar, but not identical, to the class hierarchy used in the OpenFOAM library.

- **Base type** The type of patch described purely in terms of geometry or a data 'communication link'.
- **Primitive type** The base numerical patch condition assigned to a field variable on the patch.
- **Derived type** A complex patch condition, derived from the primitive type, assigned to a field variable on the patch.



Figure 5.2: Patch attributes

## 5.2.1 Specification of patch types in OpenFOAM

The patch types are specified in the mesh and field files of a OpenFOAM case. More precisely:

• the base type is specified under the type keyword for each patch in the *boundary* file, located in the *constant/polyMesh* directory;

• the numerical patch type, be it a primitive or derived type, is specified under the type keyword for each patch in a field file.

An example *boundary* file is shown below for a sonicFoam case, followed by a pressure field file, p, for the same case:

```
17
     6
18
19
     (
          inlet
20
          {
21
                                  patch;
50;
10325;
22
               type
              nFaces
23
24
              startFace
          7
25
          outlet
26
27
          {
                                  patch;
40;
10375;
^{28}
               type
              nFaces
29
              startFace
30
          }
31
          bottom
32
          ł
33
                                  symmetryPlane;
25;
              type
nFaces
34
35
                                  10415;
36
              startFace
          }
37
          top
38
39
          {
                                  symmetryPlane;
125;
10440;
40
               type
              nFaces
41
              startFace
42
          }
43
          obstacle
44
45
          {
                                  patch;
110;
10565;
              type
nFaces
46
47
              startFace
48
49
          }
          defaultFaces
50
51
          Ł
                                  empty;
10500;
52
               type
53
              nFaces
54
               startFace
                                  10675;
          }
55
     )
56
57
     58
                         [1 -1 -2 0 0 0 0];
     dimensions
17
18
     internalField
                        uniform 1;
19
20
     boundaryField
21
22
     {
          inlet
23
          {
24
              type
value
25
                                  fixedValue;
                                  uniform 1;
26
          }
27
28
29
          outlet
30
          {
              type
field
31
                                  waveTransmissive;
32
                                  p;
                                  phi;
33
              phi
34
              rho
                                  rho;
                                  psi;
1.4;
35
              psi
              gamma
fieldInf
36
                                  1;
3;
37
38
               lInf
                                  uniform 1;
39
               value
          }
40
41
42
          bottom
43
          ł
                                  symmetryPlane;
44
               type
          }
45
46
          top
47
48
          {
                                  symmetryPlane;
49
               type
          }
50
```

```
51
      obstacle
52
53
         type
                     zeroGradient;
54
      }
55
56
      defaultFaces
57
58
59
         type
                     empty;
60
      }
   }
61
62
                     63
   11
```

The type in the boundary file is patch for all patches except those that have some geometrical constraint applied to them, *i.e.* the symmetryPlane and empty patches. The p file includes primitive types applied to the inlet and bottom faces, and a more complex derived type applied to the outlet. Comparison of the two files shows that the base and numerical types are consistent where the base type is not a simple patch, *i.e.* for the symmetryPlane and empty patches.

## 5.2.2 Base types

The base and geometric types are described below; the keywords used for specifying these types in OpenFOAM are summarised in Table 5.2.



Figure 5.3: Axi-symmetric geometry using the wedge patch type.

Selection Key	Description
patch	generic patch
symmetryPlane	plane of symmetry
empty	front and back planes of a 2D geometry
wedge	wedge front and back for an axi-symmetric geometry
cyclic	cyclic plane
wall	wall — used for wall functions in turbulent flows
processor	inter-processor boundary

Table 5.2: Basic patch types.

- patch The basic patch type for a patch condition that contains no geometric or topological information about the mesh (with the exception of wall), *e.g.* an inlet or an outlet.
- wall There are instances where a patch that coincides with a wall needs to be identifiable as such, particularly where specialist modelling is applied at wall boundaries. A good example is wall turbulence modelling where a wall must be specified with a wall patch type, so that the distance from the wall to the cell centres next to the wall are stored as part of the patch.

 $\mathsf{symmetryPlane}\xspace$  For a symmetry plane.

- empty While OpenFOAM always generates geometries in 3 dimensions, it can be instructed to solve in 2 (or 1) dimensions by specifying a special empty condition on each patch whose plane is normal to the 3rd (and 2nd) dimension for which no solution is required.
- wedge For 2 dimensional axi-symmetric cases, *e.g.* a cylinder, the geometry is specified as a wedge of small angle  $(e.g. < 5^{\circ})$  and 1 cell thick running along the plane of symmetry, straddling one of the coordinate planes, as shown in Figure 5.3. The axi-symmetric wedge planes must be specified as separate patches of wedge type. The details of generating wedge-shaped geometries using blockMesh are described in section 5.3.3.
- cyclic Enables two patches to be treated as if they are physically connected; used for repeated geometries, *e.g.* heat exchanger tube bundles. One cyclic patch is linked to another through a neighbourPatch keyword in the *boundary* file. Each pair of connecting faces must have similar area to within a tolerance given by the matchTolerance keyword in the *boundary* file. Faces do not need to be of the same orientation.
- processor If a code is being run in parallel, on a number of processors, then the mesh must be divided up so that each processor computes on roughly the same number of cells. The boundaries between the different parts of the mesh are called processor boundaries.

## 5.2.3 Primitive types

The primitive types are listed in Table 5.3.

## 5.2.4 Derived types

There are numerous derived types of boundary conditions in OpenFOAM, too many to list here. Instead a small selection is listed in Table 5.4. If the user wishes to obtain a list of all available models, they should consult the OpenFOAM source code. Derived boundary condition source code can be found at the following locations:

- in \$FOAM\_SRC/finiteVolume/fields/fvPatchFields/derived
- within certain model libraries, that can be located by typing the following command in a terminal window

find \$FOAM\_SRC -name "\*derivedFvPatch\*"

Type	Description of condition for patch field $\phi$	Data to specify
fixedValue	Value of $\phi$ is specified	value
fixedGradient	Normal gradient of $\phi$ is specified	gradient
zeroGradient	Normal gradient of $\phi$ is zero	
calculated	Boundary field $\phi$ derived from other fields	
mixed	Mixed fixedValue/ fixedGradient condition depend-	refValue,
	ing on the value in valueFraction	refGradient,
		valueFraction,
		value
directionMixed	A mixed condition with tensorial valueFraction,	refValue,
	e.g. for different levels of mixing in normal and	refGradient,
	tangential directions	valueFraction,
		value

Table 5.3: Primitive patch field types.

• within certain solvers, that can be located by typing the following command in a terminal window

find \$FOAM\_SOLVERS -name "\*fvPatch\*"

## 5.3 Mesh generation with the blockMesh utility

This section describes the mesh generation utility, **blockMesh**, supplied with OpenFOAM. The **blockMesh** utility creates parametric meshes with grading and curved edges.

The mesh is generated from a dictionary file named *blockMeshDict* located in the *constant/polyMesh* directory of a case. **blockMesh** reads this dictionary, generates the mesh and writes out the mesh data to *points* and *faces*, *cells* and *boundary* files in the same directory.

The principle behind **blockMesh** is to decompose the domain geometry into a set of 1 or more three dimensional, hexahedral blocks. Edges of the blocks can be straight lines, arcs or splines. The mesh is ostensibly specified as a number of cells in each direction of the block, sufficient information for **blockMesh** to generate the mesh data.

Each block of the geometry is defined by 8 vertices, one at each corner of a hexahedron. The vertices are written in a list so that each vertex can be accessed using its label, remembering that OpenFOAM always uses the C++ convention that the first element of the list has label '0'. An example block is shown in Figure 5.4 with each vertex numbered according to the list. The edge connecting vertices 1 and 5 is curved to remind the reader that curved edges can be specified in **blockMesh**.

It is possible to generate blocks with less than 8 vertices by collapsing one or more pairs of vertices on top of each other, as described in section 5.3.3.

Each block has a local coordinate system  $(x_1, x_2, x_3)$  that must be right-handed. A right-handed set of axes is defined such that to an observer looking down the Oz axis, with O nearest them, the arc from a point on the Ox axis to a point on the Oy axis is in a clockwise sense.

The local coordinate system is defined by the order in which the vertices are presented in the block definition according to:

• the axis origin is the first entry in the block definition, vertex 0 in our example;

Types derived from $fixedVa$	alue	Data to specify
movingWallVelocity	Replaces the normal of the patch value so the flux across the patch is zero	value
pressureInletVelocity	When $p$ is known at inlet, <b>U</b> is evaluated from the flux, normal to the patch	value
pressureDirectedInletVelocity	yWhen $p$ is known at inlet, <b>U</b> is calculated from the flux in the inletDirection	value,
		inletDirection
surfaceNormalFixedValue	Specifies a vector boundary condition, normal to the patch, by its magnitude; +ve for vectors pointing out of the domain	value
totalPressure	Total pressure $p_0 = p + \frac{1}{2}\rho  \mathbf{U} ^2$ is fixed; when <b>U</b> changes, p is adjusted accordingly	p0
turbulentInlet	Calculates a fluctuating variable based on a scale of a mean value	referenceField,
		fluctuationScale
Types derived from fixedGr	radient/zeroGradient	
fluxCorrectedVelocity	Calculates normal component of $\mathbf{U}$ at inlet from flux	value
wallBuoyantPressure	Sets $fixedGradient$ pressure based on the atmospheric pressure gradient	
Types derived from mixed		
inletOutlet	Switches $\mathbf{U}$ and $p$ between fixedValue and zeroGradient depending on direction of $\mathbf{U}$	inletValue, valu
outletInlet	Switches U and $p$ between fixedValue and zeroGradient depending on direction of U	outletValue, value
pressureInletOutletVelocity	Combination of pressureInletVelocity and inletOutlet	value
pressureDirected-	Combination of pressureDirectedInletVelocity and inletOutlet	value,
InletOutletVelocity		inletDirection
pressureTransmissive	Transmits supersonic pressure waves to surrounding pressure $p_{\infty}$	pInf
supersonicFreeStream	Transmits oblique shocks to surroundings at $p_{\infty}, T_{\infty}, \mathbf{U}_{\infty}$	pInf, TInf, UInf
Other types		
slip	zeroGradient if $\phi$ is a scalar; if $\phi$ is a vector, normal component is fixedValue zero,	
	tangential components are zeroGradient	

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- the  $x_1$  direction is described by moving from vertex 0 to vertex 1;
- the  $x_2$  direction is described by moving from vertex 1 to vertex 2;
- vertices 0, 1, 2, 3 define the plane  $x_3 = 0$ ;
- vertex 4 is found by moving from vertex 0 in the  $x_3$  direction;
- vertices 5,6 and 7 are similarly found by moving in the  $x_3$  direction from vertices 1,2 and 3 respectively.



Figure 5.4: A single block

Keyword	Description	Example/selection
convertToMeters	Scaling factor for the vertex	0.001 scales to mm
	coordinates	
vertices	List of vertex coordinates	(0 0 0)
edges	Used to describe <b>arc</b> or	arc 1 4 (0.939 0.342 -0.5)
	t spline edges	
block	Ordered list of vertex labels	hex (0 1 2 3 4 5 6 7)
	and mesh size	(10 10 1)
		simpleGrading (1.0 1.0 1.0)
patches	List of patches	symmetryPlane base
		((0123))
mergePatchPairs	List of patches to be merged	see section 5.3.2

Table 5.5: Keywords used in *blockMeshDict*.

## 5.3.1 Writing a blockMeshDict file

The *blockMeshDict* file is a dictionary using keywords described in Table 5.5. The convertToMeters keyword specifies a scaling factor by which all vertex coordinates in the mesh description are multiplied. For example,

convertToMeters 0.001;

means that all coordinates are multiplied by 0.001, *i.e.* the values quoted in the *blockMesh-Dict* file are in mm.

### 5.3.1.1 The vertices

The vertices of the blocks of the mesh are given next as a standard list named vertices, *e.g.* for our example block in Figure 5.4, the vertices are:

```
vertices
(
    ( 0
                         // vertex number 0
           0
                0)
                0.1)
                         // vertex number 1
    (1
           0
    (1.1
                0.1)
                         // vertex number 2
           1
    ( 0
                0.1)
                         // vertex number 3
           1
    (-0.1 - 0.1)
                1)
                         // vertex number 4
    (1.3
          0
                1.2)
                         // vertex number 5
                         // vertex number 6
    (1.4
           1.1
                1.3)
    ( 0
                         // vertex number 7
           1
                1.1)
);
```

### 5.3.1.2 The edges

Each edge joining 2 vertex points is assumed to be straight by default. However any edge may be specified to be curved by entries in a list named edges. The list is optional; if the geometry contains no curved edges, it may be omitted.

Each entry for a curved edge begins with a keyword specifying the type of curve from those listed in Table 5.6.

Keyword selection	Description	Additional entries
arc	Circular arc	Single interpolation point
simpleSpline	Spline curve	List of interpolation points
polyLine	Set of lines	List of interpolation points
polySpline	Set of splines	List of interpolation points
line	Straight line	

Table 5.6: Edge types available in the *blockMeshDict* dictionary.

The keyword is then followed by the labels of the 2 vertices that the edge connects. Following that, interpolation points must be specified through which the edge passes. For a arc, a single interpolation point is required, which the circular arc will intersect. For simpleSpline, polyLine and polySpline, a list of interpolation points is required. The line edge is directly equivalent to the option executed by default, and requires no interpolation points. Note that there is no need to use the line edge but it is included for completeness. For our example block in Figure 5.4 we specify an arc edge connecting vertices 1 and 5 as follows through the interpolation point (1.1, 0.0, 0.5):

edges (

```
arc 1 5 (1.1 0.0 0.5));
```

### 5.3.1.3 The blocks

The block definitions are contained in a list named **blocks**. Each block definition is a compound entry consisting of a list of vertex labels whose order is described in section 5.3, a vector giving the number of cells required in each direction, the type and list of cell expansion ratio in each direction.

Then the blocks are defined as follows:

The definition of each block is as follows:

- Vertex numbering The first entry is the shape identifier of the block, as defined in the .OpenFOAM-2.1.0/cellModels file. The shape is always hex since the blocks are always hexahedra. There follows a list of vertex numbers, ordered in the manner described on page U-136.
- Number of cells The second entry gives the number of cells in each of the  $x_1$   $x_2$  and  $x_3$  directions for that block.
- Cell expansion ratios The third entry gives the cell expansion ratios for each direction in the block. The expansion ratio enables the mesh to be graded, or refined, in specified directions. The ratio is that of the width of the end cell  $\delta_e$  along one edge of a block to the width of the start cell  $\delta_s$  along that edge, as shown in Figure 5.5. Each of the following keywords specify one of two types of grading specification available in blockMesh.
  - simpleGrading The simple description specifies uniform expansions in the local  $x_1$ ,  $x_2$  and  $x_3$  directions respectively with only 3 expansion ratios, *e.g.*

simpleGrading (1 2 3)

edgeGrading The full cell expansion description gives a ratio for each edge of the block, numbered according to the scheme shown in Figure 5.4 with the arrows representing the direction 'from first cell...to last cell' *e.g.* something like

edgeGrading (1 1 1 1 2 2 2 2 3 3 3 3)

This means the ratio of cell widths along edges 0-3 is 1, along edges 4-7 is 2 and along 8-11 is 3 and is directly equivalent to the simpleGrading example given above.

$$\delta_s \qquad \text{Expansion ratio} = \frac{\delta_e}{\delta_s} \qquad \delta_e$$
Expansion direction -

Figure 5.5: Mesh grading along a block edge

### 5.3.1.4 The boundary

The boundary of the mesh is given in a list named **boundary**. The boundary is broken into patches (regions), where each patch in the list has its name as the keyword, which is the choice of the user, although we recommend something that conveniently identifies the patch, *e.g.***inlet**; the name is used as an identifier for setting boundary conditions in the field data files. The patch information is then contained in sub-dictionary with:

- type: the patch type, either a generic patch on which some boundary conditions are applied or a particular geometric condition, as listed in Table 5.2 and described in section 5.2.2;
- faces: a list of block faces that make up the patch and whose name is the choice of the user, although we recommend something that conveniently identifies the patch, *e.g.*inlet; the name is used as an identifier for setting boundary conditions in the field data files.

blockMesh collects faces from any boundary patch that is omitted from the boundary list and assigns them to a default patch named defaultFaces of type empty. This means that for a 2 dimensional geometry, the user has the option to omit block faces lying in the 2D plane, knowing that they will be collected into an empty patch as required.

Returning to the example block in Figure 5.4, if it has an inlet on the left face, an output on the right face and the four other faces are walls then the patches could be defined as follows:

```
boundary
                         // keyword
(
    inlet
                         // patch name
    {
        type patch;
                         // patch type for patch 0
        faces
         (
             (0 4 7 3); // block face in this patch
        );
    }
                         // end of Oth patch definition
    outlet
                         // patch name
    ł
        type patch;
                         // patch type for patch 1
        faces
         (
             (1 \ 2 \ 6 \ 5)
        );
    }
    walls
```

Each block face is defined by a list of 4 vertex numbers. The order in which the vertices are given **must** be such that, looking from inside the block and starting with any vertex, the face must be traversed in a clockwise direction to define the other vertices.

When specifying a cyclic patch in blockMesh, the user must specify the name of the related cyclic patch through the neighbourPatch keyword. For example, a pair of cyclic patches might be specified as follows:

```
left
{
                     cyclic;
    type
    neighbourPatch right;
                     ((0 4 7 3));
    faces
}
right
ł
                     cyclic;
    type
    neighbourPatch left;
                     ((1 5 6 2));
    faces
}
```

### 5.3.2 Multiple blocks

A mesh can be created using more than 1 block. In such circumstances, the mesh is created as has been described in the preceeding text; the only additional issue is the connection between blocks, in which there are two distinct possibilities:

- face matching the set of faces that comprise a patch from one block are formed from the same set of vertices as a set of faces patch that comprise a patch from another block;
- **face merging** a group of faces from a patch from one block are connected to another group of faces from a patch from another block, to create a new set of internal faces connecting the two blocks.

To connect two blocks with **face matching**, the two patches that form the connection should simply be ignored from the **patches** list. **blockMesh** then identifies that the faces do not form an external boundary and combines each collocated pair into a single internal faces that connects cells from the two blocks.

The alternative, **face merging**, requires that the block patches to be merged are first defined in the **patches** list. Each pair of patches whose faces are to be merged must then

be included in an optional list named  $\tt mergePatchPairs$  . The format of  $\tt mergePatchPairs$  is:

```
mergePatchPairs
(
    ( <masterPatch> <slavePatch> ) // merge patch pair 0
    ( <masterPatch> <slavePatch> ) // merge patch pair 1
    ...
)
```

The pairs of patches are interpreted such that the first patch becomes the *master* and the second becomes the *slave*. The rules for merging are as follows:

- the faces of the master patch remain as originally defined, with all vertices in their original location;
- the faces of the slave patch are projected onto the master patch where there is some separation between slave and master patch;
- the location of any vertex of a slave face might be adjusted by **blockMesh** to eliminate any face edge that is shorter than a minimum tolerance;
- if patches overlap as shown in Figure 5.6, each face that does not merge remains as an external face of the original patch, on which boundary conditions must then be applied;
- if all the faces of a patch are merged, then the patch itself will contain no faces and is removed.



Figure 5.6: Merging overlapping patches

The consequence is that the original geometry of the slave patch will not necessarily be completely preserved during merging. Therefore in a case, say, where a cylindrical block is being connected to a larger block, it would be wise to the assign the master patch to the cylinder, so that its cylindrical shape is correctly preserved. There are some additional recommendations to ensure successful merge procedures:

- in 2 dimensional geometries, the size of the cells in the third dimension, *i.e.* out of the 2D plane, should be similar to the width/height of cells in the 2D plane;
- it is inadvisable to merge a patch twice, *i.e.* include it twice in mergePatchPairs;
- where a patch to be merged shares a common edge with another patch to be merged, both should be declared as a master patch.

### 5.3.3 Creating blocks with fewer than 8 vertices

It is possible to collapse one or more pair(s) of vertices onto each other in order to create a block with fewer than 8 vertices. The most common example of collapsing vertices is when creating a 6-sided wedge shaped block for 2-dimensional axi-symmetric cases that use the wedge patch type described in section 5.2.2. The process is best illustrated by using a simplified version of our example block shown in Figure 5.7. Let us say we wished to create a wedge shaped block by collapsing vertex 7 onto 4 and 6 onto 5. This is simply done by exchanging the vertex number 7 by 4 and 6 by 5 respectively so that the block numbering would become:

```
hex (0 1 2 3 4 5 5 4)
```



Figure 5.7: Creating a wedge shaped block with 6 vertices

The same applies to the patches with the main consideration that the block face containing the collapsed vertices, previously (4 5 6 7) now becomes (4 5 5 4). This is a block face of zero area which creates a patch with no faces in the **polyMesh**, as the user can see in a *boundary* file for such a case. The patch should be specified as **empty** in the *blockMeshDict* and the boundary condition for any fields should consequently be **empty** also.

### 5.3.4 Running blockMesh

As described in section 3.3, the following can be executed at the command line to run blockMesh for a case in the  $\langle case \rangle$  directory:

```
blockMesh -case <case>
```

The *blockMeshDict* file must exist in subdirectory *constant/polyMesh*.
## 5.4 Mesh generation with the snappyHexMesh utility

This section describes the mesh generation utility, snappyHexMesh, supplied with Open-FOAM. The snappyHexMesh utility generates 3-dimensional meshes containing hexahedra (hex) and split-hexahedra (split-hex) automatically from triangulated surface geometries in Stereolithography (STL) format. The mesh approximately conforms to the surface by iteratively refining a starting mesh and morphing the resulting split-hex mesh to the surface. An optional phase will shrink back the resulting mesh and insert cell layers. The specification of mesh refinement level is very flexible and the surface handling is robust with a pre-specified final mesh quality. It runs in parallel with a load balancing step every iteration.



Figure 5.8: Schematic 2D meshing problem for snappyHexMesh

#### 5.4.1 The mesh generation process of snappyHexMesh

The process of generating a mesh using snappyHexMesh will be described using the schematic in Figure 5.8. The objective is to mesh a rectangular shaped region (shaded grey in the figure) surrounding an object described by and STL surface, *e.g.* typical for an external aerodynamics simulation. Note that the schematic is 2-dimensional to make it easier to understand, even though the snappyHexMesh is a 3D meshing tool.

In order to run snappyHexMesh, the user requires the following:

- surface data files in STL format, either binary or ASCII, located in a *constant/triSurface* sub-directory of the case directory;
- a background hex mesh which defines the extent of the computational domain and a base level mesh density; typically generated using blockMesh, discussed in section 5.4.2.
- a *snappyHexMeshDict* dictionary, with appropriate entries, located in the *system* sub-directory of the case.

The *snappyHexMeshDict* dictionary includes: switches at the top level that control the various stages of the meshing process; and, individual sub-directories for each process. The entries are listed in Table 5.7.

All the geometry used by snappyHexMesh is specified in a *geometry* sub-dictionary in the *snappyHexMeshDict* dictionary. The geometry can be specified through an STL surface or bounding geometry entities in OpenFOAM. An example is given below:

Keyword	Description	Example		
castellatedMesh	Create the castellated mesh?	true		
snap	Do the surface snapping stage?	true		
doLayers	Add surface layers?	true		
mergeTolerance	Merge tolerance as fraction of bounding box of initial mesh	1e-06		
debug	Controls writing of intermediate meshes and			
	screen printing			
	— Write final mesh only	0		
	— Write intermediate meshes	1		
	— Write volScalarField with cellLevel for	2		
	post-processing			
	— Write current intersections as $.obj$ files	4		
geometry	Sub-dictionary of all surface geometry used			
castellatedMeshControls	Sub-dictionary of controls for castellated mes	h		
snapControls	Sub-dictionary of controls for surface snapping			
addLayersControls	Sub-dictionary of controls for layer addition			
meshQualityControls	Sub-dictionary of controls for mesh quality			

Table 5.7: Keywords at the top level of *snappyHexMeshDict*.

```
geometry
{
    sphere.stl // STL filename
        type triSurfaceMesh;
        regions
        ł
            secondSolid
                                     // Named region in the STL file
            {
                name mySecondPatch; // User-defined patch name
            }
                                      // otherwise given sphere.stl_secondSolid
        }
    }
    box1x1x1 // User defined region name
                                     // region defined by bounding box
               searchableBox;
        type
        min
                (1.5 1 - 0.5);
        max
                (3.5 \ 2 \ 0.5);
    }
    sphere2 // User defined region name
                                     // region defined by bounding sphere
               searchableSphere;
        type
        centre (1.5 1.5 1.5);
        radius 1.03;
    }
};
```

## 5.4.2 Creating the background hex mesh

Before snappyHexMesh is executed the user must create a background mesh of hexahedral cells that fills the entire region within by the external boundary as shown in Figure 5.9. This can be done simply using blockMesh. The following criteria must be observed when creating the background mesh:

- the mesh must consist purely of hexes;
- the cell aspect ratio should be approximately 1, at least near surfaces at which



Figure 5.9: Initial mesh generation in snappyHexMesh meshing process

the subsequent snapping procedure is applied, otherwise the convergence of the snapping procedure is slow, possibly to the point of failure;

• there must be at least one intersection of a cell edge with the STL surface, *i.e.* a mesh of one cell will not work.



Figure 5.10: Cell splitting by feature edge in snappyHexMesh meshing process

#### 5.4.3 Cell splitting at feature edges and surfaces

Cell splitting is performed according to the specification supplied by the user in the *castellatedMeshControls* sub-dictionary in the *snappyHexMeshDict*. The entries for *castellatedMeshControls* are presented in Table 5.8.

The splitting process begins with cells being selected according to specified edge features first within the domain as illustrated in Figure 5.10. The features list in the *castellatedMeshControls* sub-dictionary permits dictionary entries containing a name of an edgeMesh file and the level of refinement, *e.g.*:

```
features
(
        {
        file "features.eMesh"; // file containing edge mesh
            level 2; // level of refinement
     }
);
```

Keyword	Description	Example
locationInMesh	Location vector inside the region to be meshed	(5 0 0)
	N.B. vector must not coincide with a cell face	
	either before or during refinement	
maxLocalCells	Max number of cells per processor during re-	1e+06
	finement	
maxGlobalCells	Overall cell limit during refinement ( $i.e.$ before	2e+06
	removal)	
minRefinementCells	If $\geq$ number of cells to be refined, surface re-	0
	finement stops	
nCellsBetweenLevels	Number of buffer layers of cells between dif-	1
	ferent levels of refinement	
resolveFeatureAngle	Applies maximum level of refinement to cells	30
	that can see intersections whose angle exceeds	
	this	
features	List of features for refinement	
refinementSurfaces	Dictionary of surfaces for refinement	
refinementRegions	Dictionary of regions for refinement	

Table 5.8: Keywords in the *castellatedMeshControls* sub-dictionary of *snappyHexMeshDict*.

The edgeMesh containing the features can be extracted from the STL geometry file using surfaceFeatureExtract, e.g.

```
surfaceFeatureExtract -includedAngle 150 surface.stl features
```

Following feature refinement, cells are selected for splitting in the locality of specified surfaces as illustrated in Figure 5.11. The refinementSurfaces dictionary in *castel-latedMeshControls* requires dictionary entries for each STL surface and a default level specification of the minimum and maximum refinement in the form (<min> <max>). The minimum level is applied generally across the surface; the maximum level is applied to cells that can see intersections that form an angle in excess of that specified by resolveFeatureAngle.

The refinement can optionally be overridden on one or more specific region of an STL surface. The region entries are collected in a **regions** sub-dictionary. The keyword for



Figure 5.11: Cell splitting by surface in snappyHexMesh meshing process

each region entry is the name of the region itself and the refinement level is contained within a further sub-dictionary. An example is given below:

```
refinementSurfaces
{
    sphere.stl
    {
        level (2 2); // default (min max) refinement for whole surface
        regions
        {
            secondSolid
            {
                level (3 3); // optional refinement for secondSolid region
        }
    }
}
```

#### 5.4.4 Cell removal

Once the feature and surface splitting is complete a process of cell removal begins. Cell removal requires one or more regions enclosed entirely by a bounding surface within the domain. The region in which cells are retained are simply identified by a location vector within that region, specified by the locationInMesh keyword in *castellatedMeshControls*. Cells are retained if, approximately speaking, 50% or more of their volume lies within the region. The remaining cells are removed accordingly as illustrated in Figure 5.12.



Figure 5.12: Cell removal in snappyHexMesh meshing process

## 5.4.5 Cell splitting in specified regions

Those cells that lie within one or more specified volume regions can be further split as illustrated in Figure 5.13 by a rectangular region shown by dark shading. The refinement-Regions sub-dictionary in *castellatedMeshControls* contains entries for refinement of the volume regions specified in the *geometry* sub-dictionary. A refinement mode is applied to each region which can be:

- inside refines inside the volume region;
- outside refines outside the volume region
- distance refines according to distance to the surface; and can accommodate different levels at multiple distances with the levels keyword.

For the refinementRegions, the refinement level is specified by the levels list of entries with the format(<distance> <level>). In the case of inside and outside refinement, the <distance> is not required so is ignored (but it must be specified). Examples are shown below:

#### refinementRegions

```
box1x1x1
{
    mode inside;
    levels ((1.0 4)); // refinement level 4 (1.0 entry ignored)
}
sphere.stl
{
    // refinement level 5 within 1.0 m
    mode distance; // refinement level 3 within 2.0 m
    levels ((1.0 5) (2.0 3)); // levels must be ordered nearest first
}
```

#### 5.4.6 Snapping to surfaces

The next stage of the meshing process involves moving cell vertex points onto surface geometry to remove the jagged castellated surface from the mesh. The process is:

- 1. displace the vertices in the castellated boundary onto the STL surface;
- 2. solve for relaxation of the internal mesh with the latest displaced boundary vertices;
- 3. find the vertices that cause mesh quality parameters to be violated;
- 4. reduce the displacement of those vertices from their initial value (at 1) and repeat from 2 until mesh quality is satisfied.

The method uses the settings in the snapControls sub-dictionary in snappyHexMeshDict, listed in Table 5.9. An example is illustrated in the schematic in Figure 5.14 (albeit with

Keyword	Description	Example
nSmoothPatch	Number of patch smoothing iterations before	3
	finding correspondence to surface	
tolerance	Ratio of distance for points to be attracted	4.0
	by surface feature point or edge, to local	
	maximum edge length	
nSolveIter	Number of mesh displacement relaxation it-	30
	erations	
nRelaxIter	Maximum number of snapping relaxation it-	5
	erations	

Table 5.9: Keywords in the *snapControls* dictionary of *snappyHexMeshDict*.

mesh motion that looks slightly unrealistic).

## 5.4.7 Mesh layers

The mesh output from the snapping stage may be suitable for the purpose, although it can produce some irregular cells along boundary surfaces. There is an optional stage of the meshing process which introduces additional layers of hexahedral cells aligned to the boundary surface as illustrated by the dark shaded cells in Figure 5.15.



Figure 5.13: Cell splitting by region in snappyHexMesh meshing process



Figure 5.14: Surface snapping in snappyHexMesh meshing process



Figure 5.15: Layer addition in snappyHexMesh meshing process

The process of mesh layer addition involves shrinking the existing mesh from the boundary and inserting layers of cells, broadly as follows:

- 1. the mesh is projected back from the surface by a specified thickness in the direction normal to the surface;
- 2. solve for relaxation of the internal mesh with the latest projected boundary vertices;
- 3. check if validation criteria are satisfied otherwise reduce the projected thickness and return to 2; if validation cannot be satisfied for any thickness, do not insert layers;
- 4. if the validation criteria can be satisfied, insert mesh layers;
- 5. the mesh is checked again; if the checks fail, layers are removed and we return to 2.

The layer addition procedure uses the settings in the *addLayersControls* sub-dictionary in *snappyHexMeshDict*; entries are listed in Table 5.10. The layers sub-dictionary con-

Keyword	Description	Example
layers	Dictionary of layers	
relativeSizes	Are layer thicknesses relative to undistorted cell	true/false
	size outside layer or absolute?	
expansionRatio	Expansion factor for layer mesh	1.0
finalLayerRatio	Thickness of layer furthest from the wall, ei-	0.3
	ther relative or absolute according to the	
	relativeSizes entry	
minThickness	Minimum thickness of cell layer, either relative	0.25
	or absolute (as above)	
nGrow	Number of layers of connected faces that are not	1
	grown if points get not extruded; helps conver-	
	gence of layer addition close to features	
featureAngle	Angle above which surface is not extruded	60
nRelaxIter	Maximum number of snapping relaxation itera-	5
	tions	
nSmoothSurfaceNormals	Number of smoothing iterations of surface nor-	1
	mals	
nSmoothNormals	Number of smoothing iterations of interior mesh	3
	movement direction	
nSmoothThickness	Smooth layer thickness over surface patches	10
	Stop layer growth on highly warped cells	0.5
maxThicknessTo-	Reduce layer growth where ratio thickness to me-	0.3
MedialRatio	dial distance is large	
minMedianAxisAngle	Angle used to pick up medial axis points	130
	Create buffer region for new layer terminations	0
nLayerIter	Overall max number of layer addition iterations	50
nRelaxedIter	Max number of iterations after which the	20
	controls in the <i>relaxed</i> sub dictionary of	
	meshQuality are used	

Table 5.10: Keywords in the *addLayersControls* sub-dictionary of *snappyHexMeshDict*.

tains entries for each *patch* on which the layers are to be applied and the number of

surface layers required. The patch name is used because the layers addition relates to the existing mesh, not the surface geometry; hence applied to a patch, not a surface region. An example layers entry is as follows:

#### layers

```
{
    sphere.stl_firstSolid
    {
        nSurfaceLayers 1;
    }
    maxY
    {
        nSurfaceLayers 1;
    }
}
```

Keyword	Description	Example
maxNonOrtho	Maximum non-orthogonality allowed; 180 disables	65
maxBoundarySkewness	Max boundary face skewness allowed; <0 disables	20
maxInternalSkewness	Max internal face skewness allowed; $<0$ disables	4
maxConcave	Max concaveness allowed; 180 disables	80
minFlatness	Ratio of minimum projected area to actual area; -1 disables	0.5
minVol	Minimum pyramid volume; large negative number, $e.g1e30$ disables	1e-13
minArea	Minimum face area; $<0$ disables	-1
minTwist	Minimum face twist; $<-1$ disables	0.05
minDeterminant	Minimum normalised cell determinant; $1 = hex$ ; $\leq 0$ illegal cell	0.001
minFaceWeight	0→0.5	0.05
minVolRatio	0→1.0	0.01
minTriangleTwist	>0 for Fluent compatability	-1
nSmoothScale	Number of error distribution iterations	4
errorReduction	Amount to scale back displacement at error points	0.75
relaxed	Sub-dictionary that can include modified values	relaxed
	for the above keyword entries to be used when	{
	<b>nRelaxedIter</b> is exceeded in the layer addition	• • •
	process	}

Table 5.11: Keywords in the *meshQualityControls* sub-dictionary of *snappyHexMeshDict*.

## 5.4.8 Mesh quality controls

The mesh quality is controlled by the entries in the *meshQualityControls* sub-dictionary in *snappyHexMeshDict*; entries are listed in Table 5.11.

## 5.5 Mesh conversion

The user can generate meshes using other packages and convert them into the format that OpenFOAM uses. There are numerous mesh conversion utilities listed in Table 3.6.

Some of the more popular mesh converters are listed below and their use is presented in this section.

fluentMeshToFoam reads a Fluent.msh mesh file, working for both 2-D and 3-D cases;

starToFoam reads STAR-CD/PROSTAR mesh files.

gambitToFoam reads a GAMBIT.neu neutral file;

ideasToFoam reads an I-DEAS mesh written in ANSYS.ans format;

cfx4ToFoam reads a CFX mesh written in .geo format;

#### 5.5.1 fluentMeshToFoam

Fluent writes mesh data to a single file with a *.msh* extension. The file must be written in ASCII format, which is not the default option in Fluent. It is possible to convert single-stream Fluent meshes, including the 2 dimensional geometries. In OpenFOAM, 2 dimensional geometries are currently treated by defining a mesh in 3 dimensions, where the front and back plane are defined as the empty boundary patch type. When reading a 2 dimensional Fluent mesh, the converter automatically extrudes the mesh in the third direction and adds the empty patch, naming it frontAndBackPlanes.

The following features should also be observed.

- The OpenFOAM converter will attempt to capture the Fluent boundary condition definition as much as possible; however, since there is no clear, direct correspondence between the OpenFOAM and Fluent boundary conditions, the user should check the boundary conditions before running a case.
- Creation of axi-symmetric meshes from a 2 dimensional mesh is currently not supported but can be implemented on request.
- Multiple material meshes are not permitted. If multiple fluid materials exist, they will be converted into a single OpenFOAM mesh; if a solid region is detected, the converter will attempt to filter it out.
- Fluent allows the user to define a patch which is internal to the mesh, *i.e.* consists of the faces with cells on both sides. Such patches are not allowed in OpenFOAM and the converter will attempt to filter them out.
- There is currently no support for embedded interfaces and refinement trees.

The procedure of converting a Fluent.*msh* file is first to create a new OpenFOAM case by creating the necessary directories/files: the case directory containing a *controlDict* file in a *system* subdirectory. Then at a command prompt the user should execute:

```
fluentMeshToFoam <meshFile>
```

where <meshFile> is the name of the .msh file, including the full or relative path.

## 5.5.2 starToFoam

This section describes how to convert a mesh generated on the STAR-CD code into a form that can be read by OpenFOAM mesh classes. The mesh can be generated by any of the packages supplied with STAR-CD, *i.e.*PROSTAR, SAMM, ProAM and their derivatives. The converter accepts any single-stream mesh including integral and arbitrary couple matching and all cell types are supported. The features that the converter does not support are:

- multi-stream mesh specification;
- baffles, *i.e.* zero-thickness walls inserted into the domain;
- partial boundaries, where an uncovered part of a couple match is considered to be a boundary face;
- sliding interfaces.

For multi-stream meshes, mesh conversion can be achieved by writing each individual stream as a separate mesh and reassemble them in OpenFOAM.

OpenFOAM adopts a policy of only accepting input meshes that conform to the fairly stringent validity criteria specified in section 5.1. It will simply not run using invalid meshes and cannot convert a mesh that is itself invalid. The following sections describe steps that must be taken when generating a mesh using a mesh generating package supplied with STAR-CD to ensure that it can be converted to OpenFOAM format. To avoid repetition in the remainder of the section, the mesh generation tools supplied with STAR-CD will be referred to by the collective name STAR-CD.

## 5.5.2.1 General advice on conversion

We strongly recommend that the user run the STAR-CD mesh checking tools before attempting a starToFoam conversion and, after conversion, the checkMesh utility should be run on the newly converted mesh. Alternatively, starToFoam may itself issue warnings containing PROSTAR commands that will enable the user to take a closer look at cells with problems. Problematic cells and matches should be checked and fixed before attempting to use the mesh with OpenFOAM. Remember that an invalid mesh will not run with OpenFOAM, but it may run in another environment that does not impose the validity criteria.

Some problems of tolerance matching can be overcome by the use of a matching tolerance in the converter. However, there is a limit to its effectiveness and an apparent need to increase the matching tolerance from its default level indicates that the original mesh suffers from inaccuracies.

## 5.5.2.2 Eliminating extraneous data

When mesh generation in is completed, remove any extraneous vertices and compress the cells boundary and vertex numbering, assuming that fluid cells have been created and all other cells are discarded. This is done with the following PROSTAR commands:

CSET NEWS FLUID CSET INVE The CSET should be empty. If this is not the case, examine the cells in CSET and adjust the model. If the cells are genuinely not desired, they can be removed using the PROSTAR command:

CDEL CSET

Similarly, vertices will need to be discarded as well:

CSET NEWS FLUID VSET NEWS CSET VSET INVE

Before discarding these unwanted vertices, the unwanted boundary faces have to be collected before purging:

CSET NEWS FLUID VSET NEWS CSET BSET NEWS VSET ALL BSET INVE

If the BSET is not empty, the unwanted boundary faces can be deleted using:

BDEL BSET

At this time, the model should contain only the fluid cells and the supporting vertices, as well as the defined boundary faces. All boundary faces should be fully supported by the vertices of the cells, if this is not the case, carry on cleaning the geometry until everything is clean.

#### 5.5.2.3 Removing default boundary conditions

By default, STAR-CD assigns wall boundaries to any boundary faces not explicitly associated with a boundary region. The remaining boundary faces are collected into a default boundary region, with the assigned boundary type 0. OpenFOAM deliberately does not have a concept of a default boundary condition for undefined boundary faces since it invites human error, *e.g.* there is no means of checking that we meant to give all the unassociated faces the default condition.

Therefore **all** boundaries for each OpenFOAM mesh must be specified for a mesh to be successfully converted. The **default** boundary needs to be transformed into a real one using the procedure described below:

- 1. Plot the geometry with Wire Surface option.
- 2. Define an extra boundary region with the same parameters as the default region 0 and add all visible faces into the new region, say 10, by selecting a zone option in the boundary tool and drawing a polygon around the entire screen draw of the model. This can be done by issuing the following commands in PROSTAR:

RDEF 10 WALL BZON 10 ALL 3. We shall remove all previously defined boundary types from the set. Go through the boundary regions:

BSET NEWS REGI 1 BSET NEWS REGI 2 ... 3, 4, ...

Collect the vertices associated with the boundary set and then the boundary faces associated with the vertices (there will be twice as many of them as in the original set).

BSET NEWS REGI 1 VSET NEWS BSET BSET NEWS VSET ALL BSET DELE REGI 1 REPL

This should give the faces of boundary Region 10 which have been defined on top of boundary Region 1. Delete them with BDEL BSET. Repeat these for all regions.

#### 5.5.2.4 Renumbering the model

Renumber and check the model using the commands:

```
CSET NEW FLUID
CCOM CSET
VSET NEWS CSET
VSET INVE (Should be empty!)
VSET INVE
VCOM VSET
BSET NEWS VSET ALL
BSET INVE (Should be empty also!)
BSET INVE
BCOM BSET
CHECK ALL
GEOM
```

Internal PROSTAR checking is performed by the last two commands, which may reveal some other unforeseeable error(s). Also, take note of the scaling factor because PROSTAR only applies the factor for STAR-CD and not the geometry. If the factor is not 1, use the scalePoints utility in OpenFOAM.

#### 5.5.2.5 Writing out the mesh data

Once the mesh is completed, place all the integral matches of the model into the couple type 1. All other types will be used to indicate arbitrary matches.

CPSET NEWS TYPE INTEGRAL CPMOD CPSET 1 The components of the computational grid must then be written to their own files. This is done using PROSTAR for boundaries by issuing the command

#### BWRITE

by default, this writes to a .23 file (versions prior to 3.0) or a .bnd file (versions 3.0 and higher). For cells, the command

#### CWRITE

outputs the cells to a .14 or .cel file and for vertices, the command

#### VWRITE

outputs to file a .15 or .vrt file. The current default setting writes the files in ASCII format. If couples are present, an additional couple file with the extension .cpl needs to be written out by typing:

#### CPWRITE

After outputting to the three files, exit PROSTAR or close the files. Look through the panels and take note of all STAR-CD sub-models, material and fluid properties used – the material properties and mathematical model will need to be set up by creating and editing OpenFOAM dictionary files.

The procedure of converting the PROSTAR files is first to create a new OpenFOAM case by creating the necessary directories. The PROSTAR files must be stored within the same directory and the user must change the file extensions: from .23, .14 and .15 (below STAR-CD version 3.0), or .pcs, .cls and .vtx (STAR-CD version 3.0 and above); to .bnd, .cel and .vrt respectively.

#### 5.5.2.6 Problems with the .vrt file

The *.vrt* file is written in columns of data of specified width, rather than free format. A typical line of data might be as follows, giving a vertex number followed by the coordinates:

19422 -0.105988957 -0.413711881E-02 0.00000000E+00

If the ordinates are written in scientific notation and are negative, there may be no space between values, e.g.:

19423 -0.953953117E-01-0.338810333E-02 0.00000000E+00

The starToFoam converter reads the data using spaces to delimit the ordinate values and will therefore object when reading the previous example. Therefore, OpenFOAM includes a simple script, foamCorrectVrt to insert a space between values where necessary, *i.e.* it would convert the previous example to:

19423 -0.953953117E-01 -0.338810333E-02 0.00000000E+00

The foamCorrectVrt script should therefore be executed if necessary before running the starToFoam converter, by typing:

```
foamCorrectVrt <file>.vrt
```

#### 5.5.2.7 Converting the mesh to OpenFOAM format

The translator utility **starToFoam** can now be run to create the boundaries, cells and points files necessary for a OpenFOAM run:

```
starToFoam <meshFilePrefix>
```

where <meshFilePrefix> is the name of the the prefix of the mesh files, including the full or relative path. After the utility has finished running, OpenFOAM boundary types should be specified by editing the *boundary* file by hand.

## 5.5.3 gambitToFoam

GAMBIT writes mesh data to a single file with a *.neu* extension. The procedure of converting a GAMBIT.*neu* file is first to create a new OpenFOAM case, then at a command prompt, the user should execute:

```
gambitToFoam <meshFile>
```

where <meshFile> is the name of the .neu file, including the full or relative path.

The GAMBIT file format does not provide information about type of the boundary patch, *e.g.* wall, symmetry plane, cyclic. Therefore all the patches have been created as type patch. Please reset after mesh conversion as necessary.

#### 5.5.4 ideasToFoam

OpenFOAM can convert a mesh generated by I-DEAS but written out in ANSYS format as a *.ans* file. The procedure of converting the *.ans* file is first to create a new OpenFOAM case, then at a command prompt, the user should execute:

```
ideasToFoam <meshFile>
```

where <meshFile> is the name of the .ans file, including the full or relative path.

## 5.5.5 cfx4ToFoam

CFX writes mesh data to a single file with a .geo extension. The mesh format in CFX is block-structured, *i.e.* the mesh is specified as a set of blocks with glueing information and the vertex locations. OpenFOAM will convert the mesh and capture the CFX boundary condition as best as possible. The 3 dimensional 'patch' definition in CFX, containing information about the porous, solid regions *etc.* is ignored with all regions being converted into a single OpenFOAM mesh. CFX supports the concept of a 'default' patch, where each external face without a defined boundary condition is treated as a wall. These faces are collected by the converter and put into a defaultFaces patch in the OpenFOAM mesh and given the type wall; of course, the patch type can be subsequently changed.

Like, OpenFOAM 2 dimensional geometries in CFX are created as 3 dimensional meshes of 1 cell thickness. If a user wishes to run a 2 dimensional case on a mesh created by CFX, the boundary condition on the front and back planes should be set to empty; the user should ensure that the boundary conditions on all other faces in the plane of the calculation are set correctly. Currently there is no facility for creating an axi-symmetric geometry from a 2 dimensional CFX mesh.

The procedure of converting a CFX.*geo* file is first to create a new OpenFOAM case, then at a command prompt, the user should execute:

```
cfx4ToFoam <meshFile>
```

where <meshFile> is the name of the .geo file, including the full or relative path.

## 5.6 Mapping fields between different geometries

The mapFields utility maps one or more fields relating to a given geometry onto the corresponding fields for another geometry. It is completely generalised in so much as there does not need to be any similarity between the geometries to which the fields relate. However, for cases where the geometries are consistent, mapFields can be executed with a special option that simplifies the mapping process.

For our discussion of mapFields we need to define a few terms. First, we say that the data is mapped from the *source* to the *target*. The fields are deemed *consistent* if the geometry *and* boundary types, or conditions, of both source and target fields are identical. The field data that mapFields maps are those fields within the time directory specified by startFrom/startTime in the *controlDict* of the target case. The data is read from the equivalent time directory of the source case and mapped onto the equivalent time directory of the target case.

## 5.6.1 Mapping consistent fields

A mapping of consistent fields is simply performed by executing **mapFields** on the (target) case using the **-consistent** command line option as follows:

```
mapFields <source dir> -consistent
```

## 5.6.2 Mapping inconsistent fields

When the fields are not consistent, as shown in Figure 5.16, mapFields requires a *map-FieldsDict* dictionary in the *system* directory of the target case. The following rules apply to the mapping:

- the field data is mapped from source to target wherever possible, *i.e.* in our example all the field data within the target geometry is mapped from the source, except those in the shaded region which remain unaltered;
- the patch field data is left unaltered unless specified otherwise in the *mapFieldsDict* dictionary.

The *mapFieldsDict* dictionary contain two lists that specify mapping of patch data. The first list is patchMap that specifies mapping of data between pairs of source and target patches that are geometrically coincident, as shown in Figure 5.16. The list contains each pair of names of source and target patch. The second list is cuttingPatches that contains names of target patches whose values are to be mapped from the source internal field through which the target patch cuts. In the situation where the target patch only cuts through part of the source internal field, *e.g.* bottom left target patch in our example, those values within the internal field are mapped and those outside remain unchanged. An example *mapFieldsDict* dictionary is shown below:



```
mapFields <source dir>
```

## 5.6.3 Mapping parallel cases

If either or both of the source and target cases are decomposed for running in parallel, additional options must be supplied when executing mapFields:

-parallelSource if the source case is decomposed for parallel running;

-parallelTarget if the target case is decomposed for parallel running.

# Chapter 6

## Post-processing

This chapter describes options for post-processing with OpenFOAM. OpenFOAM is supplied with a post-processing utility paraFoam that uses ParaView, an open source visualisation application described in section 6.1.

Other methods of post-processing using third party products are offered, including EnSight, Fieldview and the post-processing supplied with Fluent.

#### 6.1 paraFoam

The main post-processing tool provided with OpenFOAM is a reader module to run with ParaView, an open-source, visualization application. The module is compiled into 2 libraries, PV3FoamReader and vtkPV3Foam using version 3.10.1 of ParaView supplied with the OpenFOAM release (PVFoamReader and vtkFoam in ParaView version 2.x). It is recommended that this version of ParaView is used, although it is possible that the latest binary release of the software will run adequately. Further details about ParaView can be found at http://www.paraview.org and further documentation is available at http://www.kitware.com/products/paraviewguide.html.

ParaView uses the Visualisation Toolkit (VTK) as its data processing and rendering engine and can therefore read any data in VTK format. OpenFOAM includes the foam-ToVTK utility to convert data from its native format to VTK format, which means that any VTK-based graphics tools can be used to post-process OpenFOAM cases. This provides an alternative means for using ParaView with OpenFOAM. For users who wish to experiment with advanced, parallel visualisation, there is also the free Vislt software, available at http://www.llnl.gov/visit.

In summary, we recommend the reader module for ParaView as the primary postprocessing tool for OpenFOAM. Alternatively OpenFOAM data can be converted into VTK format to be read by ParaView or any other VTK -based graphics tools.

#### 6.1.1 Overview of paraFoam

paraFoam is strictly a script that launches ParaView using the reader module supplied with OpenFOAM. It is executed like any of the OpenFOAM utilities either by the single command from within the case directory or with the -case option with the case path as an argument, *e.g.*:

#### paraFoam -case <caseDir>

ParaView is launched and opens the window shown in Figure 6.1. The case is controlled from the left panel, which contains the following:



Figure 6.1: The paraFoam window

- Pipeline Browser lists the *modules* opened in ParaView, where the selected modules are highlighted in blue and the graphics for the given module can be enabled/disabled by clicking the eye button alongside;
- Properties panel contains the input selections for the case, such as times, regions and fields;

Display panel controls the visual representation of the selected module, e.g. colours;

Information panel gives case statistics such as mesh geometry and size.

ParaView operates a tree-based structure in which data can be filtered from the toplevel case module to create sets of sub-modules. For example, a contour plot of, say, pressure could be a sub-module of the case module which contains all the pressure data. The strength of ParaView is that the user can create a number of sub-modules and display whichever ones they feel to create the desired image or animation. For example, they may add some solid geometry, mesh and velocity vectors, to a contour plot of pressure, switching any of the items on and off as necessary.

The general operation of the system is based on the user making a selection and then clicking the green Apply button in the Properties panel. The additional buttons are: the Reset button which can be used to reset the GUI if necessary; and, the Delete button that will delete the active module.

#### 6.1.2 The Properties panel

The Properties panel for the case module contains the settings for time step, regions and fields. The controls are described in Figure 6.2. It is particularly worth noting that



Figure 6.2: The Properties panel for the case module

in the current reader module, data in all time directories are loaded into ParaView (in the reader module for ParaView 2.x, a set of check boxes controlled the time that were displayed). In the current reader module, the buttons in the Current Time Controls and VCR Controls toolbars select the time data to be displayed, as shown is section 6.1.4.

As with any operation in paraFoam, the user must click Apply after making any changes to any selections. The Apply button is highlighted in green to alert the user if changes have been made but not accepted. This method of operation has the advantage of allowing the user to make a number of selections before accepting them, which is particularly useful in large cases where data processing is best kept to a minimum.

There are occasions when the case data changes on file and ParaView needs to load the changes, *e.g.* when field data is written into new time directories. To load the changes, the user should check the Update GUI button at the top of the Properties panel and then apply the changes.

#### 6.1.3 The Display panel

The **Display** panel contains the settings for visualising the data for a given case module. The following points are particularly important:

- the data range may not be automatically updated to the max/min limits of a field, so the user should take care to select **Rescale to Data Range** at appropriate intervals, in particular after loading the initial case module;
- clicking the Edit Color Map button, brings up a window in which there are two panels:

		Properties	Display	Info	ormation		
View case data	_	<b>X</b> ∨isible			<u>Z</u>	oom To Data	a
		_Color					]
		🕱 Interpolat	te Colors				
		🕱 Map Sca	alars				
		Apply Textu	re 🛛	one			-
Colour geometry/entity by	-	Color by	0	p		•	•
Set colour map range/appearance	-	E	Edit Color N	Лар.	Resca	ale to Data F	Range
		_Slice					
		Slice Direct	tion				-
		Slice	0			0	
		_Style					
Outline, surface, wireframe or points	_	- Representa	tion		Surface		-
Data interpolation method	_	Interpolation	ı		Gourau	d	-
Ĩ		Material			None		•
		Point size			5.00		<b>•</b>
Change image opacity		Line width			1.00		•
	_	Opacity			1.00		<b>^</b>
e.g. to make transluscent		Volume map	oper				-
		Edge Style-					
				at Ec	ige Color.		
					ige coloi.		
		-Annotation-					
		Show cul	be axes			Edit	
		Lighting—					]
		Specular Inf	tensity		0.10		<b></b>
		Specular Fo	ocus		100		*
		🕱 Specular	White				
		Transformat	tion——				
	П	Translate	0.00	-	0.00	• 0.00	<b>•</b>
		Scale	1.00	•	1.00	<b>▲</b> 1.00	<b></b>
Geometry manipulation tools		Orientation	0.00	•	0.00	• 0.00	<b>•</b>
	Ц	_ Origin	0.00	•	0.00	• 0.00	<b>•</b>

Figure 6.3: The Display panel

- 1. The Color Scale panel in which the colours within the scale can be chosen. The standard blue to red colour scale for CFD can be selected by clicking Choose Preset and selecting Blue to Red Rainbox HSV.
- 2. The Color Legend panel has a toggle switch for a colour bar legend and contains settings for the layout of the legend, e.g. font.
- the underlying mesh can be represented by selecting Wireframe in the Representation menu of the Style panel;
- the geometry, *e.g.* a mesh (if Wireframe is selected), can be visualised as a single colour by selecting Solid Color from the Color By menu and specifying the colour in the Set Ambient Color window;

• the image can be made translucent by editing the value in the Opacity text box (1 = solid, 0 = invisible) in the Style panel.

## 6.1.4 The button toolbars

ParaView duplicates functionality from pull-down menus at the top of the main window and the major panels, within the toolbars below the main pull-down menus. The displayed toolbars can be selected from Toolbars in the main View menu. The default layout with all toolbars is shown in Figure 6.4 with each toolbar labelled. The function of many of the buttons is clear from their icon and, with tooltips enabled in the Help menu, the user is given a concise description of the function of any button.



Figure 6.4: Toolbars in ParaView

## 6.1.5 Manipulating the view

This section describes operations for setting and manipulating the view of objects in  $\mathsf{paraFoam}.$ 

#### 6.1.5.1 View settings

The View Settings are selected from the Edit menu, which opens a View Settings (Render View) window with a table of 3 items: General, Lights and Annotation. The General panel includes the following items which are often worth setting at startup:

- the background colour, where white is often a preferred choice for printed material, is set by choosing background from the down-arrow button next to Choose Color button, then selecting the color by clicking on the Choose Color button;
- Use parallel projection which is the usual choice for CFD, especially for 2D cases.

The Lights panel contains detailed lighting controls within the Light Kit panel. A separate Headlight panel controls the direct lighting of the image. Checking the Headlight button with white light colour of strength 1 seems to help produce images with strong bright colours, e.g. with an isosurface.

The Annotation panel includes options for including annotations in the image. The Orientation Axes feature controls an axes icon in the image window, e.g. to set the colour of the axes labels x, y and z.

#### 6.1.5.2 General settings

The general Settings are selected from the Edit menu, which opens a general Options window with General, Colors, Animations, Charts and Render View menu items.

The General panel controls some default behaviour of ParaView. In particular, there is an Auto Accept button that enables ParaView to accept changes automatically without clicking the green Apply button in the Properties window. For larger cases, this option is generally not recommended: the user does not generally want the image to be re-rendered between each of a number of changes he/she selects, but be able to apply a number of changes to be re-rendered in their entirety once.

The Render View panel contains 3 sub-items: General, Camera and Server. The General panel includes the level of detail (LOD) which controls the rendering of the image while it is being manipulated, *e.g.* translated, resized, rotated; lowering the levels set by the sliders, allows cases with large numbers of cells to be re-rendered quickly during manipulation.

The Camera panel includes control settings for 3D and 2D movements. This presents the user with a map of rotation, translate and zoom controls using the mouse in combination with Shift- and Control-keys. The map can be edited to suit by the user.

## 6.1.6 Contour plots

A contour plot is created by selecting **Contour** from the **Filter** menu at the top menu bar. The filter acts on a given module so that, if the module is the 3D case module itself, the contours will be a set of 2D surfaces that represent a constant value, *i.e.* isosurfaces. The **Properties** panel for contours contains an **Isosurfaces** list that the user can edit, most conveniently by the **New Range** window. The chosen scalar field is selected from a pull down menu.

#### 6.1.6.1 Introducing a cutting plane

Very often a user will wish to create a contour plot across a plane rather than producing isosurfaces. To do so, the user must first use the Slice filter to create the cutting plane, on which the contours can be plotted. The Slice filter allows the user to specify a cutting Plane, Box or Sphere in the Slice Type menu by a center and normal/radius respectively. The user can manipulate the cutting plane like any other using the mouse.

The user can then run the Contour filter on the cut plane to generate contour lines.

## 6.1.7 Vector plots

Vector plots are created using the Glyph filter. The filter reads the field selected in Vectors and offers a range of Glyph Types for which the Arrow provides a clear vector plot images. Each glyph has a selection of graphical controls in a panel which the user can manipulate to best effect.

The remainder of the Properties panel contains mainly the Scale Mode menu for the glyphs. The most common options are Scale Mode are: Vector, where the glyph length is proportional to the vector magnitude; and, Off where each glyph is the same length. The Set Scale Factor parameter controls the base length of the glyphs.

#### 6.1.7.1 Plotting at cell centres

Vectors are by default plotted on cell vertices but, very often, we wish to plot data at cell centres. This is done by first applying the Cell Centers filter to the case module, and then applying the Glyph filter to the resulting cell centre data.

## 6.1.8 Streamlines

Streamlines are created by first creating tracer lines using the Stream Tracer filter. The tracer Seed panel specifies a distribution of tracer points over a Line Source or Point Cloud. The user can view the tracer source, *e.g.* the line, but it is displayed in white, so they may need to change the background colour in order to see it.

The distance the tracer travels and the length of steps the tracer takes are specified in the text boxes in the main **Stream Tracer** panel. The process of achieving desired tracer lines is largely one of trial and error in which the tracer lines obviously appear smoother as the step length is reduced but with the penalty of a longer calculation time.

Once the tracer lines have been created, the **Tubes** filter can be applied to the *Tracer* module to produce high quality images. The tubes follow each tracer line and are not strictly cylindrical but have a fixed number of sides and given radius. When the number of sides is set above, say, 10, the tubes do however appear cylindrical, but again this adds a computational cost.

#### 6.1.9 Image output

The simplest way to output an image to file from ParaView is to select Save Screenshot from the File menu. On selection, a window appears in which the user can select the resolution for the image to save. There is a button that, when clicked, locks the aspect ratio, so if the user changes the resolution in one direction, the resolution is adjusted in the other direction automatically. After selecting the pixel resolution, the image can be saved. To achieve high quality output, the user might try setting the pixel resolution to 1000 or more in the x-direction so that when the image is scaled to a typical size of a figure in an A4 or US letter document, perhaps in a PDF document, the resolution is sharp.

#### 6.1.10 Animation output

To create an animation, the user should first select **Save Animation** from the File menu. A dialogue window appears in which the user can specify a number of things including the image resolution. The user should specify the resolution as required. The other noteworthy setting is number of frames per timestep. While this would intuitively be set to 1, it can be set to a larger number in order to introduce more frames into the animation artificially. This technique can be particularly useful to produce a slower animation because some movie players have limited speed control, particularly over mpeg movies.

On clicking the Save Animation button, another window appears in which the user specifies a file name *root* and file format for a set of images. On clicking OK, the set of files will be saved according to the naming convention "<fileRoot>\_<imageNo>.<fileExt>", *e.g.* the third image of a series with the file root "animation", saved in jpg format would be named "animation\_0002.jpg" (<imageNo> starts at 0000).

Once the set of images are saved the user can convert them into a movie using their software of choice. The **convert** utility in the **ImageMagick** package can do this from the command line, *e.g.* by

```
convert animation*jpg movie.mpg
```

When creating an mpg movie it can be worth increasing the default quality setting, *e.g.* with -quality 90%, to reduce the graininess that can occur with the default setting.

## 6.2 Post-processing with Fluent

It is possible to use Fluent as a post-processor for the cases run in OpenFOAM. Two converters are supplied for the purpose: foamMeshToFluent which converts the OpenFOAM mesh into Fluent format and writes it out as a *.msh* file; and, foamDataToFluent converts the OpenFOAM results data into a *.dat* file readable by Fluent. foamMeshToFluent is executed in the usual manner. The resulting mesh is written out in a *fluentInterface* subdirectory of the case directory, *i.e.<caseName>/fluentInterface/<caseName>.msh* 

foamDataToFluent converts the OpenFOAM data results into the Fluent format. The conversion is controlled by two files. First, the *controlDict* dictionary specifies startTime, giving the set of results to be converted. If you want to convert the latest result, startFrom can be set to latestTime. The second file which specifies the translation is the *foamDataToFluentDict* dictionary, located in the *constant* directory. An example *foamDataToFluentDict* dictionary is given below:

---\*- C++ -\*-----1 ======== 2 OpenFOAM: The Open Source CFD Toolbox 3 F ield Version: 2.1.0 0 peration 4 www.OpenFOAM.org A nd Т Web: 5M anipulation \\/ 1 6 \_\_\_\_\_ 8 FoamFile 9 2.0; ascii; 10 version 11 format 12class dictionary; location "system" 13 object foamDataToFluentDict; 14 } 15 \* \* \* \* \* \* \* \* \* \* 16 17 1; 18 р 19 U 2; 2021 Т 3; 22 23 h 4; 24255; k 26 27epsilon 6; 28 29 150; gamma 30 31 32 33

The dictionary contains entries of the form

<fieldName> <fluentUnitNumber>

The <fluentUnitNumber> is a label used by the Fluent post-processor that only recognises a fixed set of fields. The basic set of <fluentUnitNumber> numbers are quoted in Table 6.1. The dictionary must contain all the entries the user requires to post-process, *e.g.* in our example we have entries for pressure p and velocity U. The list of default entries described in Table 6.1. The user can run foamDataToFluent like any utility.

To view the results using Fluent, go to the *fluentInterface* subdirectory of the case directory and start a 3 dimensional version of Fluent with

fluent 3d

The mesh and data files can be loaded in and the results visualised. The mesh is read by selecting Read Case from the File menu. Support items should be selected to read

Fluent name	Unit number	Common OpenFOAM name
PRESSURE	1	р
MOMENTUM	2	U
TEMPERATURE	3	Т
ENTHALPY	4	h
TKE	5	k
TED	6	epsilon
SPECIES	7	
G	8	
XF_RF_DATA_VOF	150	gamma
TOTAL_PRESSURE	192	
TOTAL_TEMPERATURE	193	_

Table 6.1: Fluent unit numbers for post-processing.

certain data types, *e.g.* to read turbulence data for k and epsilon, the user would select k-epsilon from the Define->Models->Viscous menu. The data can then be read by selecting Read Data from the File menu.

A note of caution: users MUST NOT try to use an original Fluent mesh file that has been converted to OpenFOAM format in conjunction with the OpenFOAM solution that has been converted to Fluent format since the alignment of zone numbering cannot be guaranteed.

## 6.3 Post-processing with Fieldview

OpenFOAM offers the capability for post-processing OpenFOAM cases with Fieldview. The method involves running a post-processing utility foamToFieldview to convert case data from OpenFOAM to Fieldview.uns file format. For a given case, foamToFieldview is executed like any normal application. foamToFieldview creates a directory named *Fieldview* in the case directory, deleting any existing Fieldview directory in the process. By default the converter reads the data in all time directories and writes into a set of files of the form < case > nn.uns, where nn is an incremental counter starting from 1 for the first time directory with the option -time < time>, where <time> is a time in general, scientific or fixed format.

Fieldview provides certain functions that require information about boundary conditions, *e.g.* drawing streamlines that uses information about wall boundaries. The converter tries, wherever possible, to include this information in the converted files by default. The user can disable the inclusion of this information by using the **-noWall** option in the execution command.

The data files for Fieldview have the *.uns* extension as mentioned already. If the original OpenFOAM case includes a dot '.', Fieldview may have problems interpreting a set of data files as a single case with multiple time steps.

## 6.4 Post-processing with EnSight

OpenFOAM offers the capability for post-processing OpenFOAM cases with  $\mathsf{EnSight},$  with a choice of 2 options:

- converting the OpenFOAM data to EnSight format with the foamToEnsight utility;
- reading the OpenFOAM data directly into EnSight using the ensight74FoamExec module.

## 6.4.1 Converting data to EnSight format

The foamToEnsight utility converts data from OpenFOAM to EnSight file format. For a given case, foamToEnsight is executed like any normal application. foamToEnsight creates a directory named *Ensight* in the case directory, *deleting any existing Ensight directory in the process*. The converter reads the data in all time directories and writes into a case file and a set of data files. The case file is named *EnSight\_Case* and contains details of the data file names. Each data file has a name of the form *EnSight\_nn.ext*, where *nn* is an incremental counter starting from 1 for the first time directory, 2 for the second and so on and *ext* is a file extension of the name of the field that the data refers to, as described in the case file, *e.g.*T for temperature, **mesh** for the mesh. Once converted, the data can be read into EnSight by the normal means:

- 1. from the EnSight GUI, the user should select Data (Reader) from the File menu;
- 2. the appropriate *EnSight\_Case* file should be highlighted in the Files box;
- 3. the Format selector should be set to Case, the EnSight default setting;
- 4. the user should click (Set) Case and Okay.

## 6.4.2 The ensight74FoamExec reader module

EnSight provides the capability of using a user-defined module to read data from a format other than the standard EnSight format. OpenFOAM includes its own reader module ensight74FoamExec that is compiled into a library named libuserd-foam. It is this library that EnSight needs to use which means that it must be able to locate it on the filing system as described in the following section.

#### 6.4.2.1 Configuration of EnSight for the reader module

In order to run the EnSight reader, it is necessary to set some environment variables correctly. The settings are made in the *bashrc* (or *cshrc*) file in the *\$WM\_PROJECT\_DIR/etc/-apps/ensightFoam* directory. The environment variables associated with EnSight are pre-fixed by \$CEL or \$ENSIGHT7\_ and listed in Table 6.2. With a standard user setup, only \$CELHOME may need to be set manually, to the path of the EnSight installation.

#### 6.4.2.2 Using the reader module

The principal difficulty in using the EnSight reader lies in the fact that EnSight expects that a case to be defined by the contents of a particular file, rather than a directory as it is in OpenFOAM. Therefore in following the instructions for the using the reader below, the user should pay particular attention to the details of case selection, since EnSight does not permit selection of a directory name.

- 1. from the EnSight GUI, the user should select Data (Reader) from the File menu;
- 2. The user should now be able to select the OpenFOAM from the Format menu; if not, there is a problem with the configuration described above.

U-	$\cdot 17$	'3

Environment variable	Description and options		
\$CEI_HOME	Path where EnSight is installed, eg /usr/local/ensight, added		
	to the system path by default		
\$CEI_ARCH	Machine architecture, from a choice of names cor-		
	responding to the machine directory names in		
	<i>\$CEI_HOME/ensight74/machines</i> ; default settings include		
	linux_2.4 and sgi_6.5_n32		
\$ENSIGHT7_READER	Path that EnSight searches for the user defined libuserd-foam		
	reader library, set by default to <b>\$FOAM_LIBBIN</b>		
\$ENSIGHT7_INPUT	Set by default to dummy		

Table 6.2: Environment variable settings for EnSight.

- 3. The user should find their case directory from the File Selection window, highlight one of top 2 entries in the Directories box ending in /. or /.. and click (Set) Geometry.
- 4. The path field should now contain an entry for the case. The (Set) Geometry text box should contain a '/'.
- 5. The user may now click Okay and EnSight will begin reading the data.
- 6. When the data is read, a new Data Part Loader window will appear, asking which part(s) are to be read. The user should select Load all.
- 7. When the mesh is displayed in the EnSight window the user should close the Data Part Loader window, since some features of EnSight will not work with this window open.

## 6.5 Sampling data

OpenFOAM provides the sample utility to sample field data, either through a 1D line for plotting on graphs or a 2D plane for displaying as isosurface images. The sampling locations are specified for a case through a *sampleDict* dictionary in the case *system* directory. The data can be written in a range of formats including well-known graphing packages such as Grace/xmgr, gnuplot and jPlot.

The *sampleDict* dictionary can be generated by copying an example *sampleDict* from the sample source code directory at *\$FOAM\_UTILITIES/postProcessing/sampling/sample*. The plateHole tutorial case in the *\$FOAM\_TUTORIALS/solidDisplacementFoam* directory also contains an example for 1D line sampling:

```
17
     interpolationScheme cellPoint;
18
19
     setFormat
                        raw;
20
^{21}
22
     sets
23
          leftPatch
24
25
26
               type
                        uniform;
27
               axis
                        y;
                         ( 0 0.5 0.25 );
               start
28
                           0 2 0.25);
29
               end
               nPoints 100;
30
          }
31
     );
32
33
```

Keyword	Options	Description		
interpolation-	cell	Cell-centre value assumed constant over cell		
Scheme	cellPoint	Linear weighted interpolation using cell values		
	cellPointFace	Mixed linear weighted / cell-face interpolation		
setFormat	raw	Raw ASCII data in columns		
	gnuplot	Data in <b>gnuplot</b> format		
	xmgr	Data in Grace/xmgr format		
	jplot	Data in jPlot format		
surfaceFormat	null	Suppresses output		
	foamFile	points, faces, values file		
	dx	DX scalar or vector format		
	vtk	VTK ASCII format		
	raw	xyz values for use with $e.g.gnuplotsplot$		
	stl	ASCII STL; just surface, no values		
fields	List of fields to be	sampled, $e.g.$ for velocity U:		
	U	Writes all components of $\mathbf{U}$		
sets	List of 1D sets subdictionaries — see Table $6.4$			
surfaces	List of 2D surfaces	s subdictionaries — see Table $6.5$ and Table $6.6$		

Table 6.3: keyword entries for *sampleDict*.

The dictionary contains the following entries:

interpolationScheme the scheme of data interpolation;

sets the locations within the domain that the fields are line-sampled (1D).

surfaces the locations within the domain that the fields are surface-sampled (2D).

setFormat the format of line data output;

surfaceFormat the format of surface data output;

fields the fields to be sampled;

The interpolationScheme includes cellPoint and cellPointFace options in which each polyhedral cell is decomposed into tetrahedra and the sample values are interpolated from values at the tetrahedra vertices. With cellPoint, the tetrahedra vertices include the polyhedron cell centre and 3 face vertices. The vertex coincident with the cell centre inherits the cell centre field value and the other vertices take values interpolated from cell centres. With cellPointFace, one of the tetrahedra vertices is also coincident with a face centre, which inherits field values by conventional interpolation schemes using values at the centres of cells that the face intersects.

The setFormat entry for line sampling includes a raw data format and formats for gnuplot, Grace/xmgr and jPlot graph drawing packages. The data are written into a *sets* directory within the case directory. The directory is split into a set of time directories and

the data files are contained therein. Each data file is given a name containing the field name, the sample set name, and an extension relating to the output format, including *.xy* for raw data, *.agr* for Grace/xmgr and *.dat* for jPlot. The gnuplot format has the data in raw form with an additional commands file, with *.gplt* extension, for generating the graph. Note that any existing sets directory is deleted when sample is run.

The surfaceFormat entry for surface sampling includes a raw data format and formats for gnuplot, Grace/xmgr and jPlot graph drawing packages. The data are written into a *surfaces* directory within the case directory. The directory is split into time directories and files are written much as with line sampling.

The fields list contains the fields that the user wishes to sample. The sample utility can parse the following restricted set of functions to enable the user to manipulate vector and tensor fields, *e.g.* for U:

U.component(n) writes the nth component of the vector/tensor, n = 0, 1...;

mag(U) writes the magnitude of the vector/tensor.

The sets list contains sub-dictionaries of locations where the data is to be sampled. The sub-dictionary is named according to the name of the set and contains a set of entries, also listed in Table 6.4, that describes the locations where the data is to be sampled. For example, a uniform sampling provides a uniform distribution of nPoints sample locations along a line specified by a start and end point. All sample sets are also given: a type; and, means of specifying the length ordinate on a graph by the axis keyword.

		Required entrie		es			
Sampling type	Sample locations	name	axis	start	end	nPoints	points
uniform	Uniformly distributed points on a line	٠	٠	٠	٠	٠	
face	Intersection of specified line and cell faces	•	•	•	•		
midPoint	Midpoint between line-face intersections	٠	•	•	•		
midPointAndFace	Combination of midPoint and face	•	•	•	•		
curve	Specified points, tracked along a curve	•	•				٠
cloud	Specified points	•	•				•
	1 1						

Entries	Description	Options	
type	Sampling type	see list abo	ove
axis	Output of sample location	х	x ordinate
		У	y ordinate
		Z	z ordinate
		xyz	xyz coordinates
		distance	distance from point 0
start	Start point of sample line	<i>e.g.</i> (0.0 0	.0 0.0)
end	End point of sample line	<i>e.g.</i> (0.0 2	.0 0.0)
nPoints	Number of sampling points	e.g.200	
points	List of sampling points		

Table 6.4: Entries within **sets** sub-dictionaries.

The surfaces list contains sub-dictionaries of locations where the data is to be sampled. The sub-dictionary is named according to the name of the surface and contains

Keyword	Description	Options
basePoint	Point on plane	<i>e.g.</i> (0 0 0)
normalVector	Normal vector to plane	<i>e.g.</i> (1 0 0)
interpolate	Interpolate data?	true/false
triangulate	Triangulate surface? (optional)	true/false

Table 6.5: Entries for a plane in surfaces sub-dictionaries.

Keyword	Description	Options
patchName	Name of patch	e.g.movingWall
interpolate	Interpolate data?	true/false
triangulate	Triangulate surface? (optional)	true/false

Table 6.6: Entries for a patch in surfaces sub-dictionaries.

a set of entries beginning with the type: either a plane, defined by point and normal direction, with additional sub-dictionary entries specified in Table 6.5; or, a patch, coinciding with an existing boundary patch, with additional sub-dictionary entries a specified in Table 6.6.

## 6.6 Monitoring and managing jobs

This section is concerned primarily with successful running of OpenFOAM jobs and extends on the basic execution of solvers described in section 3.3. When a solver is executed, it reports the status of equation solution to standard output, *i.e.* the screen, if the level debug switch is set to 1 or 2 (default) in *DebugSwitches* in the *\$WM\_PROJECT\_DIR/etc/controlDict* file. An example from the beginning of the solution of the cavity tutorial is shown below where it can be seen that, for each equation that is solved, a report line is written with the solver name, the variable that is solved, its initial and final residuals and number of iterations.

```
Starting time loop
Time = 0.005
Max Courant Number = 0
BICCG: Solving for Ux, Initial residual = 1, Final residual = 2.96338e-06, No Iterations 8
ICCG: Solving for p, Initial residual = 1, Final residual = 4.9336e-07, No Iterations 35
time step continuity errors : sum local = 3.29376e-09, global = -6.41065e-20, cumulative = -6.41065e-20
ICCG: Solving for p, Initial residual = 0.47484, Final residual = 5.41068e-07, No Iterations 34
time step continuity errors : sum local = 6.60947e-09, global = -6.22619e-19, cumulative = -6.86725e-19
ExecutionTime = 0.14 s
Time = 0.01
Max Courant Number = 0.585722
BICCG: Solving for Ux, Initial residual = 0.148584, Final residual = 7.15711e-06, No Iterations 6
BICCG: Solving for Uy, Initial residual = 0.256618, Final residual = 8.94127e-06, No Iterations 6
ICCG: Solving for p, Initial residual = 0.37146, Final residual = 6.67464e-07, No Iterations 33
time step continuity errors : sum local = 6.34431e-09, global = 1.20603e-19, cumulative = -5.66122e-19
ICCG: Solving for p, Initial residual = 0.271556, Final residual = 3.69316e-07, No Iterations 33
time step continuity errors : sum local = 3.96176e-09, global = 6.9814e-20, cumulative = -4.96308e-19
ExecutionTime = 0.16 \text{ s}
```

Time = 0.015

Max Courant Number = 0.758267

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BICCG: Solving for Uy, Initial residual = 0.0480673, Final residual = 2.425010-06, No Iterations 6 BICCG: Solving for Uy, Initial residual = 0.0782042, Final residual = 1.470090-06, No Iterations 7 ICCG: Solving for p, Initial residual = 0.107474, Final residual = 4.83620-07, No Iterations 32 time step continuity errors : sum local = 3.990280-09, global = -5.697620-19, cumulative = -1.066070-18 ICCG: Solving for p, Initial residual = 0.0806771, Final residual = 9.471710-07, No Iterations 31 time step continuity errors : sum local = 7.921760-09, global = 1.075330-19, cumulative = -9.585370-19 ExecutionTime = 0.19 s

## 6.6.1 The foamJob script for running jobs

The user may be happy to monitor the residuals, iterations, Courant number *etc.* as report data passes across the screen. Alternatively, the user can redirect the report to a log file which will improve the speed of the computation. The foamJob script provides useful options for this purpose with the following executing the specified *<solver>* as a background process and redirecting the output to a file named *log*:

foamJob <solver>

For further options the user should execute foamJob -help. The user may monitor the *log* file whenever they wish, using the UNIXtail command, typically with the -f 'follow' option which appends the new data as the *log* file grows:

tail -f log

## 6.6.2 The foamLog script for monitoring jobs

There are limitations to monitoring a job by reading the log file, in particular it is difficult to extract trends over a long period of time. The **foamLog** script is therefore provided to extract data of residuals, iterations, Courant number *etc.* from a log file and present it in a set of files that can be plotted graphically. The script is executed by:

foamLog <logFile>

The files are stored in a subdirectory of the case directory named *logs*. Each file has the name  $\langle var \rangle_{-} \langle sublter \rangle$  where  $\langle var \rangle$  is the name of the variable specified in the log file and  $\langle sublter \rangle$  is the iteration number within the time step. Those variables that are solved for, the initial residual takes the variable name  $\langle var \rangle$  and final residual takes  $\langle var \rangle$ *FinalRes*. By default, the files are presented in two-column format of time and the extracted values.

For example, in the cavity tutorial we may wish to observe the initial residual of the Ux equation to see whether the solution is converging to a steady-state. In that case, we would plot the data from the  $logs/Ux_0$  file as shown in Figure 6.5. It can be seen here that the residual falls monotonically until it reaches the convergence tolerance of  $10^{-5}$ .

 $\mathsf{foamLog}$  generates files for everything it feasibly can from the  $\log$  file. In the  $\mathsf{cavity}$  tutorial example, this includes:

- the Courant number, Courant\_0;
- Ux equation initial and final residuals, Ux\_O and UxFinalRes\_O, and iterations, UxIters\_O (and equivalent Uy data);
- cumulative, global and local continuity errors after each of the 2 p equations, contCumulative\_0, contGlobal\_0, contLocal\_0 and contCumulative\_1, contGlobal\_1, contLocal\_1;



Figure 6.5: Initial residual of Ux in the cavity tutorial

- residuals and iterations from the the 2 p equations p\_0, pFinalRes\_0, pIters\_0 and p\_1, pFinalRes\_1, pIters\_1;
- and execution time, executionTime.

## Chapter 7

## Models and physical properties

OpenFOAM includes a large range of solvers each designed for a specific class of problem. The equations and algorithms differ from one solver to another so that the selection of a solver involves the user making some initial choices on the modelling for their particular case. The choice of solver typically involves scanning through their descriptions in Table 3.5 to find the one suitable for the case. It ultimately determines many of the parameters and physical properties required to define the case but leaves the user with some modelling options that can be specified at runtime through the entries in dictionary files in the *constant* directory of a case. This chapter deals with many of the more common models and associated properties that may be specified at runtime.

## 7.1 Thermophysical models

Thermophysical models are concerned with the energy, heat and physical properties.

The thermophysicalProperties dictionary is read by any solver that uses the thermophysical model library. A thermophysical model is constructed in OpenFOAM as a pressuretemperature p - T system from which other properties are computed. There is one compulsory dictionary entry called thermoType which specifies the complete thermophysical model that is used in the simulation. The thermophysical modelling starts with a layer that defines the basic equation of state and then adds more layers of modelling that derive properties from the previous layer(s). The naming of the thermoType reflects these multiple layers of modelling as listed in Table 7.1.

Equation of State — equationOfState		
icoPolynomial	Incompressible polynomial equation of state, <i>e.g.</i> for liquids	
perfectGas	Perfect gas equation of state	
Basic thermophysical	properties - thermo	
eConstThermo	Constant specific heat $c_p$ model with evaluation of internal	
	energy $e$ and entropy $s$	
hConstThermo	Constant specific heat $c_p$ model with evaluation of enthalpy	
	h and entropy $s$	
hPolynomialThermo	$c_p$ evaluated by a function with coefficients from polynomi-	
	als, from which $h, s$ are evaluated	
janafThermo	$c_p$ evaluated by a function with coefficients from JANAF	
	thermodynamic tables, from which $h, s$ are evaluated	

#### Derived thermophysical properties — specieThermo

Continued on next page

specieThermo	Thermophysical properties of species, derived from $c_p$ , $b$ and/or $s$	
Transport properties —	- transport	
constTransport	Constant transport properties	
polynomialTransport	Polynomial based temperature-dependent transport properties	
sutherlandTransport	Sutherland's formula for temperature-dependent transpor properties	
Mixture properties — r	nixture	
pureMixture	General thermophysical model calculation for passive ga mixtures	
homogeneousMixture	Combustion mixture based on normalised fuel mass fraction $\boldsymbol{b}$	
inhomogeneousMixture	Combustion mixture based on $b$ and total fuel mass fraction $f_t$	
veryInhomogeneousMixture	•	
dieselMixture	Combustion mixture based on $f_t$ and $f_u$	
basicMultiComponent- Mixture	Basic mixture based on multiple components	
multiComponentMixture	Derived mixture based on multiple components	
reactingMixture	Combustion mixture using thermodynamics and reaction schemes	
egrMixture	Exhaust gas recirculation mixture	
Thermophysical model		
hPsiThermo	General thermophysical model calculation based on entropy $h$ and compressibility $\psi$	
hsPsiThermo	General thermophysical model calculation based on sens ble enthalpy $h_s$ and compressibility $\psi$	
ePsiThermo	General thermophysical model calculation based on internal energy $e$ and compressibility $\psi$	
hRhoThermo	General thermophysical model calculation based on er thalpy $h$	
hsRhoThermo	General thermophysical model calculation based on sens ble enthalpy $h_s$	
hPsiMixtureThermo	Calculates enthalpy for combustion mixture based on er thalpy $h$ and $\psi$	
hsPsiMixtureThermo	Calculates enthalpy for combustion mixture based on ser sible enthalpy $h_s$ and $\psi$	
hRhoMixtureThermo	Calculates enthalpy for combustion mixture based on er thalpy $h$ and $\rho$	
hsRhoMixtureThermo	Calculates enthalpy for combustion mixture based on ser sible enthalpy $h_s$ and $\rho$	
hhuMixtureThermo	Calculates enthalpy for unburnt gas and combustion mix	

Continued on next page
Continued from previous page

Table 7.1: Layers of thermophysical modelling.

The thermoType entry typically takes the form:

so that the following is an example entry for thermoType:

hThermo<pureMixture<constTransport<specieThermo<hConstThermo<perfectGas>>>>>

# 7.1.1 Thermophysical property data

The basic thermophysical properties are specified for each species from input data. Data entries must contain the name of the specie as the keyword, *e.g.* 02, H20, mixture, followed by sub-dictionaries of coefficients, including:

- specie containing i.e. number of moles, nMoles, of the specie, and molecular weight, molWeight in units of g/mol;
- thermodynamics containing coefficients for the chosen thermodynamic model (see below);

transport containing coefficients for the chosen transport model (see below).

The thermodynamic coefficients are ostensibly concerned with evaluating the specific heat  $c_p$  from which other properties are derived. The current **thermo** models are described as follows:

- hConstThermo assumes a constant  $c_p$  and a heat of fusion  $H_f$  which is simply specified by a two values  $c_p$   $H_f$ , given by keywords Cp and Hf.
- eConstThermo assumes a constant  $c_v$  and a heat of fusion  $H_f$  which is simply specified by a two values  $c_v H_f$ , given by keywords Cv and Hf.
- janafThermo calculates  $c_p$  as a function of temperature T from a set of coefficients taken from JANAF tables of thermodynamics. The ordered list of coefficients is given in Table 7.2. The function is valid between a lower and upper limit in temperature  $T_l$ and  $T_h$  respectively. Two sets of coefficients are specified, the first set for temperatures above a common temperature  $T_c$  (and below  $T_h$ , the second for temperatures below  $T_c$  (and above  $T_l$ ). The function relating  $c_p$  to temperature is:

$$c_p = R((((a_4T + a_3)T + a_2)T + a_1)T + a_0)$$
(7.1)

In addition, there are constants of integration,  $a_5$  and  $a_6$ , both at high and low temperature, used to evaluating h and s respectively.

 $\label{eq:holynomialThermo} \mbox{ hPolynomialThermo} \mbox{ calculates } C_p \mbox{ as a function of temperature by a polynomial of any order. The following case provides an example of its use: $FOAM_TUTORIALS/-lagrangian/porousExplicitSourceReactingParcelFoam/filter}$ 

The transport coefficients are used to to evaluate dynamic viscosity  $\mu$ , thermal conductivity  $\kappa$  and laminar thermal conductivity (for enthalpy equation)  $\alpha$ . The current transport models are described as follows:

Description	Entry	Keyword
Lower temperature limit	$T_l$ (K)	Tlow
Upper temperature limit	$T_h$ (K)	Thigh
Common temperature	$T_c$ (K)	Tcommon
High temperature coefficients	$a_0 \dots a_4$	highCpCoeffs (a0 a1 a2 a3 a4
High temperature enthalpy offset	$a_5$	a5
High temperature entropy offset	$a_6$	a6)
Low temperature coefficients	$a_0 \dots a_4$	lowCpCoeffs (a0 a1 a2 a3 a4
Low temperature enthalpy offset	$a_5$	a5
Low temperature entropy offset	$a_6$	a6)

Table $7.2$ :	JANAF	thermodynamics	coefficients.
---------------	-------	----------------	---------------

- constTransport assumes a constant  $\mu$  and Prandtl number  $Pr = c_p \mu / \kappa$  which is simply specified by a two keywords, mu and Pr, respectively.
- sutherlandTransport calculates  $\mu$  as a function of temperature T from a Sutherland coefficient  $A_s$  and Sutherland temperature  $T_s$ , specified by keywords As and Ts;  $\mu$  is calculated according to:

$$\mu = \frac{A_s \sqrt{T}}{1 + T_s/T} \tag{7.2}$$

polynomial Transport calculates  $\mu$  and  $\kappa$  as a function of temperature T from a polynomial of any order.

The following is an example entry for a specie named fuel modelled using sutherland-Transport and janafThermo:

```
fuel
{
    specie
    {
        nMoles
                      1;
        molWeight
                      16.0428;
    thermodynamics
    ł
        Tlow
                      200;
                      6000;
        Thigh
        Tcommon
                      1000;
        highCpCoeffs (1.63543 0.0100844 -3.36924e-06 5.34973e-10
                       -3.15528e-14 -10005.6 9.9937);
        lowCpCoeffs (5.14988 -0.013671 4.91801e-05 -4.84744e-08
                       1.66694e-11 - 10246.6 - 4.64132);
    ł
    transport
    ł
        As
                      1.67212e-06;
        Ts
                      170.672;
```

}

The following is an example entry for a specie named  $\mathtt{air}$  modelled using  $\mathtt{constTransport}$  and  $\mathsf{hConstThermo}$ :

```
air
{
    specie
     ł
         nMoles
                            1;
                            28.96;
         molWeight
     }
    thermodynamics
     ł
                            1004.5;
         Ср
         Hf
                            2.544e+06;
     }
    transport
     ł
                            1.8e-05;
         m11
         Pr
                            0.7;
    }
}
```

# 7.2 Turbulence models

The *turbulenceProperties* dictionary is read by any solver that includes turbulence modelling. Within that file is the **simulationType** keyword that controls the type of turbulence modelling to be used, either:

laminar uses no turbulence models;

RASModel uses Reynolds-averaged stress (RAS) modelling;

LESModel uses large-eddy simulation (LES) modelling.

If RASModel is selected, the choice of RAS modelling is specified in a *RASProperties* file, also in the *constant* directory. The RAS turbulence model is selected by the RASModel entry from a long list of available models that are listed in Table 3.9. Similarly, if LESModel is selected, the choice of LES modelling is specified in a *LESProperties* dictionary and the LES turbulence model is selected by the LESModel entry.

The entries required in the *RASProperties* are listed in Table 7.3 and those for *LESProperties* dictionaries are listed in Table 7.4.

RASModel	Name of RAS turbulence model
turbulence	Switch to turn turbulence modelling on/off
printCoeffs	Switch to print model coeffs to terminal at simulation startup
<rasmodel>Coeffs</rasmodel>	Optional dictionary of coefficients for the respective RASModel

Table 7.3: Keyword entries in the *RASProperties* dictionary.

<b>U-184</b>	
--------------	--

LESModel	Name of LES model
delta	Name of delta $\delta$ model
< LESModel > Coeffs	Dictionary of coefficients for the respective LESModel
<delta>Coeffs</delta>	Dictionary of coefficients for each delta model

Table 7.4: Keyword entries in the *LESProperties* dictionary.

The incompressible and compressible RAS turbulence models, isochoric and anisochoric LES models and delta models are all named and described in Table 3.9. Examples of their use can be found in the  $FOAM_TUTORIALS$ .

# 7.2.1 Model coefficients

The coefficients for the RAS turbulence models are given default values in their respective source code. If the user wishes to override these default values, then they can do so by adding a sub-dictionary entry to the *RASProperties* file, whose keyword name is that of the model with Coeffs appended, *e.g.* kEpsilonCoeffs for the kEpsilon model. If the printCoeffs switch is on in the *RASProperties* file, an example of the relevant ...Coeffs dictionary is printed to standard output when the model is created at the beginning of a run. The user can simply copy this into the *RASProperties* file and edit the entries as required.

# 7.2.2 Wall functions

A range of wall function models is available in OpenFOAM that are applied as boundary conditions on individual patches. This enables different wall function models to be applied to different wall regions. The choice of wall function model is specified through:  $\nu_t$  in the 0/nut file for incompressible RAS;  $\mu_t$  in the 0/mut file for compressible RAS;  $\nu_{sgs}$  in the 0/nuSgs file for incompressible LES;  $\mu_{sgs}$  in the 0/muSgs file for incompressible LES. For example, a 0/nut file:

```
17
                    [0 \ 2 \ -1 \ 0 \ 0 \ 0];
    dimensions
18
19
    internalField
                    uniform 0;
20
21
    boundaryField
22
23
    {
24
        movingWall
25
            type
value
                            nutkWallFunction;
26
27
                            uniform 0:
28
        fixedWalls
29
30
                            nutkWallFunction;
31
            type
            value
                            uniform 0;
32
33
        frontAndBack
34
        {
35
            type
                            empty;
36
        }
37
    }
38
39
40
```

There are a number of wall function models available in the release, *e.g.* nutWallFunction, nutRoughWallFunction, nutSpalartAllmarasStandardRoughWallFunction, nutSpalartAllmarasStandardWallFunction and nutSpalartAllmarasWallFunction. The user can consult the relevant directories for a full list of wall function models:

#### find \$FOAM\_SRC/turbulenceModels -name wallFunctions

Within each wall function boundary condition the user can over-ride default settings for E,  $\kappa$  and  $C_{\mu}$  through optional E, kappa and Cmu keyword entries.

Having selected the particular wall functions on various patches in the nut/mut file, the user should select epsilonWallFunction on corresponding patches in the epsilon field and kqRwallFunction on corresponding patches in the turbulent fields k, q and R.

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\*

#### Symbols Numbers A B C D E F G H I J K L M N O P Q R S T U V W X Z

#### Symbols

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